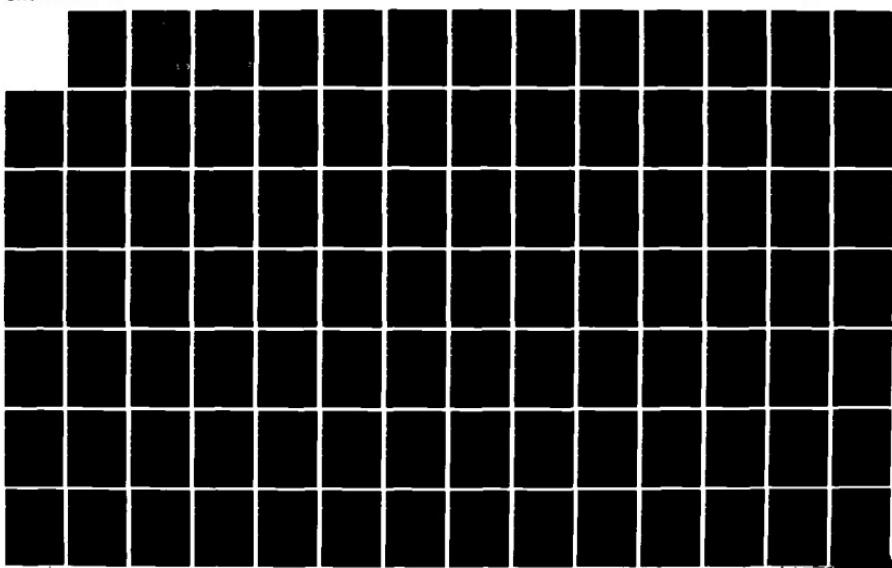
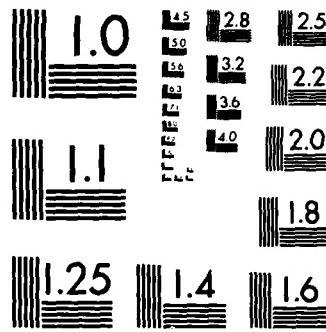


RD-A153 039

HEATING PARAMETER ESTIMATION USING COAXIAL THERMOCOUPLE 1/2
GAGES IN WIND TUN. (U) AIR FORCE INST OF TECH
WRIGHT-PATTERSON AFB OH SCHOOL OF ENGI. N T CAHOON
DEC 84 AFIT/GAE/RA/84D-3 F/G 9/2 NL

UNCLASSIFIED





MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS-1963-A

AD-A153 039



HEATING PARAMETER ESTIMATION USING
COAXIAL THERMOCOUPLE GAGES IN
WIND TUNNEL TEST ARTICLES

THESIS

Neil T. Cahoon
Captain, USAF

AFIT/GAE/AA/84D-3

DISTRIBUTION STATEMENT A
Approved for public release
Distribution Unlimited

DTIC ELECTE
S MAY 1 1985
B

DTIC FILE COPY

DEPARTMENT OF THE AIR FORCE
AIR UNIVERSITY
AIR FORCE INSTITUTE OF TECHNOLOGY

Wright-Patterson Air Force Base, Ohio

AFIT/GAE/AA/84D-3

HEATING PARAMETER ESTIMATION USING
COAXIAL THERMOCOUPLE GAGES IN
WIND TUNNEL TEST ARTICLES

THESIS

Neil T. Cahoon
Captain, USAF

AFIT/GAE/AA/84D-3

DTIC
ELECTED
MAY 1 1985
S D
B

Approved for public release; distribution unlimited

**HEATING PARAMETER ESTIMATION USING COAXIAL
THERMOCOUPLE GAUGES IN WIND TUNNEL
TEST ARTICLES**

THESIS

**Presented to the Faculty of the School of Engineering
of the Air Force Institute of Technology
Air University
In Partial Fulfillment of the
Requirements for the Degree of
Master of Science in Aeronautical Engineering**

Neil T. Cahoon, B.S.E.

Captain, USAF

December 1984

Approved for public release; distribution unlimited

Acknowledgements

I wish to express my sincere thanks and gratitude to my thesis advisor, Major James K. Hodge for his expert guidance during the course of this study. His ability to provide engineering insight ("voodoo engineering") and open the doors of understanding never ceased. I would also like to say thanks for the encouragement when the obstacles appeared overwhelming.

Also to my wife, Shirley, for the sacrifices she made during the course of this academic experience, I would like to say thanks. She provided immeasurable support and encouragement during my long hours of study.

Approved For
Distribution
by _____
Availability Codes
Avail and/or
Special

A-1

Approved For	_____
Distribution	_____
Availability Codes	_____
Avail and/or	_____
Special	_____

Table of Contents

Acknowledgments	ii
List of Figures	iv
List of Symbols	v
Abstract	vii
I. Introduction	1
1.1 Background	1
1.2 Objectives	5
1.3 Overview	6
II. The Thermal Model	7
2.1 Temperature Equation	7
2.2 Sensitivity Equations	11
2.3 Covariance Equations	12
III. HEATEST Overview	14
IV. Results	20
4.1 Test Procedure	20
4.2 Test Cases	22
V. Conclusions and Recommendations	44
5.1 Conclusions	44
5.2 Recommendations	45
Appendix A: One-dimensional Energy Balance	46
Appendix B: Sensitivity Equations	52
Appendix C: HEATEST	57
Bibliography	58
Vita	59

List of Figures

Figure	Page
1.1 Coaxial Thermocouple Gage	4
3.1 HEATEST Algorithm Summary	15
4.1 Input Data	27
4.2 Discrete Data	28
4.3 Temp. vs Node Point, 1 iteration	29
4.4 Temp. vs Node Point, converged	30
4.5 Temp. vs Node Point, $\Delta t = .5$ sec	31
4.6 Temp. vs Node Point, $\Delta t = .25$ sec	32
4.7 Temp. vs Node Point, $\Delta t = 1$ and $.25$ sec	33
4.8 Temp. vs Node Point, T/C length=.1 ft.	34
4.9 Temp. vs Node Point, T/C length=.05 ft.	35
4.10 Temp. vs Node Point, T/C length=.025 ft.	36
4.11 Temp. vs Position, T/C length=.1,.05,.025 ft. ...	37
4.12 Temp. vs Node Point, 10 sec, 6 nodes	38
4.13 Temp. vs Node Point, 3 sec, 6 nodes	39
4.14 Temp. vs Node Point, 10 sec, 12 nodes	40
4.15 Temp. vs Node Point, 3 sec, 12 nodes	41
4.16 Temp. vs Position, 6 vs 12 nodes, 10 sec.	42
4.17 Temp. vs Position, 6 vs 12 nodes, 3 sec.	43

List of Symbols

A, A' , b, d	Coefficient Matrices (i x i)
c	Specific Heat
E	Residual Error Vector (m)
G	Kalman Gain Vector (i)
H	Thermocouple Location Matrix (m x i)
h_0	Magnitude of Heat Transfer Coefficient Ratio
h_s	Heat Transfer Coefficient Derivative
$h_{\bar{}}_0$	Heat Transfer Coefficient Ratio
h_{ref}	Reference Heat Transfer Coefficient at Zero State
I	Identity Matrix
J_k	Conditional Information Matrix
k	Thermal Conductivity
L	Total Number of Spatial Node Points
P	Covariance Matrix (i x i)
Q	Model Error Covariance Matrix (i x i)
q	Heating Rate
R_m	Covariance for mth measurement
S	Score Vector (k)
$S_{i,k}$	Sensitivity Vectors (i) for the kth Parameter
T_{aw}	Adiabatic Wall Temperature
t	Time
U	Temperature Vector (i)

x	Spatial Coordinate
y	Thermocouple Measurement Vector
a	Angle of Attack
e	Emissivity
θ	Parameter Vector
μ_n	Measurement Vector at nth Time Point
ρ	Density
σ	Stefan-Boltzmann Constant
δ	Transition Matrix
ϕ_c	Scaling Parameter for Specific Heat
ϕ_k	Scaling Parameter for Thermal Conductivity

Superscripts

-	a priori Propogation
+	a posteriori Propogation
*	Parameter Estimate
n	Time Level
s	Iteration Level
T	Transpose

Subscripts

i	Spatial Node Point
k	Number of Model Parameter
m	Number of Thermocouples
o	Freestream Conditions

Abstract

A heat energy balance is applied to a coaxial thermocouple gage for parameter estimation in wind tunnel test articles. This method can significantly reduce wind tunnel test costs and time. Modifications to the data reduction technique HEATEST (HEATing ESTimation) are made. The program allows for transient test techniques to be used as well as assuming an isothermal wall. A non-linear convective heat transfer coefficient model may also be used. Data is generated to test the new program. Temperature profiles throughout the thermocouple gage were good and were compared with changes in time step, thermocouple length, and number of discrete node points. The estimation of the convective heat transfer coefficient and thermal conductivity were excellent.

**HEATING PARAMETER ESTIMATION USING COAXIAL
THERMOCOUPLE GAUGES IN WIND TUNNEL
TEST ARTICLES**

I. INTRODUCTION

1.1 Background

The determination of heat transfer rates on hypersonic configurations upon reentry is important for the survival of the vehicle. The problem is that the heat rate is not a quantity which may be directly scaled from model tests in wind tunnels. However, the parameters which make up the heat rate equation (thermal conductivity and specific heat, for example) can be scaled from which the heat rate may then be calculated. Wind tunnel heat transfer measurements have traditionally used a thin walled test model fabricated with thermocouples mounted on the inside skin surface. The "thermal model" then yields a heat rate based on temperature measurement from the thermocouple. Another technique uses a coaxial thermocouple gage mounted in a thick skin model. A discussion of the two methods follows.

The Traditional Thin Skin Model

The "thermal model" of a traditional wind tunnel thin

skin model assumes that all of the heat penetrates the thin skin via conduction to a standard thermocouple gage mounted on the back face. No lateral conduction is assumed and since the emittance of the steel model is low, radiation is assumed negligible. A typical heat rate measurement data point is acquired by injecting the cooled model into the wind tunnel at a known temperature and time and by measuring the temperature at later times. The model is then removed from the tunnel, cooled, and a change in configuration is made in preparation for the next injection and subsequent data point. There are three very severe limitations associated with this technique (Ref 1). The first is the inability to acquire more than one data point during any one injection. The thermal model simply does not allow for the type of change in configuration or model attitude which can be accomplished using dynamic testing techniques (to be discussed later). Associated with this limitation is the long cooling time between each test which significantly increases the cost per data point for the overall test. A second limitation is the special thin skin model which must be fabricated, further contributing to increased test cost. Finally, the assumption of no lateral conduction through the model may in fact be a poor assumption at some critical locations with large curvature. An alternate type of gage, the coaxial thermocouple, can eliminate these limitations with an overall effect of reducing time and cost.

A New Application For An Old Thermocouple

The coaxial thermocouple gage is shown in Fig. 1.1. It consists of a constantan (a metal alloy) jacket surrounding a chromel core with a thin layer of insulation separating the two metals. The coaxial gage is mounted in a steel model thick enough so that the thermal pulse is not sensed on the backface (ie. the model wall is considered a semi-infinite slab). The thermocouple surface is formed when the gage is lightly sanded to match the contour of the model. Some gages are available with backface temperature monitoring to assure that the thermal pulse does not reach the backface in any given run so that an analytical integration of the heat equation can be used to determine the heating rate history. The backface temperature information prior to this investigation is not factored into the data reduction process, however.

Operation of the coax gage is based on uniform conduction along the gage length which would necessitate the model be made of a material with similar thermal properties (Ref 3). The thermal properties of stainless steel match very closely with the gage properties, therefore, the presence of the gage is negligible. The matching of thermal properties also enhances accuracy. A coaxial gage which is matched thermally with the model allows an isothermal wall assumption, whereas other gauges such as calorimeters and thin film gages are not thermally matched, and cause a non-isothermal wall. Measured heat transfer can be in error by

$$\begin{aligned}\{\bar{U}\} &= [A]\{U\} + \{b\} + W(t) \\ \{S_k\} &= [A]\{S_k\} + \{d_k\}\end{aligned}\quad (3-1)$$

These equations are solved using a tridiagonal algorithm in subroutine TPS3 for the temperature states, and subroutine SENS for the sensitivity of the temperature to the kth parameter. Propogation of the covariance, P, of the temperature state at each node is accomplished by the approximate difference equation,

$$\begin{aligned}P(t_n^-) &= \phi(\Delta t)P(t_{n-1}^+)\phi^T(\Delta t) \\ &+ \int_{t_{n-1}}^{t_n} \phi(t_n-\lambda)Q\phi^T(t_n-\lambda)d\lambda\end{aligned}\quad (3-2)$$

where ϕ is the transition matrix and where the - and + superscripts are used to denote the expected values before an update (or a priori) and updated (or a posteriori) values, respectively. This calculation is made in subroutine TPS0SP2.

A model of the temperature measurement process must be used for the Kalman filter equations. The measurement equation to identify thermocouple location is,

$$Y(t_n) = H\{U(t_n)\} + \{\mu_n\} \quad (3-3)$$

The updated temperature is calculated by,

$$U(t_n^+) = U(t_n^-) + GE(t_n) \quad (3-4)$$

time. They are found by employing a Kalman filter - a set of recursive equations that optimally combine the propagation of the model equations with measurement updates at each sample time. After the entire temperature - time (state) history has been calculated, a gradient algorithm is used to solve for best estimates of the parameters according to a maximum likelihood criterion.

The second type of estimate consists of the parameters defined in the parameter vector, $\{\theta\}$, as given in Equation 2-10. These parameters remain essentially constant throughout a transient maneuver profile such as a pitch sweep and are estimated based on data from the entire maneuver history. The process of estimating states and parameters is then iterated for convergence to some optimal estimate.

The method used to estimate the states and parameters is formulated from stochastic estimation theory and is known as adaptive estimation. A detailed development of the estimation equations is beyond the scope of this thesis and the reader is referred to References 7 and 8 for more detail.

The thermal model equations for temperature and sensitivity have already been written in the matrix stochastic estimation form of Equation 2-12 as,

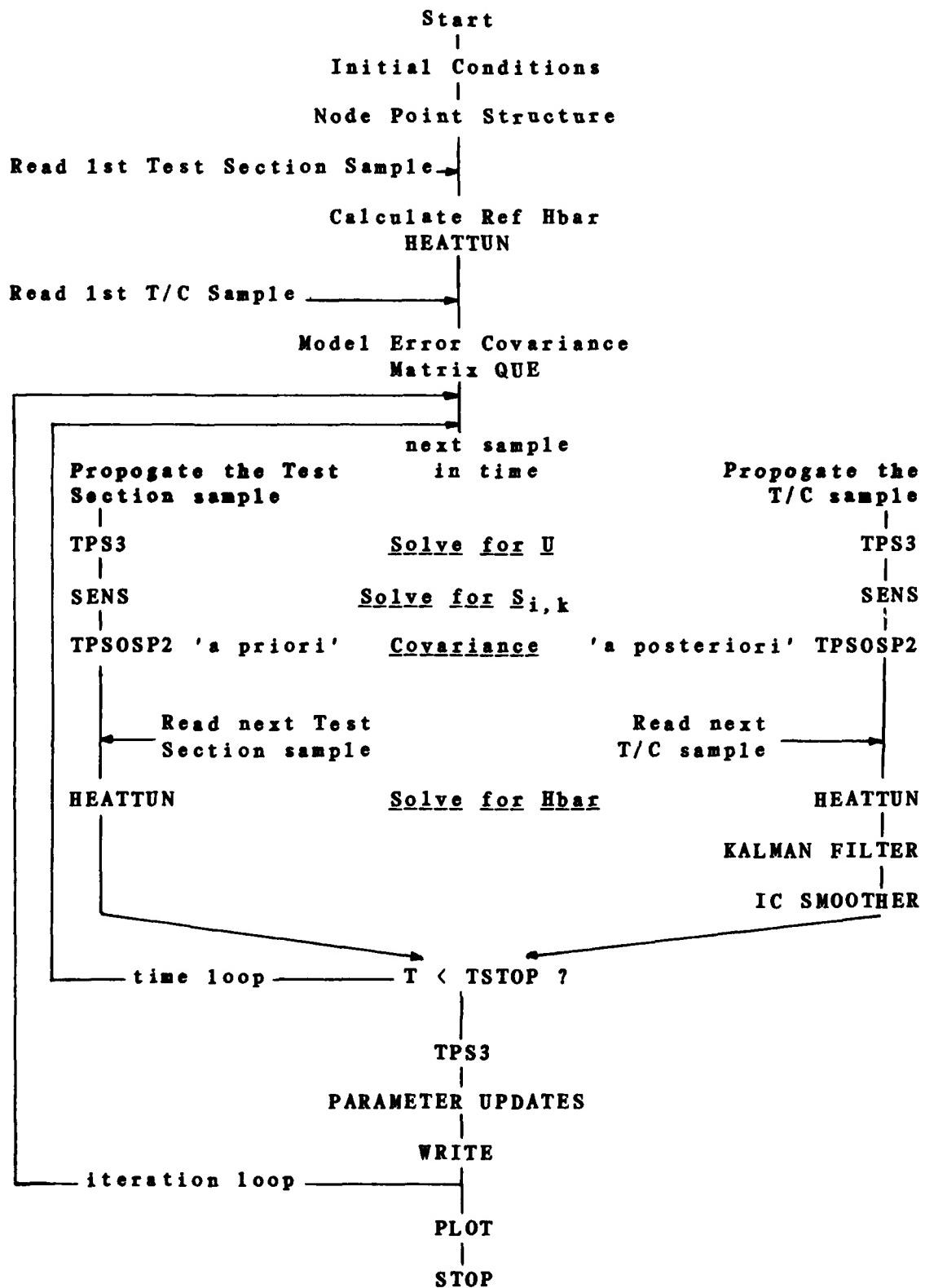


Figure 3.1 HEATEST Algorithm Summary

III. HEATEST OVERVIEW

The HEATEST program was originally developed to determine and model heat rates from the Space Shuttle Orbiter thermocouple data, hence most of its nomenclature references flight data samples and trajectory samples. The wind tunnel equivalence of the trajectory sample would be the test section conditions at the time of the sample (ie. density, velocity, pressure, etc.). The flight data are the thermocouple measurements from the coaxial gages.

An algorithm summary of the HEATEST program is given in Figure 3.1. The initial conditions for the temperature profile, $U(t)$, and the initial covariance, $P(t)$, are specified at the start of the wind tunnel test run. Heating model initial parameters, and the initial reference values for the heating model are read in as inputs to the program. Initial sensitivities of the state are specified to be zero. The node point structure throughout the depth of the thermocouple is then calculated from the input of the length of the gage and the number of node points.

Two types of estimates must be made in order to describe the thermodynamic environment in the wind tunnel. The first type are the state estimates, which are defined by each node temperature. These state estimates are not constant since the temperature varies throughout the maneuver, and hence, must be estimated at each node point in

The {b} vector is not used directly for the covariance equation, but is approximated by an error model given by,

$$Q_{\text{error}} = [h_{\text{bar}} h_{\text{ref}} (T_{\text{aw}} - U_1) \Delta x / \phi_k k]^2 \quad (2-15)$$

where S is the sensitivity. The i subscript identifies the node point and the second subscript identifies the particular parameter number. The sensitivity equations may also be written in the familiar form,

$$[A'] \{S_{i,k}^n\} + \{d\} = 0 \quad (2-12)$$

The sensitivity equations are developed and summarized in Appendix B.

2.3 Covariance Equation

Propagation of the covariance of the temperature state at each node requires the equations to be of the form

$$\dot{\{U\}} = [A]\{U\} + \{b\}$$

$$\text{and } \dot{\{S_{i,k}\}} = [A]\{S_k\} + \{d_k\} \quad (2-13)$$

Substituting the definitions of Equations (A-14) into Equations (2-7) and (2-8) and rearranging yields a common tridiagonal $[A]$ matrix for the above equations which is shown presently,

$$[A] = \begin{bmatrix} -\frac{[RM_1 + RP_1 + 4\pi\sigma(U_1^n)^3 + h_{bar}h_{ref}]}{RCX_1} & \frac{RP_1}{RCX_1} & 0 \\ \frac{RM_i}{RCX_i} & -\frac{(RM_i + RP_i)}{RCX_i} & \frac{RP_i}{RCX_i} \\ 0 & & \end{bmatrix} \quad (2-14)$$

$$[A'] \{U_i^n\} + \{b\} = 0 \quad (2-9)$$

where $[A']$ is an $n \times n$ tridiagonal matrix of material properties and $\{U_i^n\}$ is the n -dimensional column vector of unknown temperature at each node point for each time.

In general, the unknown parameters in this model formulation are the heat transfer coefficient intercept, h_0 , the slopes h_{a1} and h_{a2} , and the scaling parameters for specific heat and thermal conductivity, ϕ_c and ϕ_k , respectively. These parameters may be defined as a vector, θ , of unknown parameters for use in the system identification scheme as,

$$\theta = \{h_0, h_{a1}, h_{a2}, \phi_c, \phi_k\}^T \quad (2-10)$$

The primary purpose of the heating estimation program is to obtain best estimates of these parameters during transient test maneuvers. To estimate these parameters it is necessary to calculate the model sensitivity to each unknown parameter.

2.2 Sensitivity Equations

The derivative of Equation (2-7) with respect to each parameter yields equations of the same form as Equation (2-9) from which the HEATEST program propagates the sensitivity. For example, the sensitivity of the temperature with respect to h_0 would be written as follows,

$$\frac{\partial U}{\partial \theta_1} = \frac{\partial U}{\partial h_0} = S_{h0} = S_{i,1} \quad (2-11)$$

level defined by the superscript n,

$$\begin{aligned}
 (U_1^{n,s+1})^4 &= (U_1^{n,s})^4 + 4(U_1^{n,s})^3[U_1^{n,s+1} - U_1^{n,s}] \\
 &= -3(U_1^{n,s})^4 + 4(U_1^{n,s})^3U_1^{n,s+1} \quad (2-6)
 \end{aligned}$$

Substituting Equation (2-6) into (2-5) yields,

$$\begin{aligned}
 \frac{\rho \phi_c c \Delta x}{2} \frac{U_1^{n-1} - U_1^{n-1}}{\Delta t} &= \frac{-\phi_k k_{1/2}}{\Delta x} U_1^n + \frac{\phi_k k_{1/2}}{\Delta x} U_2^n \\
 &- \sigma [-3(U_1^{n,s})^4 + 4(U_1^{n,s})^3U_1^{n,s+1} - (U_{\infty}^n)^4] \\
 &+ [h_0 + h_{a1}(a-a_1) + h_{a2}(a-a_2)] h_{ref} (T_{aw} - U_1^{n,s+1}) \quad (2-7)
 \end{aligned}$$

The model equation for the interior node points, ($i=2, imax$), yields,

$$\begin{aligned}
 \frac{\rho \phi_l c c \Delta x}{\Delta t} \frac{U_i^{n-1} - U_i^{n-1}}{\Delta x} &= \frac{\phi_k k_{i-1/2}}{\Delta x} U_{i-1} \\
 &- \frac{\phi_k (k_{i-1/2} + k_{i+1/2})}{\Delta x} U_i \\
 &+ \frac{\phi_k k_{i+1/2}}{\Delta x} U_{i+1} \quad (2-8)
 \end{aligned}$$

Equations (2-7) and (2-8) can be rearranged into the familiar matrix form,

parameters other than those included in the reference heat transfer coefficient are summarized by the static transfer relation or heat transfer coefficient ratio, $h_{bar} = h/h_{ref}$. Here, the ratio is assumed to be piecewise linear with respect to angle of attack as derived from Lagrange Interpolation Theory (Ref 6 and Ref therin).

$$h_{bar} = [h_0 + h_{a1}(a - a_1) + h_{a2}(a - a_2)] \quad (2-4)$$

where h_0 is the magnitude of the heat transfer coefficient, h at the reference conditions, a_1 , specified by the one subscript. The heating parameters h_0 , h_{a1} , and h_{a2} are considered to be unknown and constant over a prescribed time period, and will be estimated by the HEATEST program. The parameters correspond to a derivative with respect to deflection angle of the model. Thus, for constant step size, the model equation at the surface node, ($i=1$), becomes,

$$\begin{aligned} \frac{\rho \beta_c c \Delta x}{2} \frac{U_1^n - U_1^{n-1}}{\Delta t} &= \frac{-\beta_k k_{1+1/2}}{\Delta x} U_1^n \\ &+ \frac{\beta_k k_{1+1/2}}{\Delta x} U_2^n - \epsilon \sigma [(U_1^n)^4 - (U_{\infty}^n)^4] \\ &+ [h_0 + h_{a1}(a - a_1) + h_{a2}(a - a_2)] h_{ref} (T_{aw} - U_1^n) \end{aligned} \quad (2-5)$$

The non-linear radiation term is quasi-linearized on an iteration level defined by the superscript s and by the time

where ϵ radiative emissivity
 σ StefanBoltzmann constant
 c material Specific Heat
 ρ material density
 k Thermal Conductivity
 δ_c Specific Heat scaling parameter
 δ_k Thermal Conductivity scaling parameter

The material specific heat and thermal conductivity are both scaled by the two factors δ_c and δ_k , respectively, hence the value for c and k will remain unchanged. The parameters δ_c and δ_k will be estimated by the HEATEST program. Coefficients with subscripts which are less than one or greater than n are zero. The radiation and heat rate terms are also zero except at the surface node. Equation 2-1 includes terms due to conduction from adjacent node points $k_{i-1/2}/\Delta x_{i-1}$, surface radiation σU_i^4 , and the convective transfer of energy as obtained from the heating model. The resulting system of implicit difference equations must be solved simultaneously.

The heating model for the convective transfer of energy is based on Newton's Law of Cooling,

$$q = h(T_{aw} - T) \quad (2-2)$$

Non-dimensionalizing by a reference heat transfer coefficient, h_{ref} yields,

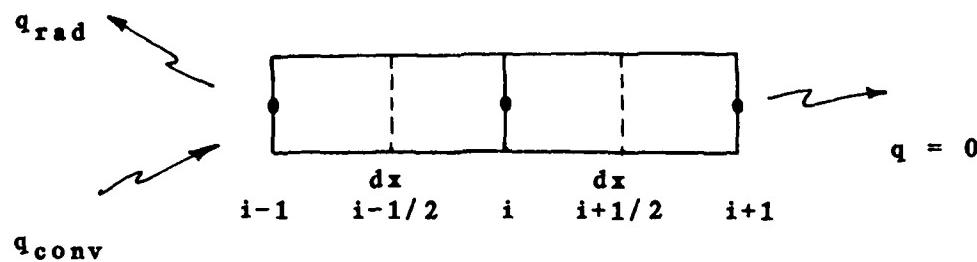
$$q = h_{bar} h_{ref} (T_{aw} - U_i) \quad (2-3)$$

The dependance of the heat transfer coefficient on

II. THE THERMAL MODEL

2.1 Temperature Equations

A cross section of the one-dimensional model is given in Figure 1.1 and below as a typical coaxial thermocouple gage.



An energy balance is performed on each element. The thermal conductivity, k , is taken as an average between each node. Fourier's Law of Heat Conduction throughout the gage, the Stefan-Boltzmann Law for radiation and Newton's Law of Cooling for convection on the surface face yield a system of n nonlinear differential equations of the form:

$$\begin{aligned}
 & [(\rho_i \phi_c c_i \Delta x_i + \rho_{i-1} \phi_c c_{i-1} \Delta x_{i-1}) / 2] [(U_i^n - U_i^{n-1}) \Delta t] \\
 & = \phi_k [k_{i-1/2} / \Delta x_{i-1}] U_{i-1}^n \\
 & \quad - \phi_k [k_{i-1/2} / \Delta x_{i-1} + k_{i+1/2} / \Delta x_{i+1}] U_i^n \\
 & \quad + \phi_k [k_{i+1/2} / \Delta x_{i+1}] U_{i+1}^n \\
 & \quad - \epsilon \sigma (U_i^4 - U_{i-1}^4) + h_{bar} h_{ref} (T_{aw} - U_i) \quad (2-1)
 \end{aligned}$$

gage using a Kalman filter(Ref 2). The purpose of this investigation is to incorporate the appropriate thermal model equations into the HEATEST program for application to coaxial thermocouple gages as used in the wind tunnel. Two reasons for doing this are, 1) replace the analytical with the semi-infinite assumption, ie. extend the run time or shorten the gage, and, 2) estimate thermal properties. Testing and verification of the modified program is necessary for verification of the validity of the results. Simulated data are generated by an analytical solution, and are processed for testing purposes for which the results are known.

1.3 Overview

A development of the pertinent temperature, sensitivity, and covariance equations will be developed for introduction into the HEATEST program in Chapter II. Details in format for programming may be found in Appendix A & B. Chapter III is an overview of the HEATEST algorithm and shows how the equations developed in Chapter II are utilized. Chapter IV outlines the method for testing the program and offers a discussion of the test cases made and results. Finally, conclusions about the validity of the modifications, and suggestions for further improvement are included in Chapter V.

up to 40% because of the non-isothermal (Ref 2). The same rugged model built for pressure measurements can be used for temperature measurements as well, which would further reduce the wind tunnel costs. The data reduction technique, which uses the temperature time history, eliminates the requirement to fix the model configuration or attitude during any one run, hence dynamic testing techniques may be used similar to the flight test technique used for the Space Shuttle Orbiter (Ref 4). The model may be swept in angle of attack, for example, to determine heat rates as a function of angle of attack. All of these attributes along with a short response time and no required calibration (ref Knox) combine to yield the wind tunnel engineer a tool of marked improvement over previous methods.

1.2 OBJECTIVES

A method of analysis to identify the aerothermodynamic flight environment and update the thermal model of the engineering simulation of the Space Shuttle Orbiter was designed by the Air Force Flight Test Center. This method is in the form of a digital computer program called HEATEST (HEATING ESTimation). The program provides a correlation of the heating as well as a heat rate time history. The program integrates numerically instead of relying on some analytical assumption. It satisfies a maximum likelihood criteria for each parameter and obtains best estimates for the temperature at discrete nodes throughout the length of the

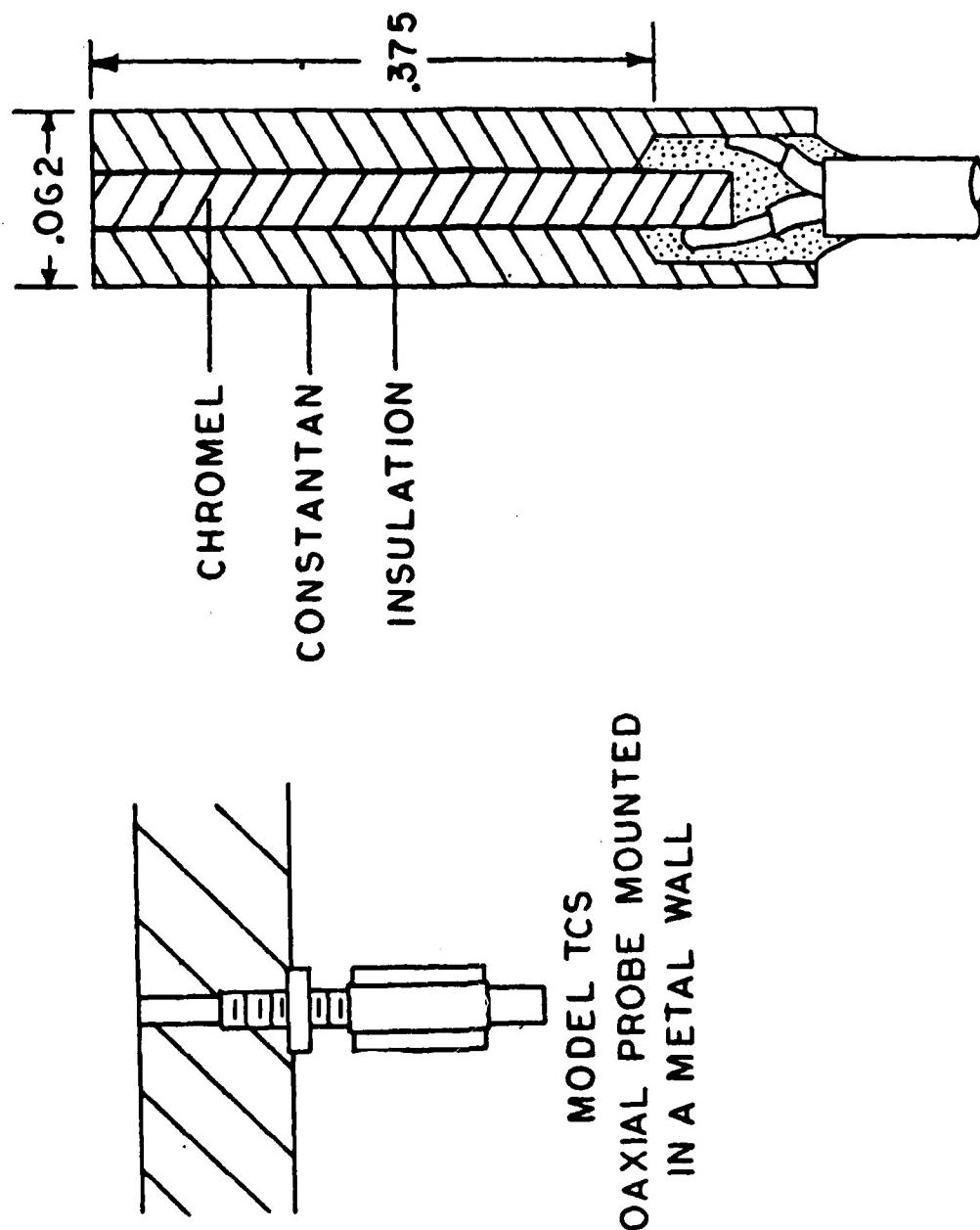


FIGURE 1.1 COAXIAL THERMOCOUPLE GAGE

where

$$G = P(t_n^-)H^T [HP(t_n^-)H^T + R_m]^{-1}$$

$$E = Y(t_n) - HU(t_n^-)$$

The updated sensitivities are calculated by,

$$S_k(t_n^+) = [I - GH]P(t_n^-)[I - GH]^T + GR_mG^T \quad (3-5)$$

The updated covariance is calculated by,

$$P(t_n^+) = [I - GH]P(t_n^-)[I - GH]^T + GR_mG^T \quad (3-6)$$

To alleviate the problem of imprecise initial conditions, a fixed point smoothing algorithm has been added to the HEATEST program. Details of the smoother and its effects in the adaptive estimation scheme may be found in Reference 6.

Finally, the best estimates of the parameters are then estimated at the end of a specified time segment by the gradient algorithm,

$$\theta^* = \theta - [\partial^2 F / \partial \theta^2]^{-1} \partial F / \partial \theta = \theta + J^{-1} S \quad (3-7)$$

where,

$$J_{i,j} = \sum_{n=1}^N S_{i,k}(t_n^-) H^T [HP(t_n^-)H^T + R_m]^{-1} H S_{j,k}(t_n^-)$$

$$S_k = \sum_{n=1}^N S_{i,k}(t_n^-) H^T [HP(t_n^-)H^T + R_m]^{-1} [Y(t_n) - HU(t_n^-)]$$

The matrix J is an approximation for the Jacobian or conditional information matrix and is given in component

form by $J_{i,k}$. The score vector, S_k , is used to approximate the gradient of the likelihood function for a large number of time samples.

Using these equations, best estimates for the temperature time history (states) at each node can be found, as well as the deviation in temperature as provided by the covariance matrix. Also, an estimate of the parameter uncertainty is provided by the Cramer-Rao bound. The Cramer-Rao bound relates the conditional information matrix to the covariance of the parameter estimate.

IV. RESULTS

4.1 Test Procedure

To test the validity of the program modifications, a set of contrived data was generated. It's development assumes that the heat rate due to convection at the surface node is constant and equal to the heat rate due to conduction at the surface. The heat rate due to convection is given by,

$$q = h(T_{aw} - T_w)$$

or

$$q = \bar{h} h_{ref}(T_{aw} - T_w) \quad (4-1)$$

where \bar{h} is defined as in Equation A-12. The equation for the heat rate due to conduction assuming a one-dimensional, homogenous, semi-infinite solid is as follows (Ref 9),

$$q = \frac{(\rho c k)^{1/2}}{\pi} \int_0^t \frac{dT_w(\tau)}{d\tau} \frac{d\tau}{(t-\tau)^{1/2}} \quad (4-2)$$

where t = time from start of heating
 $T(t)$ = surface temperature rise
 τ = dummy variable of integration

Equating Equations 4-1 and 4-2 yields,

$$q = h_{\text{bar}} h_{\text{ref}} (T_{\text{aw}} - T_w) = \frac{(\rho c k)^{1/2}}{\mu} \int \frac{dT_w(\tau) d\tau}{d\tau (t-\tau)^{1/2}}$$

(4-3)

Two different expressions for the derivative of the wall temperature with respect to time were used. The first implied a linear change in temperature with respect to time yielding a constant for dT_w/dt and the second expression assumes that temperature was a quadratic function of time as shown,

<u>Linear</u>	<u>Quadratic</u>
$T_w = bt + c$	$T_w = at^2 + bt + c$
$\frac{dT_w}{dt} = b$	$\frac{dT_w}{dt} = 2at + b$

Solving Equation 4-3 for T_{aw} so that h and q are constant and after making the indicated substitutions and integrating yields,

Linear assumption

$$T_{\text{aw}} = T_w + \frac{2b}{h_{\text{bar}} h_{\text{ref}}} \sqrt{\frac{\rho c k t}{\pi}}$$

(4-4)

Quadratic assumption

$$T_{\text{aw}} = T_w + \frac{2}{h_{\text{bar}} h_{\text{ref}}} \sqrt{\frac{\rho c k t}{\pi}} \left(b + \frac{4at}{3} \right)$$

(4-5)

A short computer program was written to produce a temperature-time history in the data tape format for the HEATEST program. For the above equations, h_{ref} and h_{bar} (the estimated parameter) were set equal to 1 and the coefficients a, b, and c, were selected to yield reasonable values for T_w .

4.2 Test Cases

The reference test case was taken to be a 10 sec. simulated wind tunnel test run using the linear data provided from the previous section. Thermocouple samples and test section samples were provided at the rate of one sample per second. The objective was to examine the rate of convergence of the temperature states and to estimate the first parameter, h_0 . Recall from Section 4.1 that the data was generated to yield a value of one for h_0 . Also of interest, was the validity of the model to the semi-infinite solid assumption (ie. no change in the temperature at the back face node throughout any specified time segment).

The input data is shown in Figure 4.1 and is generated digitally depending upon a desired time step (Δt). The initial temperature throughout each gage is assigned a value of 60°F. Figure 4.2 shows the input temperature values for a $\Delta t = 1$ sec. as used in the reference test case.

Figure 4.3 identifies the temperature state at the 2 sec.(lowest curve), 6 sec.(middle curve), and 10 sec.(top curve) times following the first iteration through the

updated HEATEST program. It clearly shows that at the back face node, the semi-infinite solid assumption used to derive the data is violated. This is indicated by the change in backface (node 6) temperature with time. Note also, however, that the temperature gradient at the back nodes (between nodes 5 and 6) is zero due to the adiabatic wall assumption. It should also be pointed out that surface node temperature response to the given input was immediate with no time lag.

Figure 4.4 is similar to Figure 4.3 except the temperature states and parameter estimates have been iterated to convergence, in this case, three times. Overlaying the two figures shows no perceptible difference between them, and the data shows no variations in values until after the second decimal point. In spite of the response of the back face node which would ordinarily invalidate the test, the estimated value for h_0 was .99598, within .4% of the desired value of 1! Several test cases will be compared to this reference by examining changes in time step, thermocouple length, and the number of node points. Also, an examination of the ability of the program to estimate the other parameters, follows.

4.2.1 Changes in Time Step

Figures 4.4, 4.5, and 4.6 show temperature states at 2, 6, and 10 sec. for Δt equal to 1, .5, and .25 sec., respectively. All three curves required three iterations

for convergence. The curves are all similar in shape with almost no perceptible differences. However, if the 10 sec. curves from $\Delta t = 1$ sec. and $\Delta t = .25$ sec. are overlayed as in Figure 4.7, a small difference may be noted at the backface nodes indicating that, indeed, a decrease in time step will yield a profile which will more closely approximate the model.

4.2.2 Changes in Thermocouple Length

Changes in thermocouple length offered the most dramatic changes in temperature state as can be seen in Figures 4.8, 4.9, and 4.10 where the lengths range from .1 ft., .05ft., and .025 ft., respectively. The two longer lengths converged within three iterations while the short thermocouple length took four iterations to converge. The extra iteration is most likely due to the large deviation from the model and the large differences in temperature state from one time step to the next. Figure 4.11 compares the temperature state of each thermocouple length after the 10 sec. run. The lowest curve is associated with the longest thermocouple, and the upper curve is associated with the short gage.

4.2.3 Changes in Number of Node Points

For this comparison, a 10 sec. run with 6 node points and a 3 sec. run with 6 node points (Figures 4.12 and 4.13) will be compared with a 10 sec. run with 12 node points and

a 3 sec. run with 12 node points (Figures 4.14 and 4.15). Each of the test runs were converged by the third iteration. At both of the different run times, increasing the number of nodes yielded a solution which more closely approximates the model (ie. the semi-infinite solid at the back face node) and gave correspondingly better estimates for h_0 . The comparison of temperature states is better represented by Figures 4.16 and 4.17 which directly compares 6 nodes and 12 nodes interspersed evenly throughout the .05 ft. long thermocouple for 10 sec. and 3 sec. run times, respectively.

4.2.4 Parameter Estimation

The ability of the algorithm to estimate h_0 has already been discussed. To summarize, even when the output temperature states clearly violate the semi-infinite solid assumption used in generating the data, the estimated values of h_0 remain within 15%. The 15% is a worst case number derived from the short thermocouple using a coarse grid for a long run time. An average deviation which considers all of the test cases evaluated is closer to 3%.

To estimate ϕ_k , an erroneous data value was given for K_{data} , wherin the program iterated to a value for ϕ_k which, when multiplied by the erroneous K_{data} , would yield the correct value, ie. $\phi_k K_{data} = K_{correct}$. The erroneous K_{data} which was input into the program was 14% in error of the $K_{correct}$ value. The value provided by the program for ϕ_k when multiplied by K_{data} yielded a value within .9% of

the ~~K~~_{correct} value!

The estimate of ϕ_c was not as successful, however. After 6 iterations, the solution was diverging from the expected value. A suspected sign error in the ϕ_c sensitivity calculation is the most likely cause.

The quadratic input data was used as a comparison to the linear data to challenge the algorithm, ie. the more complicated the input the more difficult the estimation process. No direct comparison may be made of temperature, however, since the input data is different. The parameter estimation of h_g using the quadratic input data was still excellent yielding .1%, while the estimate using linear data was somewhat better at .05%.

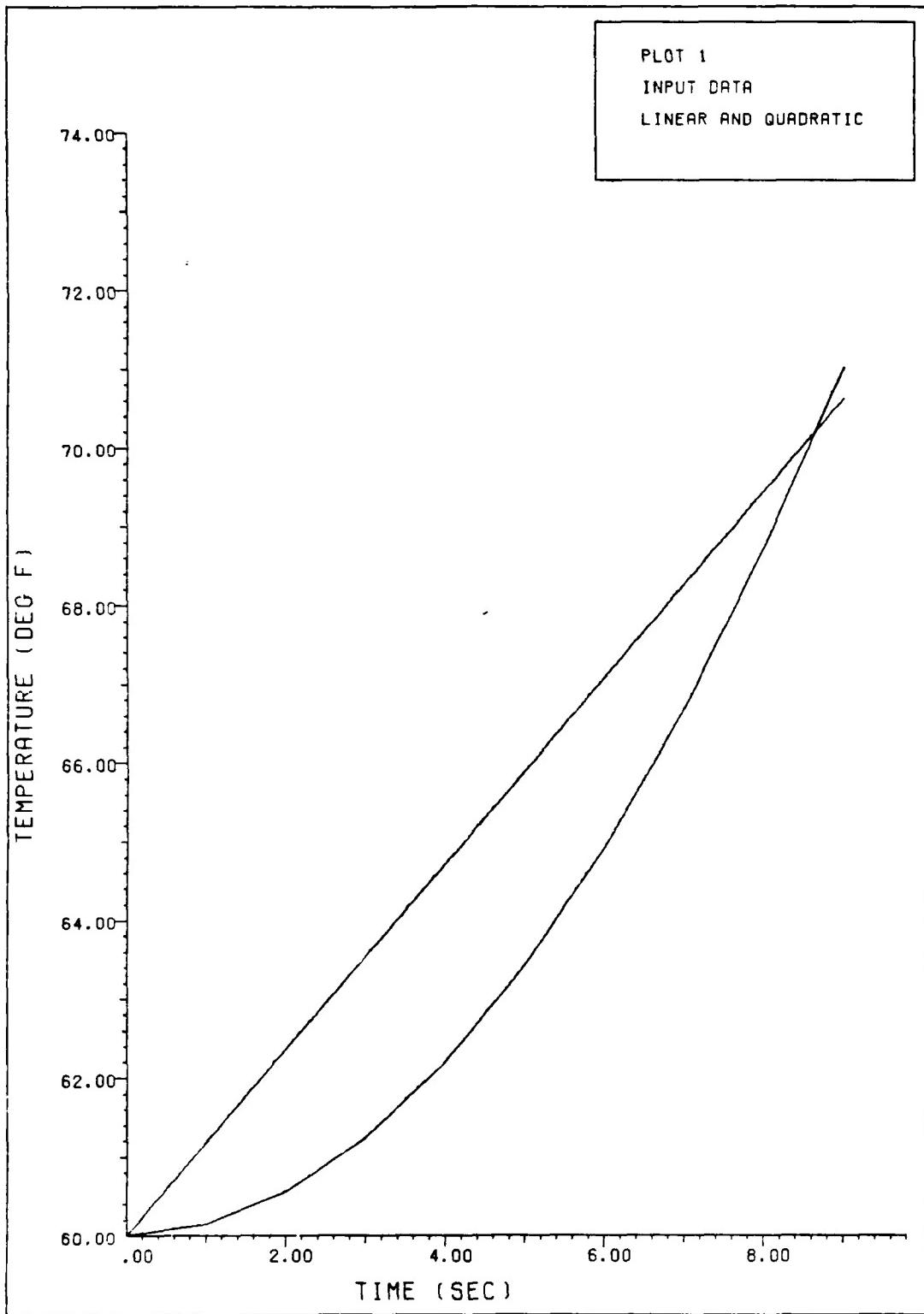


FIGURE 4.1

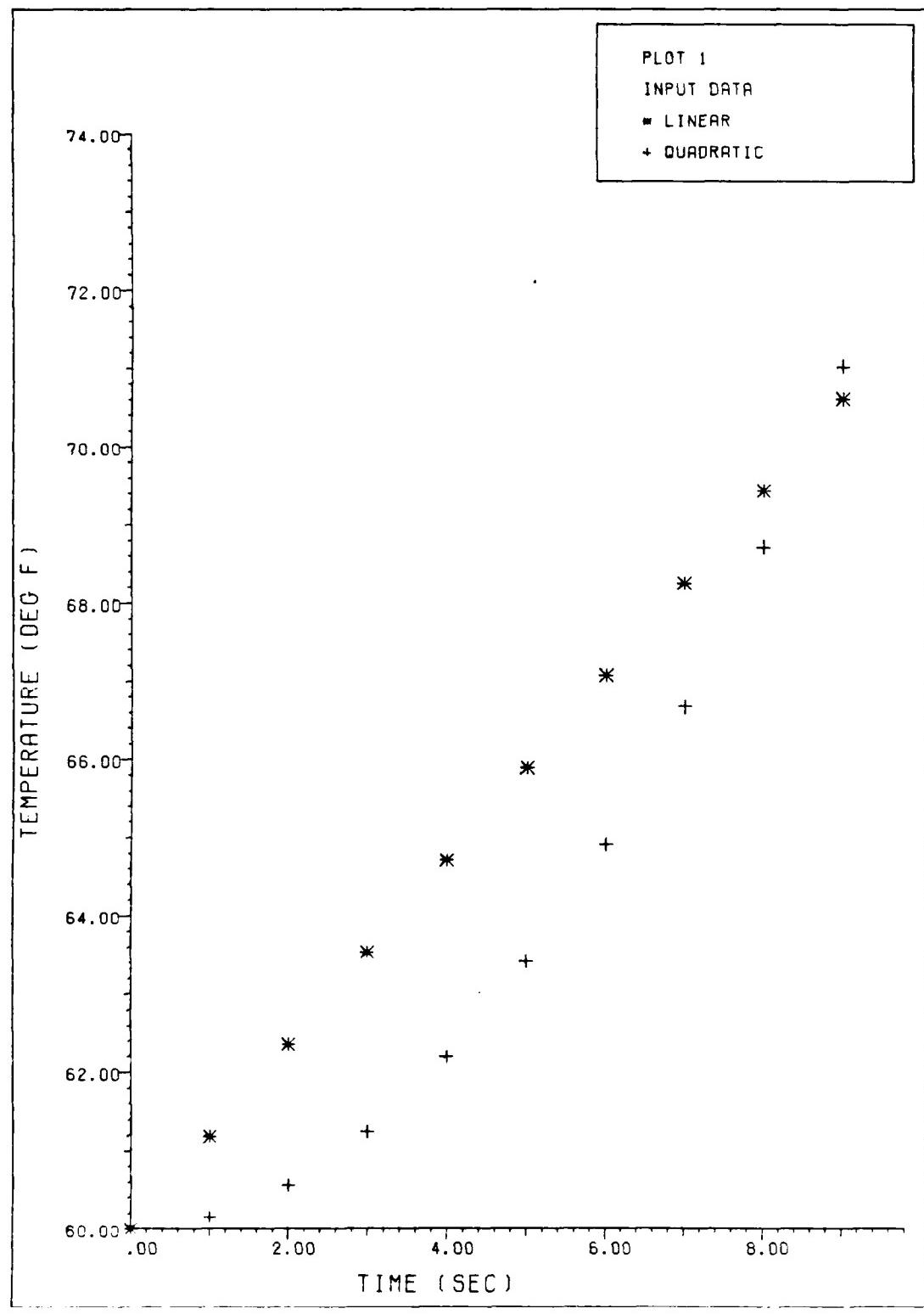


FIGURE 4.2

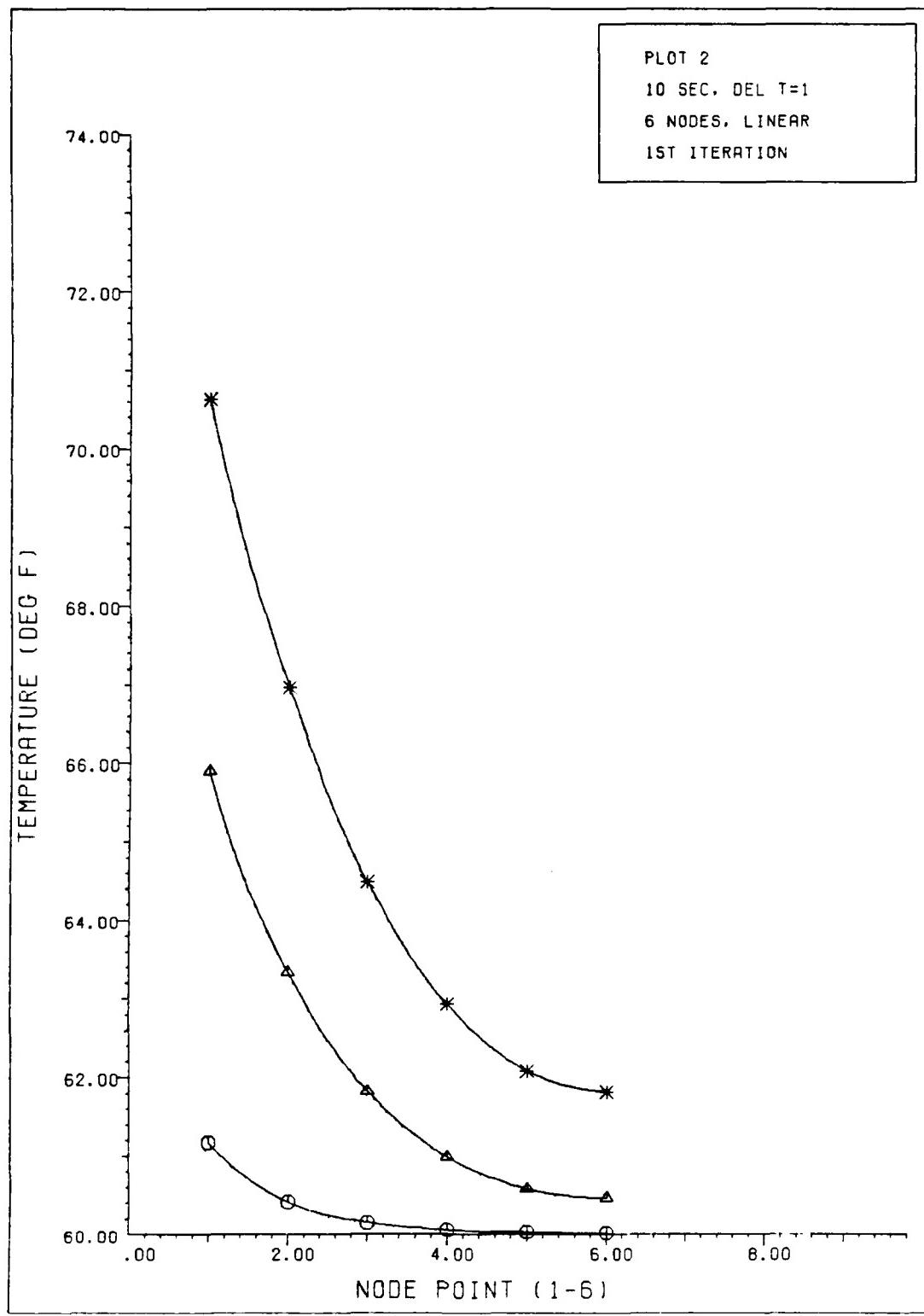


FIGURE 4.3 TEMP VS NODE POINT HISTORY (2.6.10 SEC)

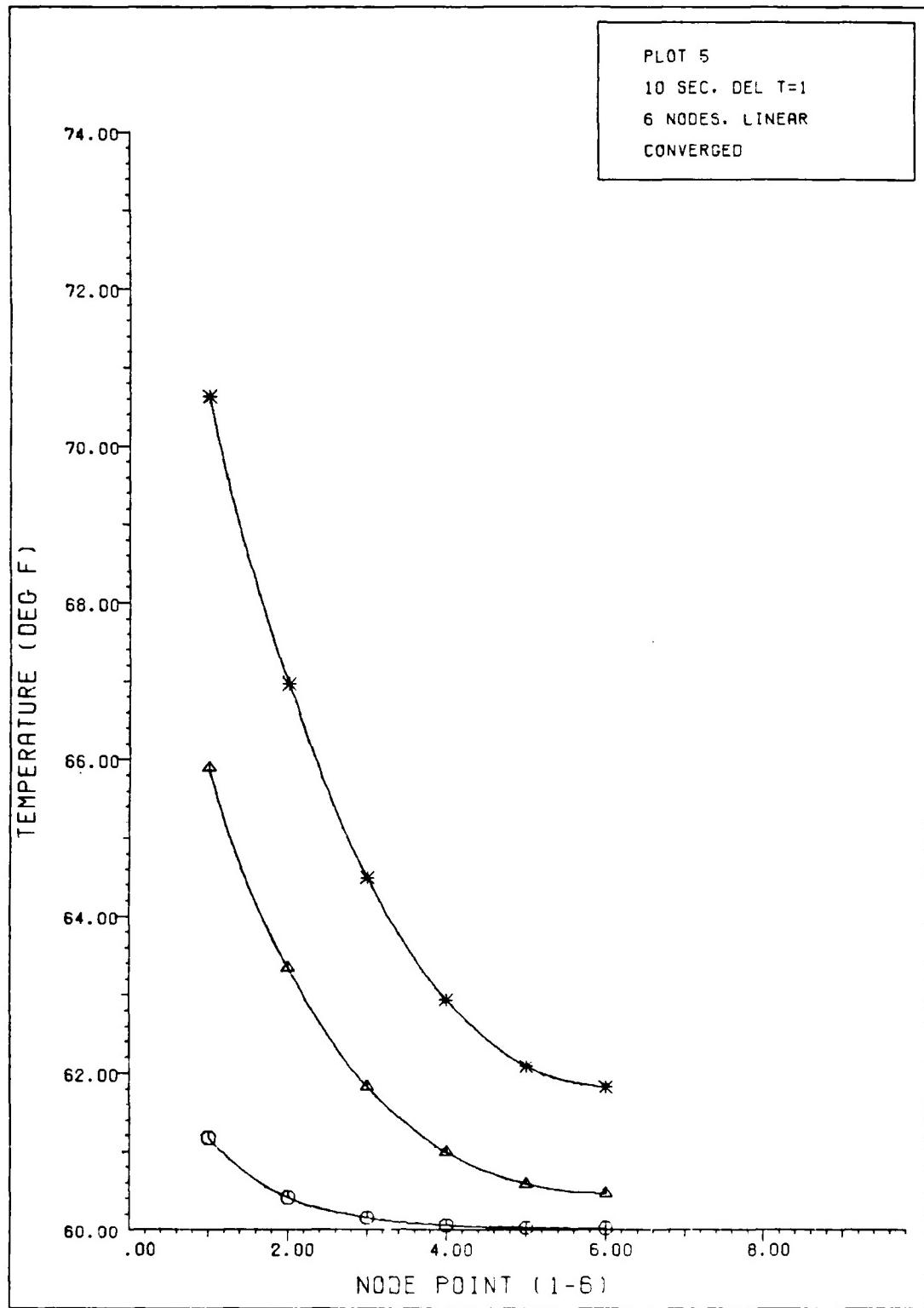


FIGURE 4.4 TEMP VS NODE POINT HISTORY (2,5,10 SEC)

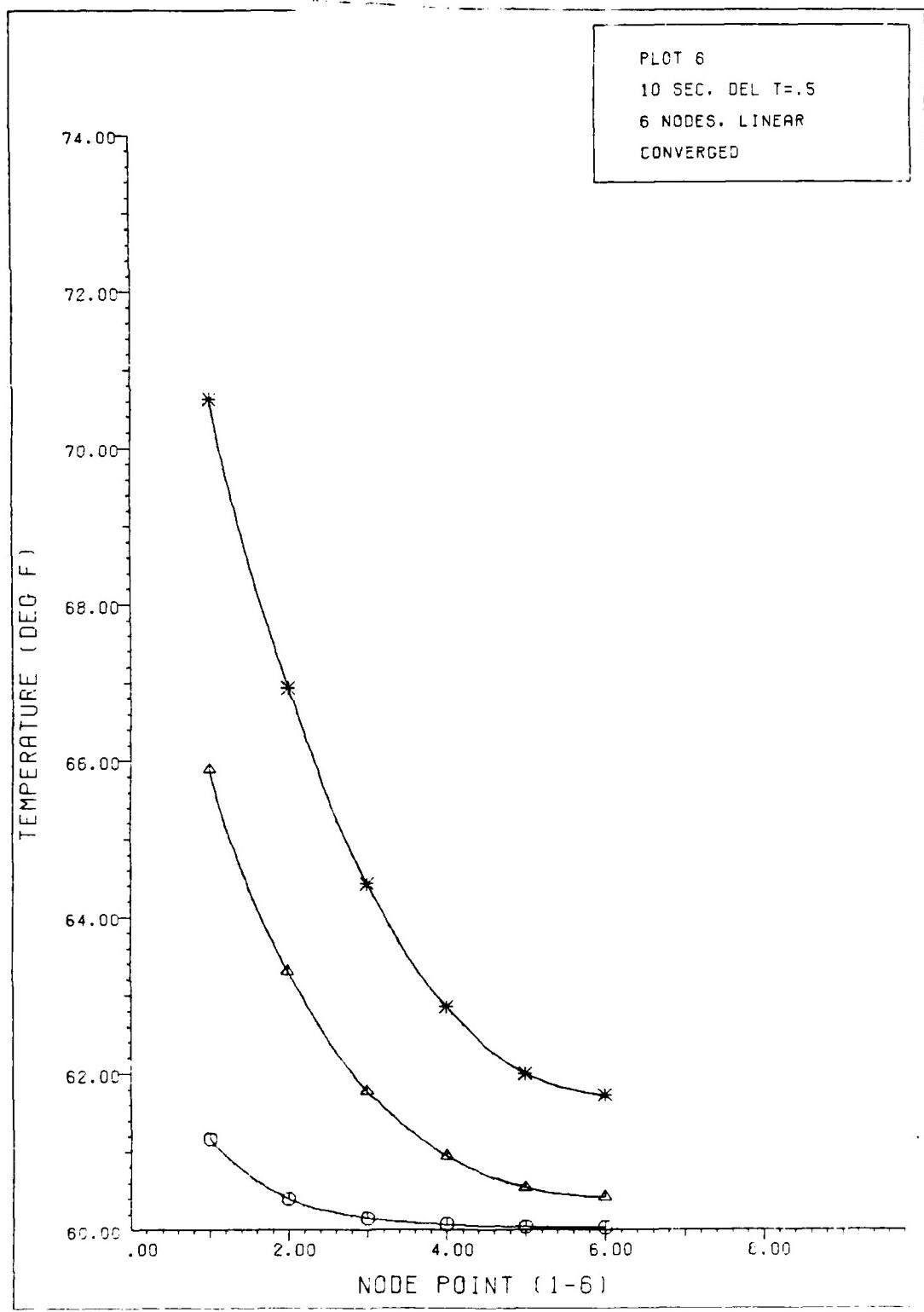


FIGURE 4.5 TEMP VS NODE POINT HISTORY (2.6.10 SEC)

the utility of the program to be used in wind tunnel runs of much longer duration.

5.2 Recommendations

Actual wind tunnel test data from coaxial gages needs to be analyzed by the program to instill more confidence in the results. This would require that the data tape format be modified to be compatible with the program inputs.

Also, a prescribed model for $h_{\bar{a}}$ as a function of angle of attack needs to be input to determine the ability of the program to estimate the piecewise linear derivatives, h_{a1} and h_{a2} . The additional input data would then be the wind tunnel model angle of attack at each thermocouple sample time.

Another potential modification to the program would be to use a second order time derivative approximation as opposed to the current first order approximation. It is suspected that the increased accuracy would improve the state estimates particularly for large time gaps in thermocouple data.

The temperature state estimates might be improved by incorporating a variable grid. An exponential grid generation scheme would concentrate node points near the surface where the largest temperature gradients exist.

A final recommendation, would be to make the program more 'user friendly' and to publish documentation much like a users manual.

V. CONCLUSIONS AND RECOMMENDATIONS

5.1 Conclusions

The program works for this simplified test case of known parameters and number of node points. The ability of the program to determine the temperature state appeared very good but definite conclusions cannot be drawn without comparing this data to actual thermocouple data. The ability of the program to estimate the parameters, h_0 and ϕ_k was demonstrated extremely well. Re-examination of the ϕ_c equations must be made before further conclusions may be made about the ability of the program to estimate this parameter. It is theoretically possible, however.

The objective, to validate the use of the modified HEATEST program for use of coaxial thermocouple gages on wind tunnel test articles, has been met for the special case of this analytical model. The program is capable of determining temperature states and estimating parameters with a high degree of accuracy. A note concerning the semi-infinite slab assumption needs to be emphasized. This assumption was made only for the data generation program to yield data for which the analytical solution was known. As mentioned in Chapter 1, some coaxial gages are available with backface temperature monitoring which would also provide another temperature measurement to enhance estimation of ϕ_k and ϕ_c . This feature would greatly enhance

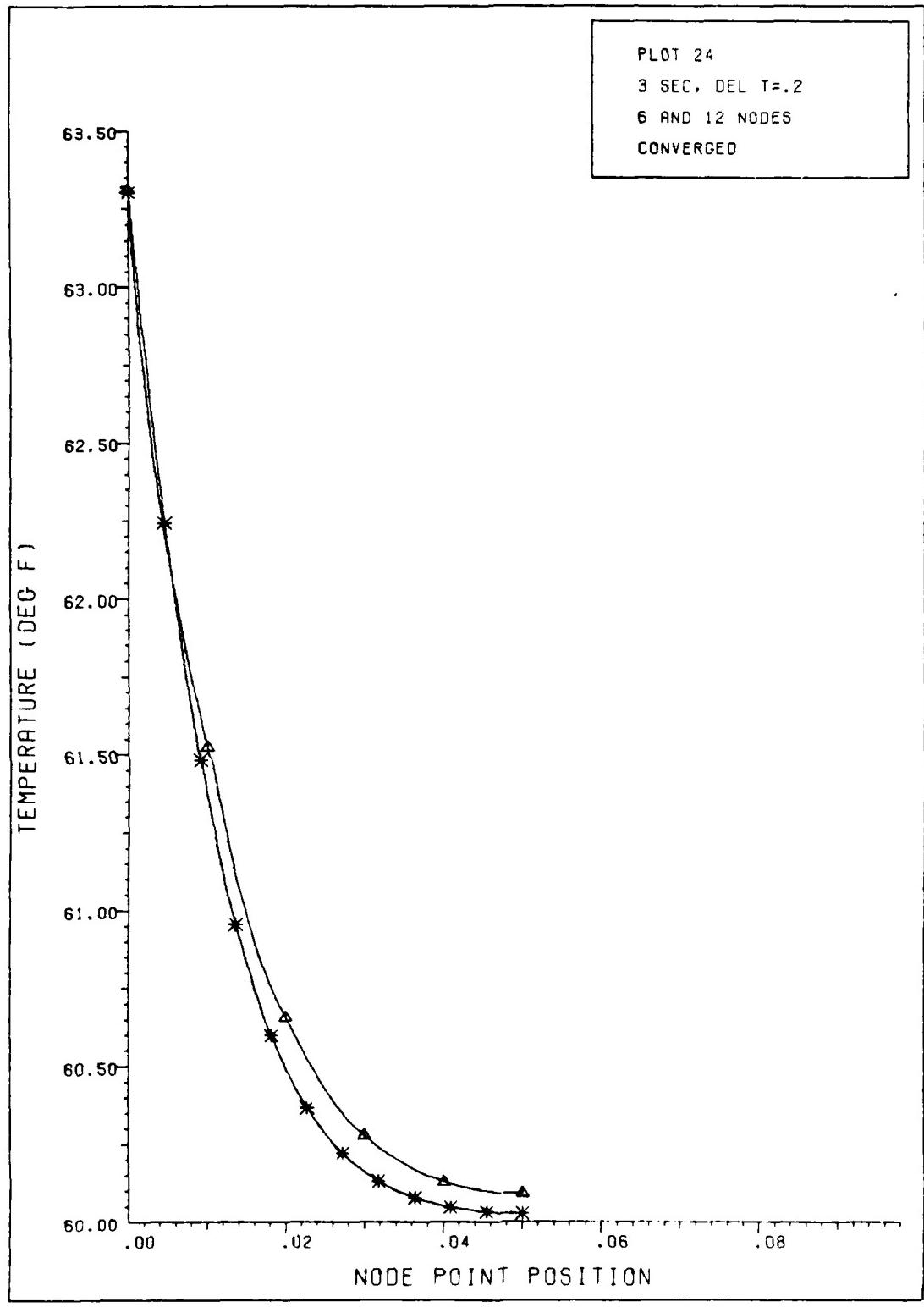


FIGURE 4.17 TEMP VS NODE POS.(16 NODES VS 12 NODES)

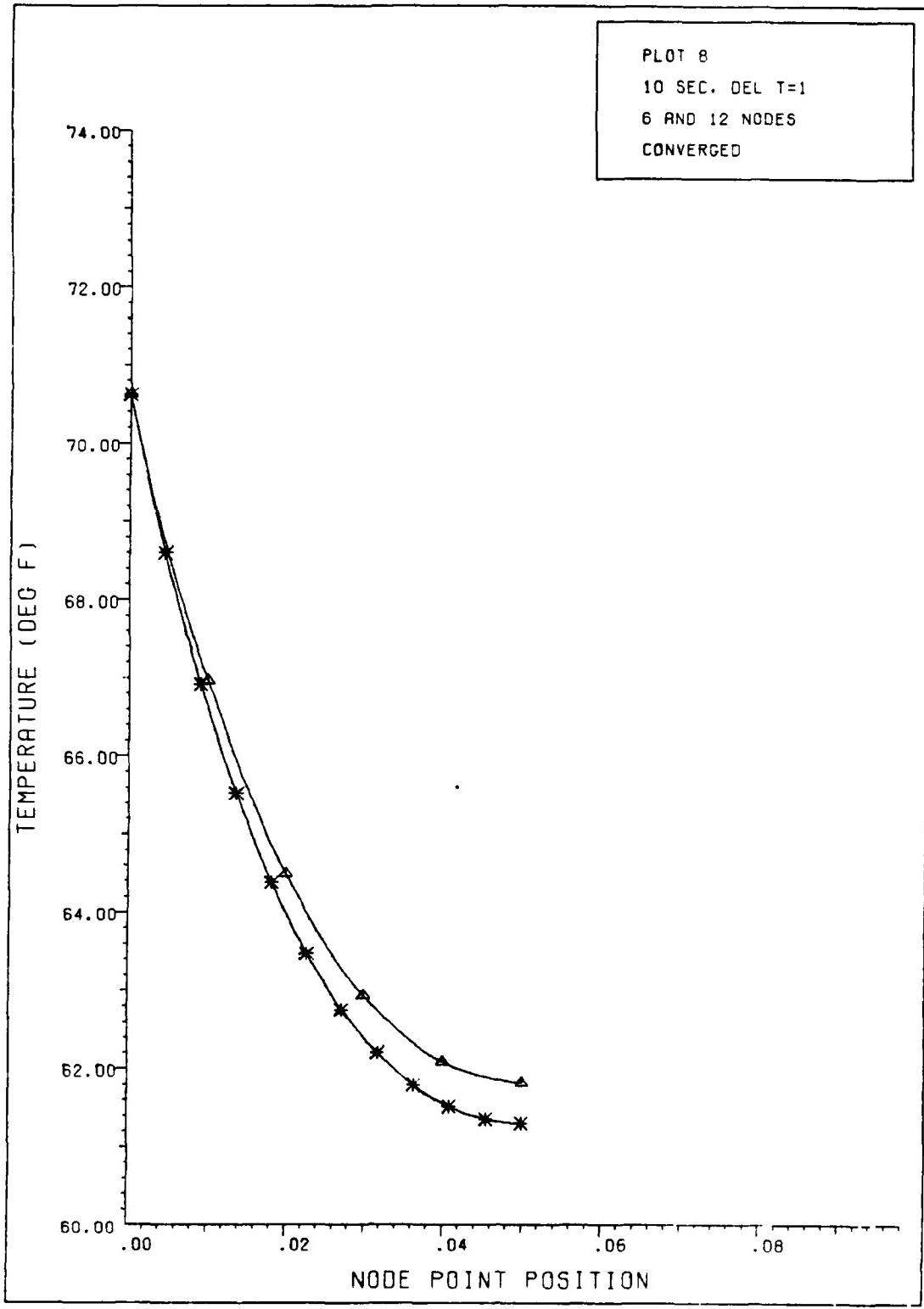


FIGURE 4.16 TEMP VS NODE POS.(6 NODES VS 12 NODES)

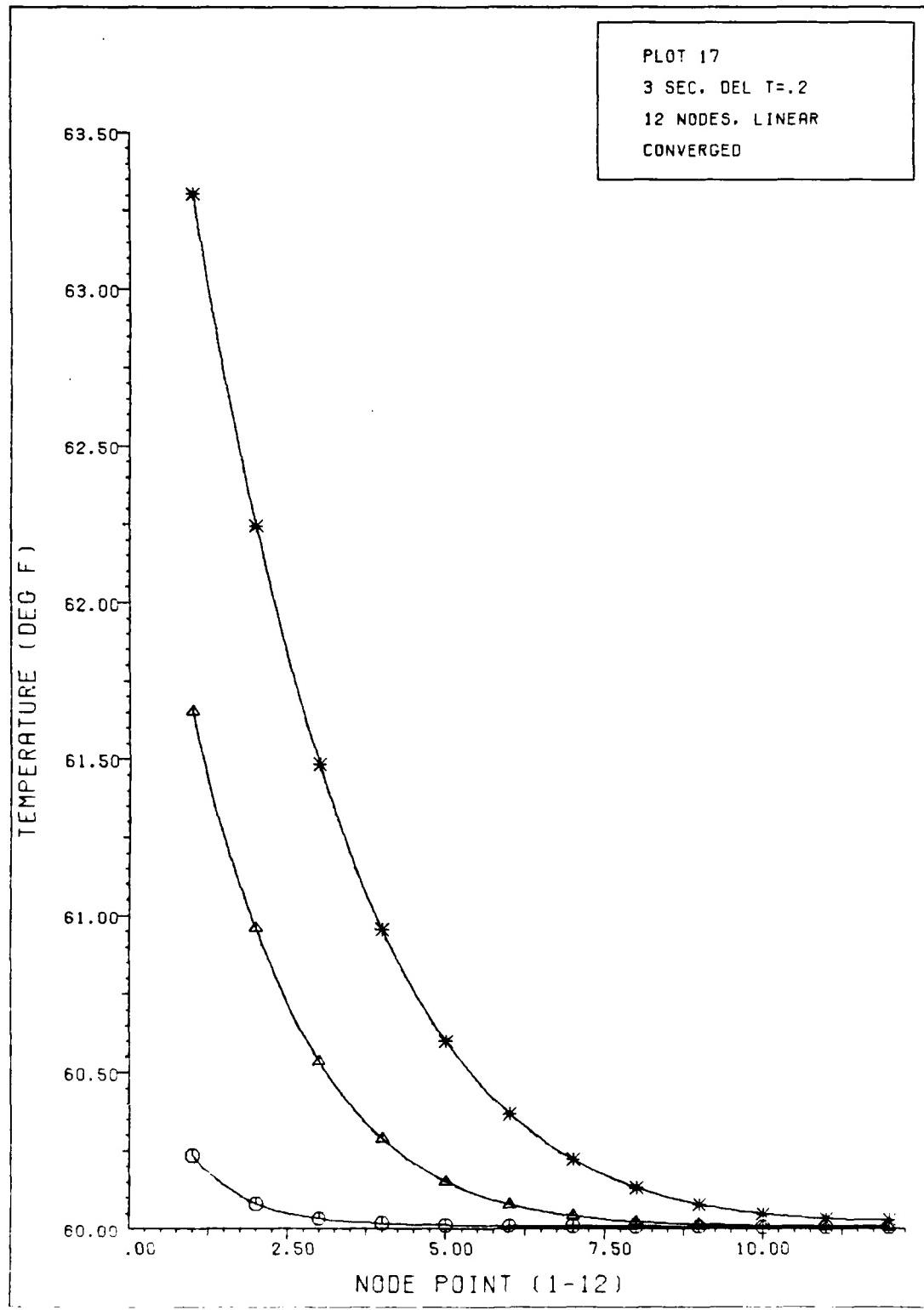


FIGURE 4.15 TEMP VS NODE POINT (.4, 1.6, 3.0 SEC.)

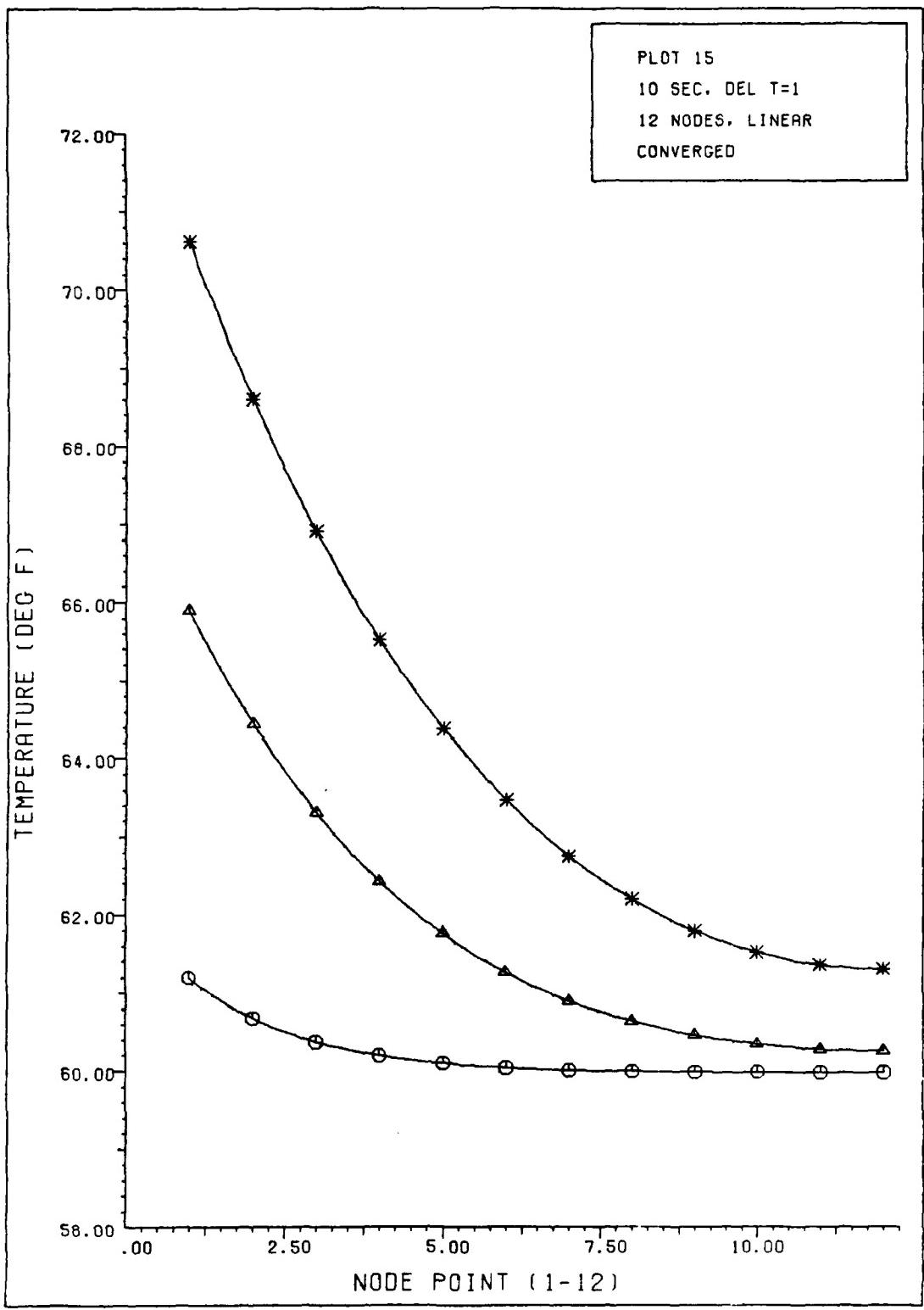


FIGURE 4.14 TEMP VS NODE POINT HISTORY(2.6,10 SEC)

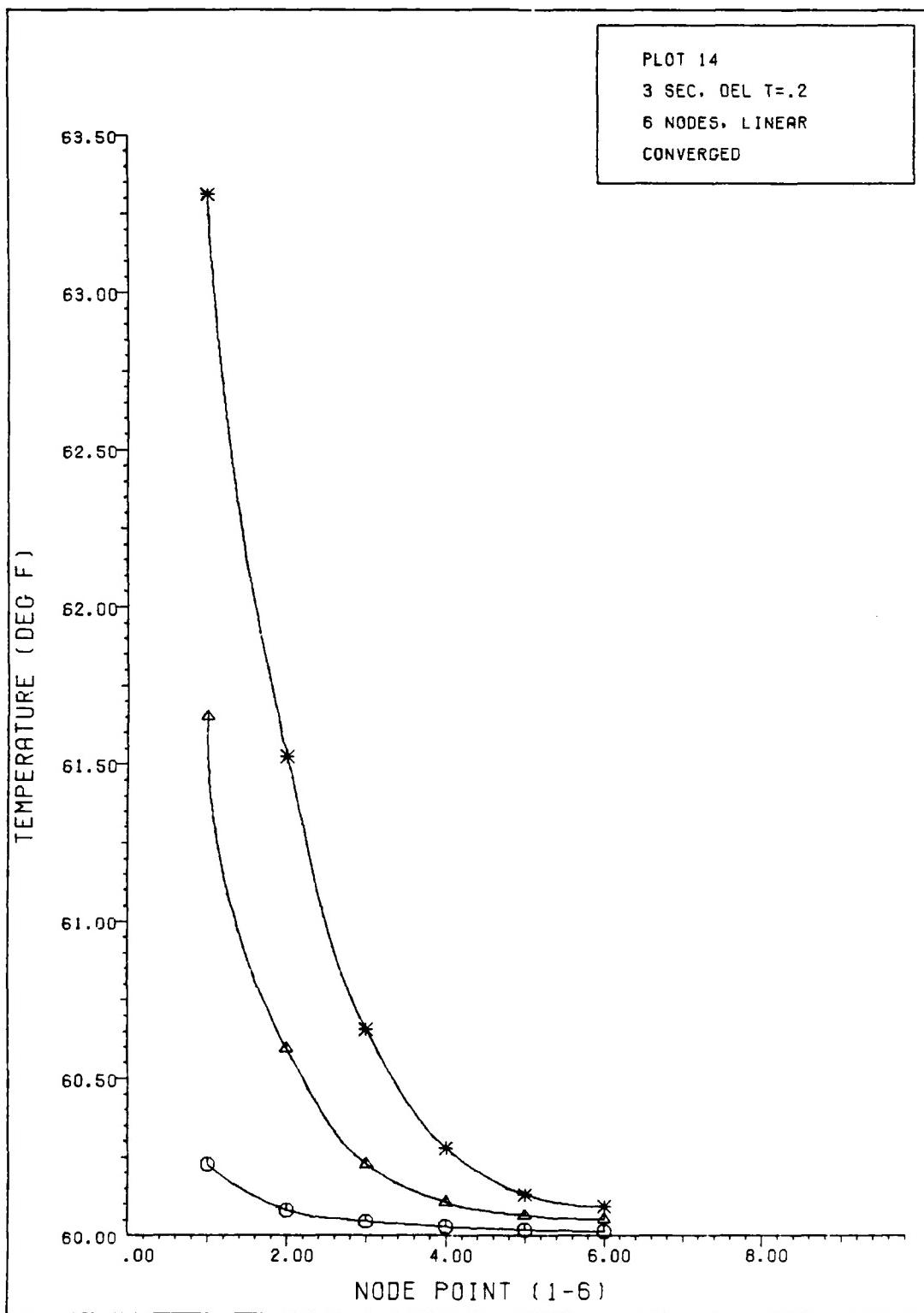


FIGURE 4.13 TEMP VS NODE POINT (.4. 1.6, 3.0 SEC)

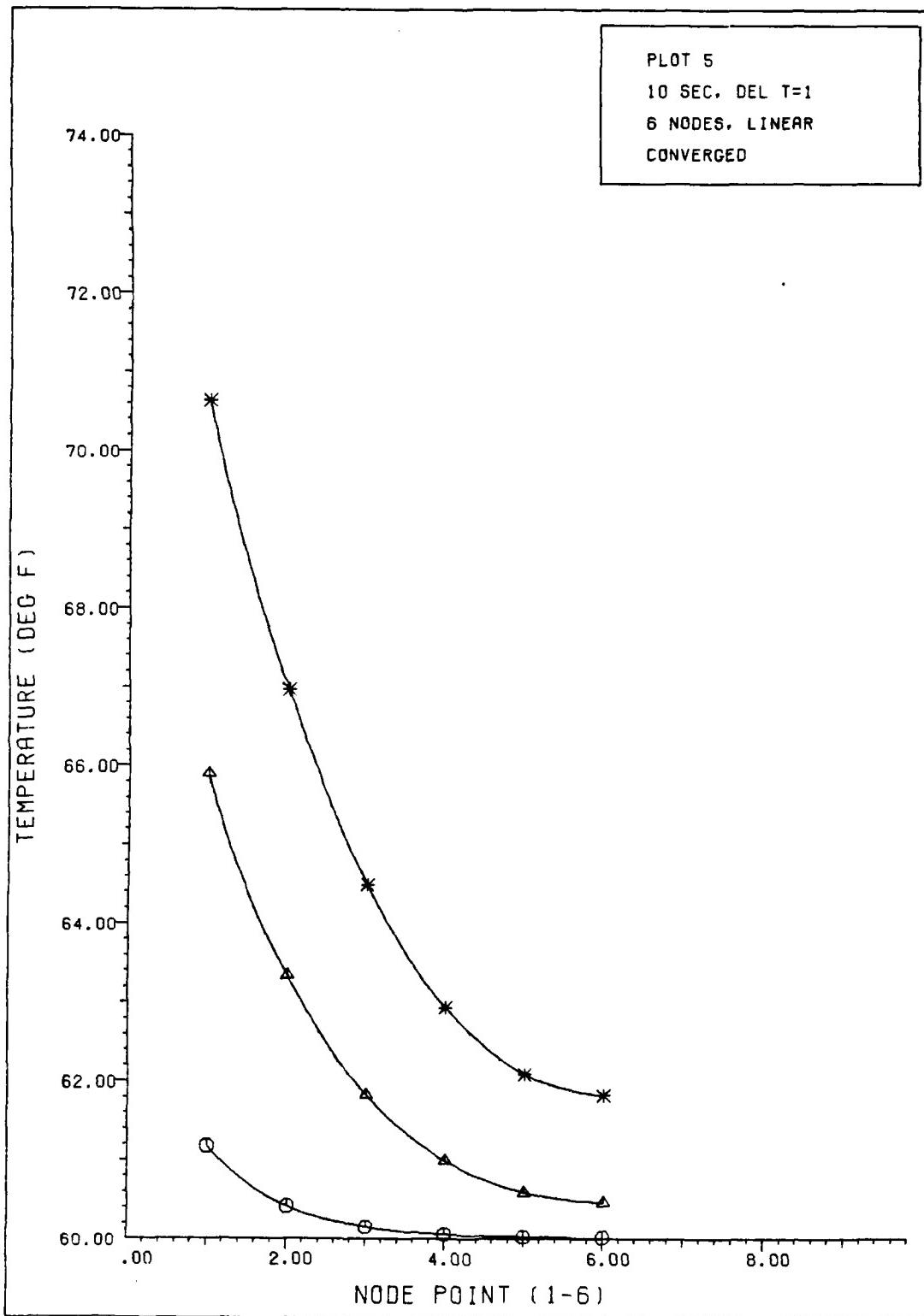


FIGURE 4.12 TEMP VS NODE POINT HISTORY(2.6.10 SEC)

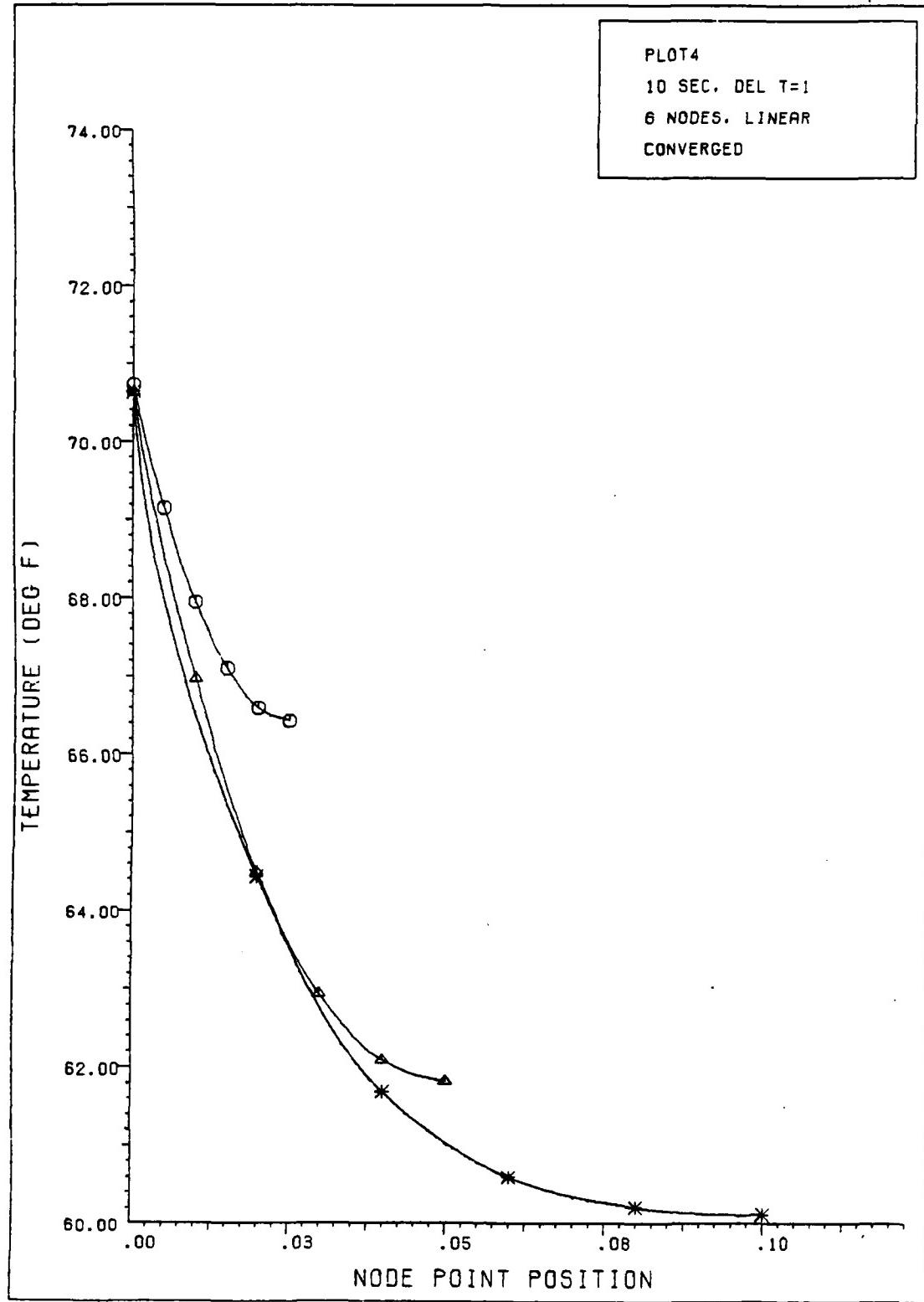


FIGURE 4.11 TEMP VS POS. (T/C LENGTH=.1,.05,.025FT)

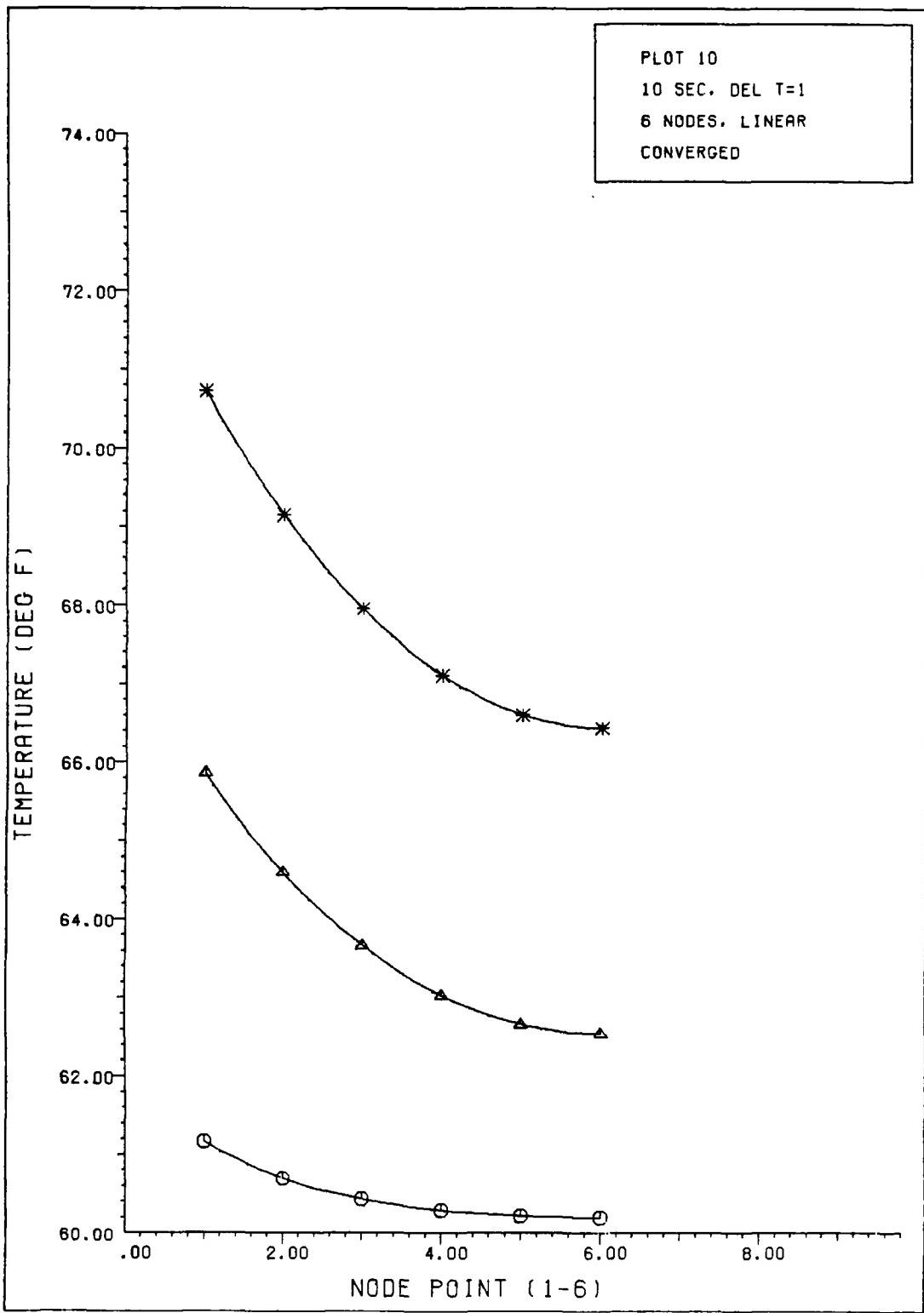


FIGURE 4.10 TEMP VS NODE POINT (T/C LENGTH=.025FT)

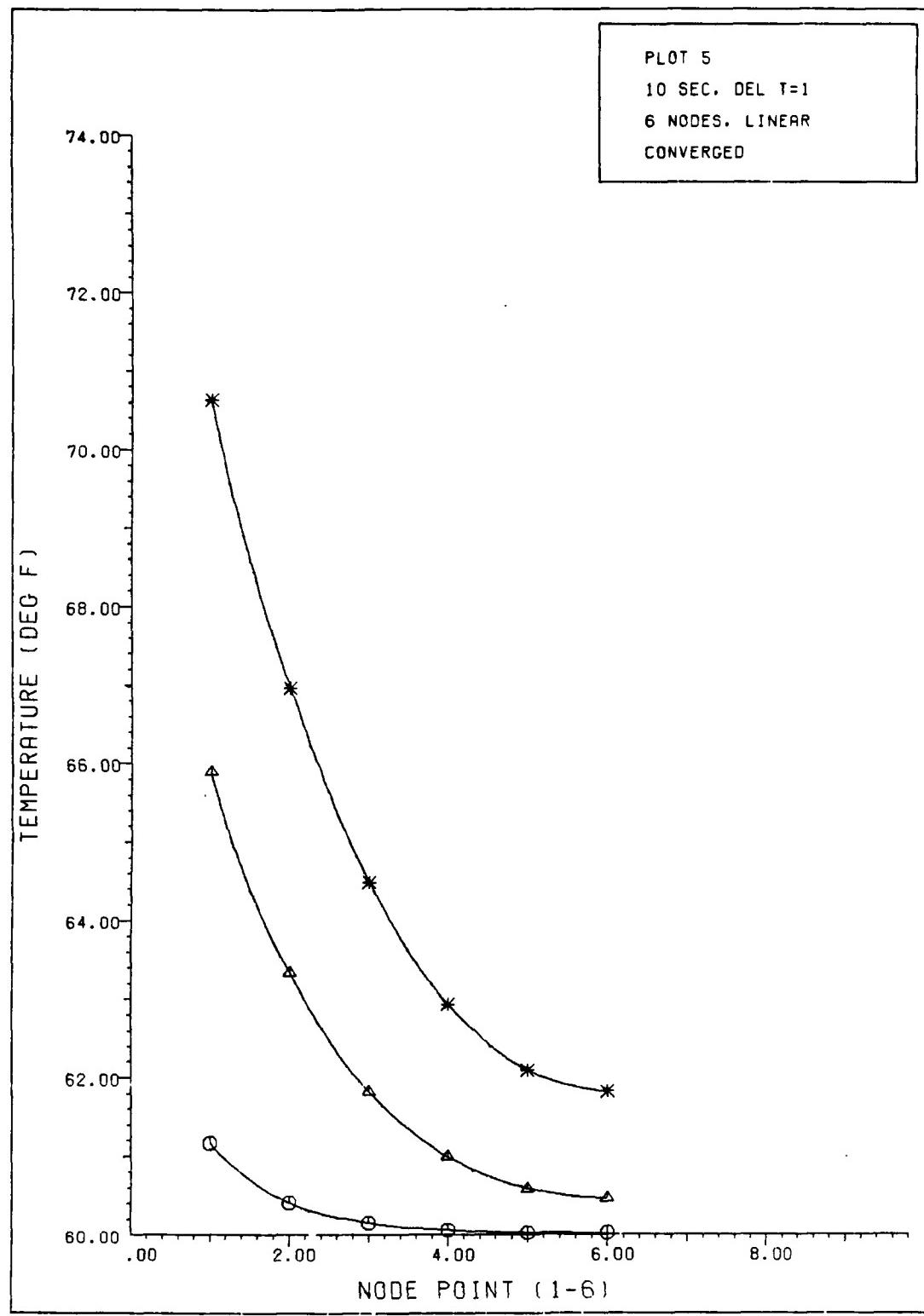


FIGURE 4.9 TEMP VS NODE POINT (T/C LENGTH=.05 FT)

PLOT 12
10 SEC. DEL T=1
6 NODES, LINEAR
T/C LENGTH=.1 FT.

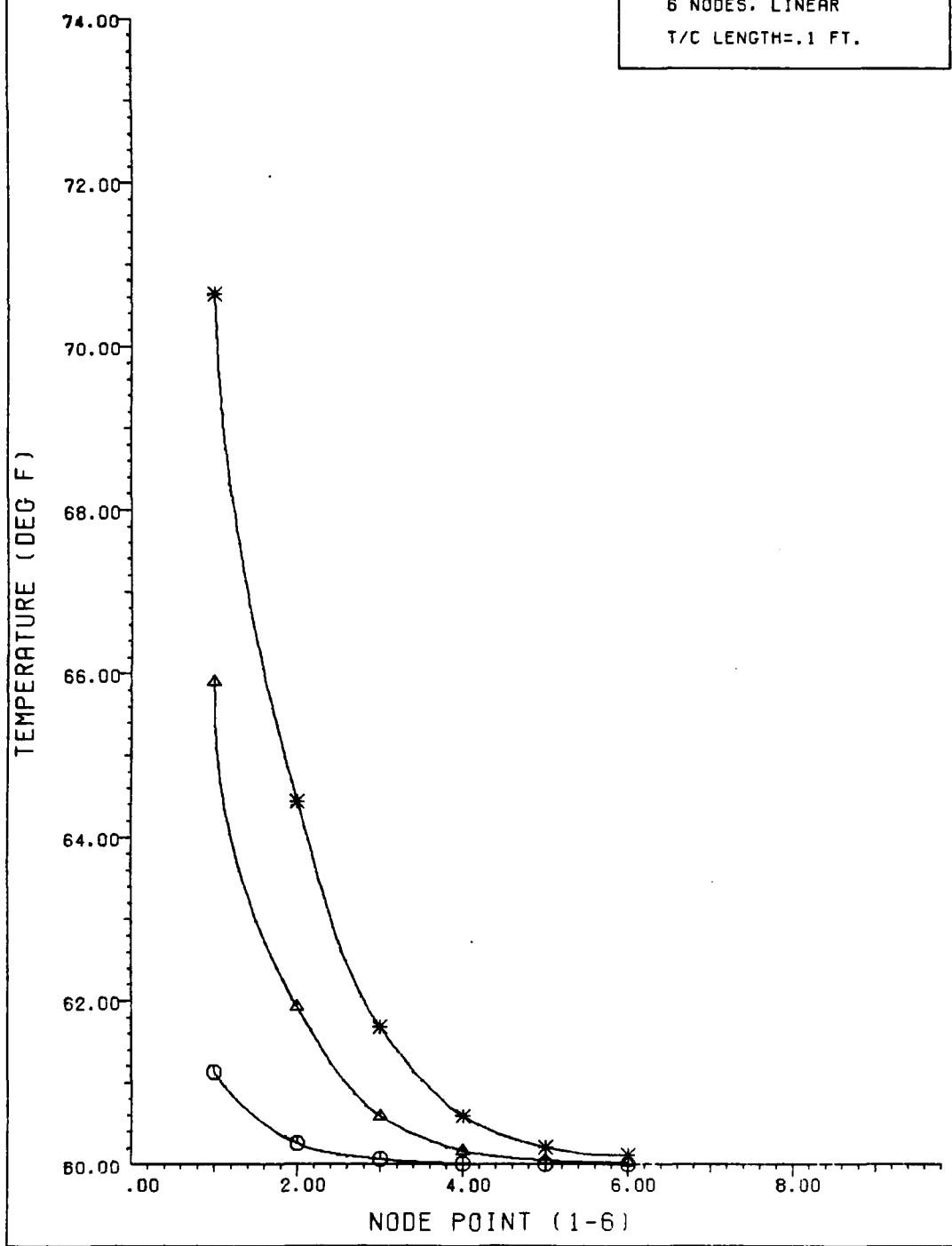


FIGURE 4.8 TEMP VS NODE POINT (T/C LENGTH= .1 FT.)

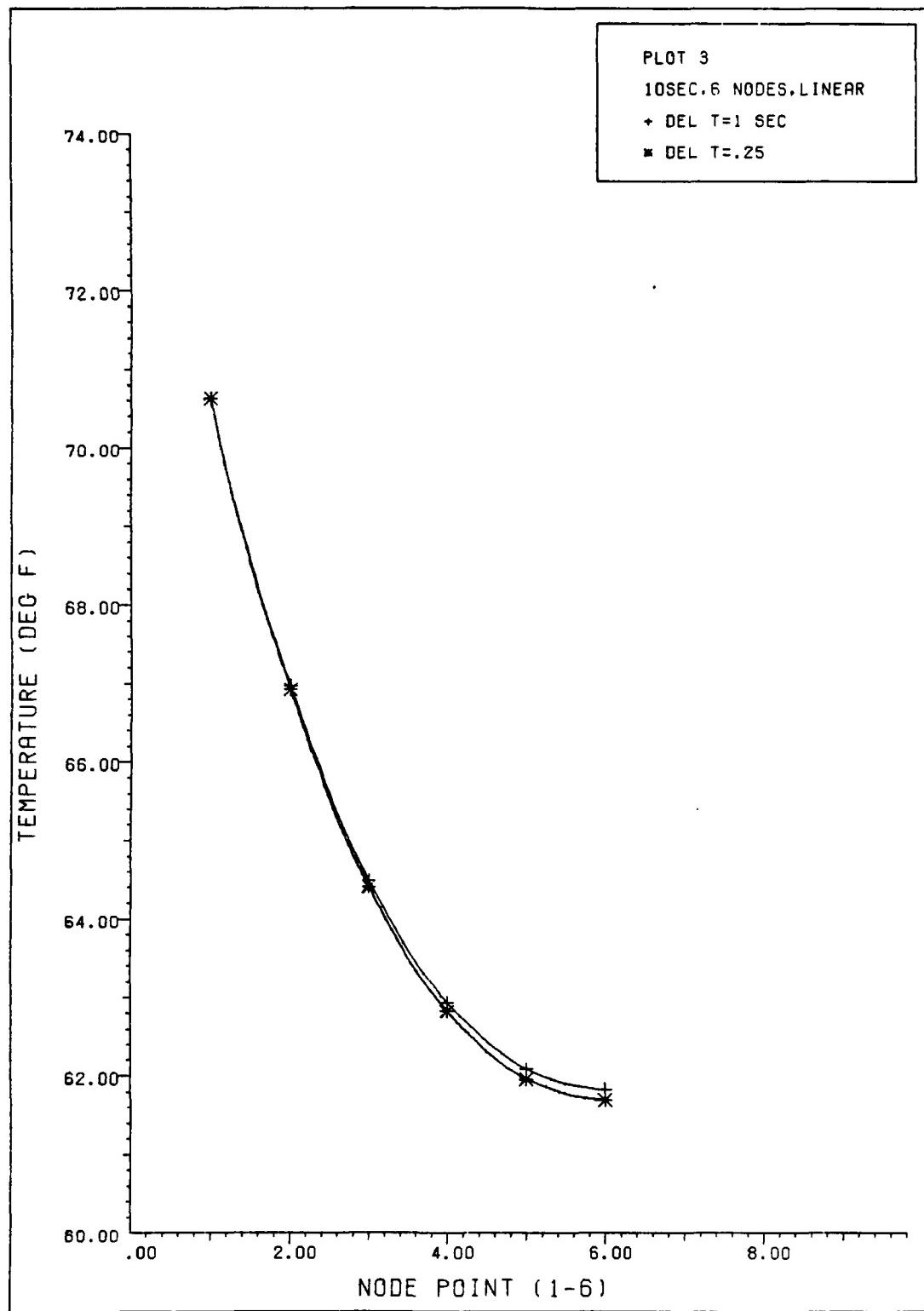


FIGURE 4.7 TEMP VS NODE POINT(DEL T=1 AND .25 SEC)

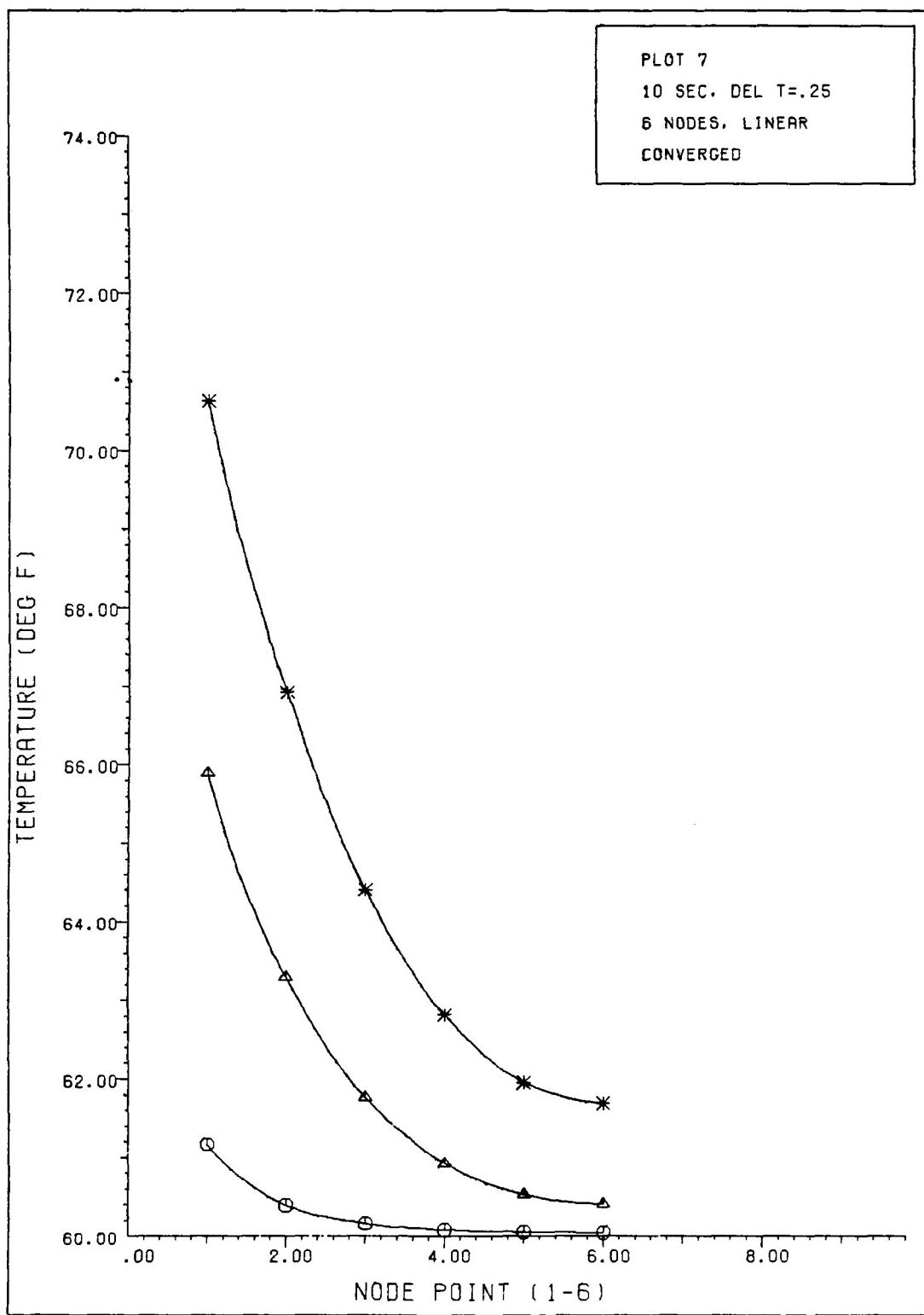
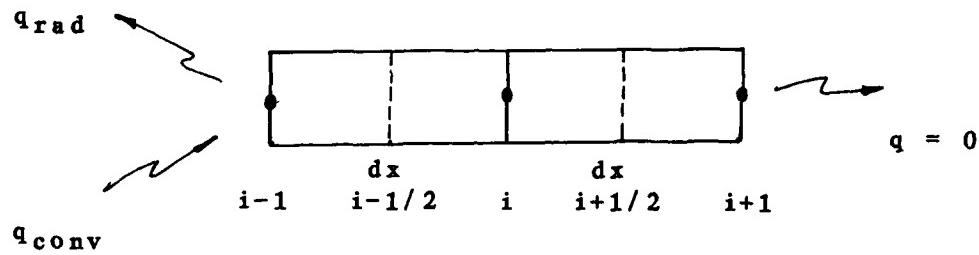


FIGURE 4.6 TEMP VS NODE POINT HISTORY (2.6.10 SEC)

APPENDIX A

Derivation of equations using a one-dimensional energy balance formulation are given as follows,



$$\text{Energy in the left face} = -k \frac{\partial T}{\partial x} = q_1$$

$$\text{Energy generated within the element} = q dx = 0$$

$$\text{Change in internal energy} = \rho c \left(\frac{\partial T}{\partial \tau} \right) dx$$

$$\begin{aligned} \text{Energy out of right face} &= -k \left(\frac{\partial T}{\partial x} \right)_{x+dx} \\ &= -\left[k \frac{\partial T}{\partial x} + \frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) dx \right] \end{aligned}$$

Then, combining the above and using Fourier's Law of Heat Conduction, ie,

$$\begin{aligned} \text{energy in + energy within left face} &= \text{change in internal energy} \\ \text{the element} &+ \text{energy out right face} \\ \text{yields,} & \end{aligned}$$

$$-\frac{k \partial T}{\partial x} + q dx = \rho c \frac{\partial T}{\partial \tau} dx - \left[k \frac{\partial T}{\partial x} + \frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) dx \right] \quad (A-1)$$

or,

$$\rho c \frac{\partial T}{\partial \tau} = \frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) \quad (A-2)$$

or, replacing T by U,

$$\rho c U_t = (k U_x)_x \quad (A-3)$$

or,

$$\rho c U_t = \left[\frac{k_{i-1/2}}{\Delta x_{i-1/2}} (U_{i-1}^n - U_i^n) - \frac{k_{i+1/2}}{\Delta x_{i+1/2}} (U_i^n - U_{i+1}^n) \right] \frac{1}{\Delta x} \quad (A-4)$$

where ΔX may be written as,

$$\Delta x = \frac{\Delta x_{i+1/2} + \Delta x_{i-1/2}}{2}$$

then, writing the time gradient in first order backward difference form and expanding yields,

$$\begin{aligned} \rho c \frac{U_i^n - U_{i-1}^n}{\Delta t} &= \frac{2k_{i-1/2}}{\Delta x_{i-1/2}(\Delta x_{i+1/2} + \Delta x_{i-1/2})} U_{i-1}^n \\ &- \frac{2k_{i-1/2}}{\Delta x_{i-1/2}(\Delta x_{i+1/2} + \Delta x_{i-1/2})} U_i^n \\ &- \frac{2k_{i+1/2}}{\Delta x_{i+1/2}(\Delta x_{i+1/2} + \Delta x_{i-1/2})} U_i^n \\ &+ \frac{2k_{i+1/2}}{\Delta x_{i+1/2}(\Delta x_{i+1/2} + \Delta x_{i-1/2})} U_{i+1}^n \end{aligned} \quad (A-5)$$

then, specifying equal spacing for each node point,

$$\Delta x_{i+1/2} = \Delta x_{i-1/2}$$

and the equation becomes,

$$\rho c \frac{U_i^n - U_i^{n-1}}{\Delta t} = \frac{k_{i-1/2}}{\Delta x^2} U_{i-1}^n - \left(\frac{k_{i-1/2}}{\Delta x^2} + \frac{k_{i+1/2}}{\Delta x^2} \right) U_i^n + \frac{k_{i+1/2}}{\Delta x^2} U_{i+1}^n$$

(A-6)

Now, instead of estimating c and k directly, define two scaling parameters ϕ_c and ϕ_k such that c and k will remain constant. These two parameters are estimated by the HEATEST program.

$$\rho \phi_c c \Delta x \frac{U_i^n - U_i^{n-1}}{\Delta t} = \phi_k \frac{k_{i-1/2}}{\Delta x} U_{i-1}^n - \phi_k \left(\frac{k_{i-1/2}}{\Delta x} + \frac{k_{i+1/2}}{\Delta x} \right) U_i^n + \phi_k \frac{k_{i+1/2}}{\Delta x} U_{i+1}^n$$

(A-7)

This equation is applicable at all interior ($i \neq 1, i \neq i_{\max}$) points.

For the back face, assuming a semi-infinite solid, $i = i_{\max}$ and the equation becomes,

$$\frac{\rho \phi_c c \Delta x}{2} \frac{U_L^n - U_L^{n-1}}{\Delta t} = \frac{\phi_k k_{L-1/2}}{\Delta x} U_{L-1}^n - \frac{\phi_k k_{L-1/2}}{\Delta x} U_L^n$$

(A-8)

For the front face, ($i = 1$), the effects of radiation away from and convection toward the solid surface must be accounted for.

The radiation is modeled using the Stefan-Boltzmann Law,

$$q = \epsilon\sigma(U_1^4 - U_{\infty}^4) \quad (A-9)$$

where ϵ radiative emissivity
 σ Stefan-Boltzmann constant
 U Temperature ($^{\circ}$ R)

The convective transfer of energy is modeled using Newton's Law of Cooling,

$$q = h(T_{aw} - T_w) \quad (A-10)$$

Non-dimensionalizing by a reference heat transfer coefficient, h_{ref} yields,

$$q = \bar{h}h_{ref}(T_{aw} - U_1) \quad (A-11)$$

where, \bar{h} = convective heat transfer coefficient ratio
 T_{aw} = adiabatic wall temp of test article

The dependence of the heat transfer coefficient on parameters other than those included in the reference heat transfer coefficient are summarized by the static transfer relation or heat transfer coefficient ratio, h/h_{ref} . Here, the ratio is assumed to be piecewise linear with respect to angle of attack as derived from Lagrange Interpolation Theory (Ref 6),

$$h_{bar} = h/h_{ref} = [h_0 + h_{a1}(a-a_1) + h_{a2}(a-a_2)] \quad (A-12)$$

where h_0 is the magnitude of the heat transfer coefficient, h , at the reference condition specified at a_1 . Combining Equations A-7, A-9, A-11, and A-12 and evaluating at node one yields,

$$\begin{aligned} \frac{\rho \phi_c c \Delta x}{2} \frac{U_1^n - U_1^{n-1}}{\Delta t} &= -\frac{\phi_k k_{k+1/2}}{\Delta x} U_1^n \\ &+ \frac{\phi_k k_{1+1/2}}{\Delta x} U_2^n - \varepsilon \sigma [(U_1^n)^4 - (U_{\infty}^n)^4] \\ &+ [h_0 + h_{a1}(a-a_1) + h_{a2}(a-a_2)] h_{ref} (T_{aw} - U_1^n) \end{aligned} \quad (A-13)$$

Using the quasi-linearization as developed in Equation 2-6, the resultant form for determining the temperature time history at each node point is given in Equations 2-7 and 2-8.

The matrix form for the equations may be found after defining the following.

$$\begin{aligned} RCX_i &= \rho \phi_c c \Delta x & RCX_1 &= \frac{RCX_1}{2} & RCX_L &= \frac{RCX_L}{2} \\ RM_i &= \frac{\phi_k k_{i-1/2}}{\Delta x} & RM_1 &= 0 \\ RP_i &= \frac{\phi_k k_{i+1/2}}{\Delta x} & RP_L &= 0 \end{aligned}$$

$$BBB_1 = \frac{RCX_1}{\Delta t} + RM_1 + RP_1 + 4\epsilon\sigma(U_1^n, s)^3 + h_{bar}h_{ref}$$

$$BBB_i = \frac{RCX_i}{\Delta t} + RM_i + RP_i$$
(A-14)

Then, using Equations A-14 in Equations 2-7 and 2-8 yields the matrix form of Equation 2-9,

$$[A]\{U_i^n\} + \{b\} = 0$$
(2-9)

where,

$$[A] = \begin{bmatrix} -1 & RP_i/BBB_i & 0 \\ RM_i/BBB_i & -1 & RP_i/BBB_i \\ 0 & RM_i/BBB_i & -1 \end{bmatrix}$$

(A-15)

and,

$$\{b\} = \left\{ \begin{array}{l} \epsilon\sigma[(3U_1^n, s)^4 + U_{f0}^4] + h_{bar}h_{ref}T_{aw} + \frac{RCX_1U_1^{n-1}}{\Delta t} \\ BBB_1 \\ \frac{RCX_iU_i^{n-1}}{\Delta t} / BBB_i \\ \frac{RCX_LU_L^{n-1}}{\Delta t} / BBB_L \end{array} \right\}$$

(A-16)

APPENDIX B

The derivation of the sensitivity equations. The derivative of Equation 2-7 with respect to each parameter yields equations from which the HEATEST program propagates the sensitivity. A vector of parameters is formed and the sensitivity notation is as shown,

$$\Theta = [h_0, h_{a1}, h_{a2}, \phi_c, \phi_k]^T \quad S_{i,k} = \frac{\partial U}{\partial \Theta_k}$$

_____ parameter no
 _____ node point

Defining, $h_{bar} = [h_0 + h_{a1}(a-a1) + (h_{a2}(a-a2))]$, the sensitivity equations at node one are,

$$\begin{aligned} \theta_1: \quad & \frac{\rho \phi_c c \Delta x}{2} \frac{s_{1,1}^n - s_{1,1}^{n-1}}{\Delta t} = \frac{-\phi_k(k_{i+1/2})}{\Delta x} s_{1,1}^n \\ & + \frac{\phi_k(k_{i+1/2})}{\Delta x} s_{2,1}^n - 4\epsilon\sigma(U_1^n)^3 s_{1,1}^n + h_{ref}(T_{aw} - U_1^n) \\ & - s_{1,1}^n h_{ref} h_{bar} \end{aligned} \quad (B-1)$$

$$\begin{aligned} \theta_2: \quad & \frac{\rho \phi_c c \Delta x}{2} \frac{s_{1,2}^n - s_{1,2}^{n-1}}{\Delta t} = \frac{-\phi_k(k_{i+1/2})}{\Delta x} s_{1,2}^n \\ & + \frac{\phi_k(k_{i+1/2})}{\Delta x} s_{2,2}^n - 4\epsilon\sigma(U_1^n)^3 s_{1,2}^n \\ & + (a-a1) h_{ref}(T_{aw} - U_1^n) - s_{1,2}^n h_{ref} h_{bar} \end{aligned} \quad (B-2)$$

$$\theta_3: \frac{\rho \phi_c c \Delta x}{2} \frac{s_{1,3}^n - s_{1,3}^{n-1}}{\Delta t} = \frac{-\phi_k(k_{i+1/2})}{\Delta x} s_{1,3}^n$$

$$+ \frac{\phi_k(k_{i+1/2})}{\Delta x} s_{2,3}^n - 4\epsilon\sigma(U_1^n)^3 s_{1,3}^n$$

$$+ (\alpha - \alpha_2) h_{ref} (T_{aw} - U_1^n) - s_{1,3}^n h_{ref} h_{bar}$$

(B-3)

$$\theta_4: \frac{s_{1,4}^n - s_{1,4}^{n-1}}{\Delta t} = \frac{-2\phi_k k_{i+1/2}}{\rho \phi_c c \Delta x^2} s_{1,4}^n + \frac{2\phi_k k_{1+1/2}}{\rho \phi_c^2 c \Delta x^2} U_1^n$$

$$+ \frac{2\phi_k k_{1+1/2}}{\rho \phi_c c \Delta x^2} s_{1,4}^n - \frac{2\phi_k k_{1+1/2}}{\rho \phi_c^2 c \Delta x^2} U_2^n$$

$$- \frac{8\epsilon\sigma(U_1^n)^3}{\rho \phi_c c \Delta x} s_{1,4}^n + \frac{2\epsilon\sigma(U_1^4 - U_p^4)}{\rho \phi_c^2 c \Delta x}$$

$$- \frac{2h_{bar}h_{ref}}{\rho \phi_c c \Delta x} s_{1,4}^n - \frac{2h_{bar}h_{ref}(T_{aw} - U_1)}{\rho \phi_c^2 c \Delta x}$$

(B-4)

$$\theta_5: \frac{\rho \phi_c c \Delta x}{2} \frac{s_{1,5}^n - s_{1,5}^{n-1}}{\Delta t} = \frac{-\phi_k k_{1+1/2}}{\Delta x} s_{1,5}^n$$

$$- \frac{k_{1+1/2}}{\Delta x} U_1^n + \frac{\phi_k k_{1+1/2}}{\Delta x} s_{2,5}^n + \frac{k_{1+1/2}}{\Delta x} U_2^n$$

$$- 4\epsilon\sigma U_1^3 s_{1,5} - s_{1,5} h_{ref} h_{bar}$$

(B-5)

The sensitivity equations at the interior node points are as

follows,

$$\theta_i, \quad i = 1, 2, 3$$

$$\begin{aligned} \rho \phi_c c \Delta x \frac{s_{i,k}^n - s_{i,k}^{n-1}}{\Delta t} &= \frac{\phi_k k_{i-1/2}}{\Delta x} s_{i-1,k} \\ &- \frac{\phi_k (k_{i-1/2} + k_{i+1/2})}{\Delta x} s_{i,k} \\ &+ \frac{\phi_k k_{i+1/2}}{\Delta x} s_{i+1,k} \end{aligned}$$

(B-6)

θ_4 :

$$\begin{aligned} \frac{s_{i,4}^n - s_{i,4}^{n-1}}{\Delta t} &= \frac{\phi_k k_{i-1/2}}{\rho \phi_c c \Delta x^2} s_{i-1,4}^n - \frac{\phi_k k_{i-1/2}}{\rho \phi_c^2 c \Delta x^2} u_{i-1}^n \\ &- \frac{\phi_k}{\rho \phi_c c \Delta x} \frac{(k_{i-1/2} + k_{i+1/2})}{\Delta x} s_{i,4}^n \\ &+ \frac{\phi_k}{\rho \phi_c^2 c \Delta x} \frac{(k_{i-1/2} + k_{i+1/2})}{\Delta x} u_i^n + \frac{\phi_k k_{i+1/2}}{\rho \phi_c c \Delta x^2} s_{i+1,4}^n \\ &- \frac{\phi_k k_{i+1/2}}{\rho \phi_c^2 c \Delta x^2} u_{i+1}^n \end{aligned}$$

(B-7)

θ_5 :

$$\begin{aligned} \rho \phi_c c \Delta x \frac{s_{i,5}^n - s_{i,5}^{n-1}}{\Delta t} &= \frac{\phi_k k_{i-1/2}}{\Delta x} s_{i-1,5} \\ &- \frac{\phi_k (k_{i-1/2} + k_{i+1/2})}{\Delta x} s_{i,5} \end{aligned}$$

$$\begin{aligned}
& + \frac{\delta_k k_{i+1/2}}{\Delta x} S_{i+1,5} + \frac{k_{i-1/2}}{\Delta x} U_{i-1} \\
& - \frac{(k_{i-1/2} + k_{i+1/2})}{\Delta x} U_i + \frac{k_{i+1/2}}{\Delta x} U_{i+1}
\end{aligned} \tag{B-8}$$

The backface equations are of the same form as the node 1 equations without the convection and radiation terms.

If Equations A-14 are used to reduce the equations to the form of Equation 2-9, the sensitivity equations become,

$$[A'] \{S_{i,k}\} + \{d_k\} = 0 \tag{B-9}$$

where the $[A]$ matrix for the sensitivity equations is the same as the $[A]$ matrix for the temperature equations, A-15. The $\{d\}$ vectors for each parameter are listed as follows,

$$\{d\}_1 = \left\{ \begin{array}{l} \left(h_{ref}(T_{aw} - U_1) + \frac{RCX_1}{\Delta t} S_{1,1}^{n-1} \right) / BBB_1 \\ \left(RCX_i S_{i,1}^{n-1} \right) / BBB_i \end{array} \right\}$$

$$\{d\}_2 = \left\{ \begin{array}{l} \left((\alpha - \alpha_1) h_{ref}(T_{aw} - U_1) + \frac{RCX_1}{\Delta t} S_{1,2}^{n-1} \right) / BBB_1 \\ \left(RCX_i S_{i,2}^{n-1} \right) / BBB_i \end{array} \right\}$$

$$\{d\}3 = \left\{ \begin{array}{l} \left((a-a_2) h_{ref}(T_{aw}-U_1) + \frac{RCX_1}{\Delta t} S_{1,3}^{n-1} \right) / BBB_1 \\ \left(\frac{RCX_i}{\Delta t} S_{i,3}^{n-1} \right) / BBB_i \end{array} \right\}$$

$$\{d\}4 = \left\{ \begin{array}{l} \left(-RP_1 \frac{(U_2-U_1)}{\delta_c} + \varepsilon \sigma \frac{(U_1^4 - U_{\infty}^4)}{\delta_c} - h_{bar} h_{ref}(T_{aw}-U_1) \right. \\ \left. + \frac{RCX_1}{\Delta t} S_{i,4}^{n-1} \right) / BBB_1 \\ \left(-RM_i \frac{(U_{i-1}-U_i)}{\delta_c} + RP_i \frac{(U_i-U_{i+1})}{\delta_c} + \frac{RCX_i}{\Delta t} S_{i,4} \right) / BBB_i \end{array} \right\}$$

$$\{d\}5 = \left\{ \begin{array}{l} \left(\frac{RP_1}{\delta_k} (U_2-U_1) + \frac{RCX_1}{\Delta t} S_{i,5}^{n-1} \right) / BBB_1 \\ \left(RM_i \frac{U_{i-1}}{\delta_k} - (RM_i + RP_i) \frac{U_i}{\delta_k} + RP_i \frac{U_{i+1}}{\delta_k} + \frac{RCX_i}{\Delta t} S_{i,5}^{n-1} \right) / BBB_i \end{array} \right\}$$

(B-10)

APPENDIX C

The HEATEST program follows.

PROGRAM HEATEST 74/855 OPT=0, ROUND=A/ S/ M/-D/-DS FYN 5.1+587
 DO=-LONG/-OT, ARG=-COMMON/-FIXED, CSF= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
 FNNS, I, ANSI=0, L=OUTS, LO=S/-A.

```

1          PROGRAM HEATEST (INPUT,OUTPUT,TAPE2,TAPE3,TAPE4,TAPE5,TAPE6=
2          *OUTPUT,TAPE9,TAPE10,TAPE12,TAPE13,TAPE21,TAPE30)
3
4          C TAPE2 = CALCOMP PLOT FILE
5          C TAPE3 = TEMP ICS
6          C TAPE4 = MANEUVER (BET) FILE
7          C TAPE9 = TEMP IC OUTPUT FILE FROM ADAPTIVE FILTER MODE
8          C TAPE10 = ADAPTIVE TEMPERATURE/STATE ESTIMATE TIME SERIES
9          C TAPE12 = FORMATTED TIME SERIES FOR PLOT
10         C TAPE13 = THERMOCOUPLE MEASUREMENTS
11
12         REAL M1
13         REAL TW1
14         LOGICAL IFICIENT, IFPRINT, IFXFLG
15         COMMON /CFLAG/IFICIENT, IFPRINT, IFXFLG
16         COMMON/CTCMIT/NTCT
17         COMMON/COSP/NPTSS, USI(6), PHI(6,6), NPT, PC(6,6), RR,
18         &QD(6,6), QDT(6,6), QUE(6,6), A(6,6), RCX(6), RP(6), RM(6)
19         COMMON/CSENS/SUSI(6,5), UM1(6)
20         LOGICAL FREAD(13)
21         DIMENSION CI(5)
22         LOGICAL FAUTO
23         DIMENSION FAUTO(7)
24         DIMENSION OP(5)
25         COMMON/CPARAM/HO, HALF(2), PHIC, PHIK, ZP, Z, ALPH(2), KA, S(5),
26         &CIF(5,5), KAF, IFX(5), ACC(5), IFXSUM, NPAR, DALPH(2)
27         EQUIVALENCE (HO, QP(1))
28         COMMON/COMTUN/T, TAW1, ALPHA, H, V, RHO, P, TEMP, C, TRAD, RHUG,
29         &TO, TSINK, XFT, DEL, PDEL
30         COMMON/CHEAT/Q, TS, QREF, TW, M1, RENS, HBAR, HREF
31         COMMON/ICTPS2/TINIT(1), ERALOW, E
32         COMMON/CTIME/TSTART, TSTOP, DIPENT, NRPITER, ITPRAM
33         COMMON/CDX/DX(1)
34         COMMON /CONST/ XP1(13)
35         COMMON/CPARAM2/AVERROR, EQUI, UMEAS
36         COMMON /CFPLOT/ IPLOT, TSCALE, TMIN, TAXL, YSCALE, YMIN, YAXL, ASCALE, AMIN
37         * AXL
38         COMMON /CSMTH/ UICSM(6), PICSM(6,6), UAP(6), PAP(6,6),
39         &SMIC, TSMTH, W(6,6)
40         LOGICAL SMIC
41         COMMON/CKF/K(6), S1(6), J1(6,6), TC(2), NODES(2), KFOPT
42         REAL K, J1
43         COMMON/CPC/NPTPC
44         DIMENSION FLT1(2), FLT2(1)
45         EQUIVALENCE (TAW1,FLT1(1))
46         EQUIVALENCE (REF, ALPH)
47         EQUIVALENCE (TW1,TC(1))
48         DIMENSION TITLE(10)
49         DIMENSION REMARK(10)
50         DIMENSION PLAB1(12), PLAB2(12), NUM(12)
51         DIMENSION VAR(13)
52         DIMENSION PU(2), PE(2), PUF(2), PEF(2)
53         DIMENSION READ(13)
54         DATA NPAR /5/
55         DATA CLAB /1HC/

```

```

48 FHEATEST2 49
49 FHEATEST2 50
50 FHEATEST2 51
51 OCT30 1
52 FHEATEST2 52
53 UPSEP19 2
54 FHEATEST2 53
55 FHEATEST2 54
56 FHEATEST2 55
57 FHEATEST2 56
58 FHEATEST2 57
59 FHEATEST2 58
60 FHEATEST2 59
61 FHEATEST2 60
62 UPSEP19 4
63 OCT24 1
64 OCT24 2
65 FHEATEST2 120
66 FHEATEST2 121
67 FHEATEST2 122
68 FHEATEST2 123
69 FHEATEST2 124
70 FHEATEST2 125
71 FHEATEST2 126
72 FHEATEST2 127
73 FHEATEST2 128
74 FHEATEST2 129
75 FHEATEST2 130
76 FHEATEST2 131
77 FHEATEST2 132
78 FHEATEST2 133
79 FHEATEST2 134
80 FHEATEST2 135
81 UPSEP24 1
82 FHEATEST2 136
83 TSTOP=TSTART+DELS
84 IF(IFXFLG)THEN
85 READ(5,4011,END=4034)NRPITER,IFX,FAUTO,TSTOP,KFOPT,NPTPC,IIC
86 READ(5,4001,INTERV,IFXFLG
87 READ(5,4012,END=4034)FREAD,READ
88 READ(5,4013,FORMAT(3X,13L1,13F8.4)
89 DO 4016 II=1,5
90 IF(FREAD(II))QP(II)=READ(II)
91 READ(5,4013,END=4034)FREAD,READ
92 DO 4017 II=1,2
93 IF(FREAD(II))ALPH(II)=READ(II)
94 READ(5,4014,FORMAT(3X,7L1,7F10.5)
95 GO TO 4033
96 IFXFLG=.FALSE.
97 TSTOP=TSTOPF
98 CONTINUE
99 END IF
100 IF(TSTOP.GT.TSTOPF)TSTOP=TSTOPF
101 IFIXSUM=0
102 DO 30 I=1,NPAR
103 IFIXSUM=IFXSUM+IFX(I)
104 NRPIT=NRPITER+1
105 DO 198 ITPRAM=1,NRPIT
106 KA=1
107 KAF=1
108 REWIND 3
109 REWIND 4
110 REWIND 10
111 REWIND 13
112 CALL ZERO(S,NPAR,1)

```

SUBROUTINE TPS3 747855 OPT=0 ROUND= A/ S/ M/-D,-DS FTN 5.1+587 84/11/19 13.14.29
 DO = LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
 FTNS, I, ANSI=0, L=OUTS, LO=S/-A.

```

1      SUBROUTINE TPS3(DTT)
2      COMMON /CTCMIT/NTCT
3      COMMON /COMTUN/T,TAW1,ALPHA,H,V,RHO,P,TEMP,C,TRAD,RHOG,
4      &TO,TSINK,XFT,DEL,POEL
5      COMMON /CHEAT/Q,TS,QREF,TW,M1,RENS,HBAR,HREF
6      COMMON /COSP/NPTSS,USI(6),PHI(6,6),NPT,PC(6,6),RR,
7      &QD(6,6),QDT(6,6),QUE(6,6),A(6,6),RCX(6),RP(6),RM(6)
8      COMMON /CSENS/SUSI(6,5),UM1(6)
9      COMMON /ICTPS2/TINIT(1),ERALOW,E
10     COMMON /CDX/DX(1)
11     REAL M1
12     DIMENSION AAAA(6),CCCCC(6),DDDD(6),G(6),W(6),
13     EQUIVALENCE (QD(1,1),AAAA(1)),(QD(1,2),CCCCC(1)),(QD(1,3),DDDD(1)),
14     &(QD(1,5),G(1)),(QD(1,1),W(1))
15     DATA SIG/4.761E-13/
16     DATA MIT/2/
17     C SHIFT STORAGE
18     DO 460 I=1,NPTSS
19     UM1(I)=USI(I)
20     C
21     C FORM TRIDIAGONAL MATRIX
22     C
23     DO 511 I=2,NPTSS
24     C
25     BBB=RCX(I)/DTT+RP(I)+RM(I)
26     AAAA(I)=RM(I)/BBB
27     CCCC(I)=RP(I)/BBB
28     DDDD(I)=RCX(I)*UM1(I)/DTT/BBB
29     511 CONTINUE
30     C TRIDIAGONAL SOLUTION
31     DO 540 M=1,MIT
32     C
33     BBB=RCX(1)/DTT+RP(1)+RM(1)+4.*E*SIG*(USI(1)+460.)*+3.+*
34     *HBAR*TREF
35     AAAA(1)=RM(1)/BBB
36     CCCC(1)=RP(1)/BBB
37     DDDD(1)=(RCX(1)*UM1(1)/DTT+E*SIG*(3*USI(1)*4+TRAD*4)+HBAR*TREF*
38     &TAW1)/BBB
39     GL(1)=DDDD(1)
40     W(1)=CCCC(1)
41     DO 520 I=2,NPTSS
42     W(I)=-CCCCC(I)/(1.+AAA(I)*W(I-1))
43     G(I)=(DDD(I)+AAA(I)*G(I-1))/(1.+AAA(I)*W(I-1))
44     UNEW=G(NPTSS)
45     UERMX=ABS(UNEW-USI(NPTSS))
46     USI(NPTSS)=UNEW
47     DO 530 L=2,NPTSS
48     I=NPTSS-L+1
49     UNEW=G(I)-W(I)*USI(I+1)
50     UERR=ABS(UNEW-USI(I))
51     UERMX=AMAX1(UERMX,UERR)
52     USI(I)=UNEW
53     CONTINUE
54     IF(UERMX.LT.ERALOW AND M.GE.5)GO TO 550
55

```

```
1      56          RCX(1)=RHOG*PHIC*ZP*DX(1)*.5
1      57          RP(1)=PHIK*Z/DX(1)
1      58          RM(1)=0.
1      59          A(1,1)=RP(1)/RCX(1)
1      60          A(1,2)=RP(1)/RCX(1)
1      61          ALIN1=-(4.*E*SIG*(USI(1)+460.)*3.+HBAR*HREF)/RCX(1)
1      62          END IF
511    CONTINUE
62          A(1,1)=A(1,1)+ALIN1
63
64          C SYSTEM MATRIX COMPLETED
65          C
66          RETURN
67
68          END
69
```

```
1      8          UPAUG1
1      9          UPAUG1
1     181        FMAKEA2
1     182        FMAKEA2
1     183        FMAKEA2
1     184        OCT30
1     14         FMAKEA2
1     185        FMAKEA2
1     186        FMAKEA2
1     187        FMAKEA2
1     193        FMAKEA2
1     194        FMAKEA2
1     195        FMAKEA2
1     196        FMAKEA2
1     197        FMAKEA2
```

SUBROUTINE MAREA 74/855 OPT=O, ROUND= A/ S/ W/-DS FTN 5.1+587
 DO = LONG/ OT, ARG=- COMMON/ -FIXED, CS= USER/- DB=- TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
 FN5, I, ANSI=0, L=OUTS, LO=S/-A.

```

1
2      SUBROUTINE MAREA
3        COMMON /CPLCL/IPT,PFRAC
4        COMMON/COMTUN/T,TAW1,ALPHA,H,V,RHD,P,TEMP,C,TRAD,RHOG,
5        &TO,TSINK,XFT,DEL,PDEL
6        COMMON /CHEAT/Q,TS,QREF,TW,M1,RENS,HBAR,HREF
7        COMMON /ICTPS2/TINIT(1),ERALOW,E
8        COMMON/COSP/NPTSS,USI(8),PHI(8,8),NPT,PC(8,8),RR,
9        &QD(8,8),QDT(8,8),QUE(8,8),A(8,8),RCX(8),RP(8),RM(8)
10       COMMON/CPC/NPTPC
11       LOGICAL FAUTO
12       DIMENSION FAUTO(7)
13       DIMENSION QP(5)
14       COMMON/CPARAM/HO,HALF(2),PHIK,PHIC,ZP,Z,ALPH(2),KA,S(5),
15       &CIF(5,5),KAF,IFX(5),ACC(5),IFXSUM,NPAR,DALPH(2),
16       EQUIVALENCE (HO,QP(1))
17       COMMON/CKF/K(6),S1(6),J1(6,8),TC(2),NODES(2),KFOPT
18       REAL K,J1
19       COMMON /CDX/DX(1)
20       REAL M1
21       COMMON /CTCMIT/NTCT
22       DATA SIG/4.761E-13/
23       DATA E/-3/
24       DATA HREF/1./
25       DATA RHOG/17.10603937/
26       DATA ZP/3.233477/
27       DATA Z/3.054E-3/
28
29       C SET UP LINEARIZED SYSTEM MATRIX, A
30       CALL ZERO (A(1,1),NPTSS,NPTSS)
31       DO 511 I=1,NPTSS
32       C CURRENT PASS TEMPERATURES
33       C FORM MATRIX
34       C
35       C
36       C I=NPTS
37       C IF ((I.EQ.NPTSS) THEN
38       C   RX(1)=RHOG*PHIC*ZP*DX(1)*.5
39       C   RM(1)=PHIK*Z/DX(1)
40       C   RP(1)=0.
41       C   A(I,I-1)=RM(1)/RCX(I)
42       C   A(I,I)=-RM(1)/RCX(I)
43
44       C BLOCK B INTERIOR POINTS
45
46       C ELSE IF ((I.GT.1).AND.(I.LT.NPTSS)) THEN
47       C   RX(1)=RHOG*PHIC*ZP*DX(1)
48       C   RP(1)=PHIK*Z/DX(1)
49       C   RM(1)=PHIK*Z/DX(1)
50       C   A(I,I-1)=RM(1)/RCX(I)
51       C   A(I,I+1)=RP(1)/RCX(I)
52       C   A(I,I)=-A(I,I-1)-A(I,I+1)
53
54       C SURFACE NODE I=1
55       C ELSE IF (I.EQ.1) THEN

```

SUBROUTINE QUEWAT 747855 OPT=0, ROUND=A/ S/ M7-D, -DS FTN 5.1+587 84/11/19. 13.14.29 PAGE 2

```
56      C WHILE I.EQ.1 ADD BOUNDARY NOISE DUE TO HEAT TERMS
57      IF (I.EQ.1) THEN
58      QUE(I,J) = QER*QER*SDME*SQRT(SDBN**2+SDME**2)*R(I,J)
59      ELSE
60      QUE(I,J) = QER*QER*SDME**2.*R(I,J)
61      END IF
62      C END WHILE
63      C 221 QUE(J,I) = QUE(I,J)
64      222 CONTINUE
65      C RETURN
66      END
67
68
```

```
54      FQUE
55      FQUE
56      FQUE
57      FQUE
58      FQUE
59      FQUE
60      FQUE
61      FQUE
62      FQUE
63      FQUE
64      FQUE
65      FQUE
66      FQUE
67      FQUE
68
```

SUBROUTINE QUEMAT 747855 OPT=0. ROUND= A/ S/ M/-D/-DS FTN 5. 1+587
 DO=-LONG/-DT ARG=-COMMON/-FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
 FTNS,I,ANSI=0,L=OUTS,LO=S/-A.

```

1      FQUE   2
2      HAROLD 28
3      UPOCT09 23
4      UPOCT09 24
5      OCT10   3
6      COMTUN   3
7      UPAUG16 35
8      UPAUG16 36
9      UPAUG16 36
10     UPAUG16 36
11     UPAUG16 37
12     UPAUG16 37
13     UPAUG16 37
14     UPAUG16 38
15     FDKF   3
16     FQUE   15
17     FQUE   17
18     UPOCT09 7
19     FQUE   19
20     FQUE   20
21     FQUE   21
22     FQUE   22
23     FQUE   23
24     FQUE   24
25     FQUE   25
26     FQUE   26
27     FQUE   27
28     OCT30   9
29     FQUE   29
30     HAROLD 29
31     FQUE   30
32     FQUE   31
33     FQUE   32
34     FQUE   33
35     FQUE   34
36     HAROLD 30
37     FQUE   35
38     FQUE   36
39     FQUE   37
40     FQUE   38
41     OCT10   10
42     FQUE   40
43     FQUE   41
44     FQUE   42
45     FQUE   43
46     OCT30   10
47     FQUE   45
48     FQUE   46
49     FQUE   47
50     FQUE   48
51     FQUE   48
52     FQUE   50
53     FQUE   51
54     FQUE   52
55     FQUE   53

SUBROUTINE QUEMAT
COMMON /CONST/ XPI(13)
LOGICAL IFICENT,IFPLOT,IFPRINT
COMMON /CFLAG/ IFPRINT,IFPLOT,IFPRINT
COMMON /COMTUN/T,TAN1,ALPHA,H,V,RHO,P,TEMP,C,TRAD,RHOG,
&TO TSINK,XFT,DEL,PDEL
COMMON /CTCMNT/NTCT
COMMON /CDX/DX(1)
COMMON /CDS/NPTSS,USI(6),NPT,PC(6,6),RR,
&QD(6,6),QD(6,6),QUE(6,6),A(6,6),RCX(6),RP(6),RM(6)
COMMON /ICTPS2/TINIT(1),ERALOW,E
COMMON /CSENS/SUSI(6,5),UM1(6)
COMMON /CKF/K(6),S1(6),J1(6,6),TC(2),NODES(2),KFOPT
REAL K,J1
COMMON /CICSTAT/TR,SDIC,SDMEA,SDBN
COMMON /CHEAT/Q,TS,QREF,TW,M1,RENS,HREF
REAL M1
DIMENSION R(6,6)
EQUivalence (QD(1,1),R(1,1))
C
C
C
C MATRIX OF SPACIAL CORRELATIONS, R
C SPACIAL CORRELATIONS IN RSI MUST ALLOW FOR VARIABLE NODE STRUCTURE
DO 200 I=1,NPTSS
SUMR=0.
DO 200 J=1,NPTSS
XP=SUMR/TR
XP=ABS(XP)
IF(I.EQ.1)XP1(J)=XP*TR
IF XP.GT.100. )XP=100.
R(I,J)=EXP(-XP)
IF ABS(RP(J)).LT.1.E-8 GO TO 200
SUMR=SUMR+RCX(J)/RP(J)
200 R(J,I)=R(I,J)
PRINT*, 'XP1= ', (XP1(I),I=1,NPTSS)
C
C MODEL ERROR MATRIX, QUE
RC=SQRT(ABS((RCX(1)+RCX(2))*5.))
QER=(HBAR*HREF*(TAN1-USI(1))/(RM(1)+RP(1))/2)**2
DO 221 I = 1,NPTSS
C WHILE I.EQ.1 ADD BOUNDARY NOISE DUE TO HEAT TERMS
44 IF (I.EQ.1) THEN
45   QUE(I,I)=((SDBN*QER)**2+(SDME*QER)**2)/2
46 ELSE
47   QUE(I,I) = (SDME*QER)**2.
48 END IF
49 C END WHILE
50 C
51 IF (I.EQ.NPTSS) GO TO 222
52 IF I = 1+1
53 DO 221 J = IP1,NPTSS
54 C

```

```

56      C READ CALCOMP PLOT SPECIFICATIONS
57      C
58      C READ(5,1000)IPLOT
59      READ(5,2000)TSCALE,TMIN,TAXL
60      READ(5,2000)YSCALE,YMIN,YAXL
61      READ(5,2000)ASCALE,AMIN,AAXL
62      READ(5,2000)FORMAT(10X,12,7(3X,12))
63      1000 FORMAT(10X,12,7(3X,12))
64      C READ TIMES
65      C
66      C START-STOP TIMES / PRINT TIME STEP
67      READ(5,2000)START,TSTOP,DTPNT
68      C DATA FOR I.C. SMOOTHER
69      C READ(5,100)SMIC,TSMTH
70      READ(5,100)FORMAT(6X,L1,BX,F10.8)
71      C # OF ITERATIONS/FIX=0 FIXES PARAMETER/AUTO FLAG FOR REFERENCE
72      READ(5,2010)NRPITER,IFX,FAUTO,KFOPT
73      READ(5,2010)ACC
74      C NEWTON-RAPHSON ACCELERATION PARAMETERS(O)ACC)2)
75      READ(5,2015)ACC
76      C HEATING MODEL INITIAL PARAMETERS
77      READ(5,2020)QP(I1),II=1,NPAR
78      C INITIAL REFERENCE VALUES FOR HEATING MODEL
79      READ(5,2030)ALPH
80      C
81      2010 FORMAT(5X,12,3X,5I1,5X,7L1,8X,I1)
82      2015 FORMAT(5X,8F5.2/5X,8F5.2)
83      2020 FORMAT(10X,8F8.4/26X,5F8.4)
84      2030 FORMAT(10X,7F8.4)
85      RETURN
END

```

```

129      FINPUT 129
130      FINPUT 130
131      FINPUT 131
132      FINPUT 132
133      FINPUT 133
134      FINPUT 134
135      FINPUT 135
140      FINPUT 140
142      FINPUT 142
143      FINPUT 143
144      FINPUT 144
145      FINPUT 145
146      FINPUT 146
CHUCK 34 UPSEP24 3
36      CHUCK 36
147      FINPUT 147
148      FINPUT 148
149      FINPUT 149
150      FINPUT 150
151      FINPUT 151
4      UPSEP24 4
153      FINPUT 153
154      UPSEP24 5
5      OCT10 8
156      FINPUT 156
157      FINPUT 157
158      FINPUT 158
159      FINPUT 159
160

```

```

SUBROUTINE TC      74755   OPT=0, ROUND=A/ S/ M/ D/ -DS   FTN 5.1+587
DO=-LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMM
FTNS, I, ANSI=0, L=OUTS, LO=S/-A.

```

PAGE 1

```

1      FIC
2      UPOCT09   21
3      UPOCT09   22
4      UPAUG16   31
5      UPAUG16   32
6      UPAUG16   33
7      UPAUG16   34
8      UPAUG16   35
9      UPAUG16   36
10     UPAUG16   37
11     UPAUG16   38
12     UPAUG16   39
13     UPAUG16   40
14     UPAUG16   41
15     UPOCT09   42
16     FDKF     43
17     FIC      44
18     FIC      45
19     FIC      46
20     FIC      47
21     FIC      48
22     FIC      49
23     FIC      50
24     FIC      51
25     FIC      52
26     FIC      53
27     FIC      54
28     OCT30    55
29     FIC      56
30     FIC      57
31     FIC      58
32     FIC      59
33     FIC      60
34     FIC      61
35     FIC      62
36     FIC      63
37     FIC      64
38     FIC      65
39     FIC      66
40     FIC      67
41     FIC      68
42     FIC      69
43     FIC      70
44     FIC      71
45     FIC      72
46     FIC      73
47     FIC      74
48     FIC      75
49     FIC      76
50     FIC      77

SUBROUTINE IC
LOGICAL IFICIENT, IFPRINT, IFPLOT, IFFPRINT
COMMON /CFLAG/IFICIENT, IFPLOT, IFFPRINT
COMMON /CTCMAT/NTCT
COMMON /CDX/DX(1)
COMMON /COSP/NPTSS, USI(6), PHI(6,6), NPT, PC(6,6), RR,
& QD(6,6), QDT(6,6), QUE(6,6), A(6,6), RCX(6), RP(6), RM(6)
COMMON /ICTPS2/TINIT(1), ERALOW, E
COMMON /CSENS/SUSI(6,5), UN1(6)
COMMON /CKF/K(6), S1(6), J1(6,6), TC(2), NODES(2), KFCOPT
REAL K, J1
COMMON /CICSTAT/TR, SDIC, SDMEA, SDBN
COMMON /CHEAT/Q, TS, QREF, TW, M1, RENS, HBAR, HREF
REAL M1
DIMENSION R(6,6)
EQUIVALENCE (QD(1,1),R(1,1))
C INITIAL SENSITIVITIES
C
RR = SDMEA
C
C MATRIX OF SPACIAL CORRELATIONS, R
C
C SPACIAL CORRELATIONS IN RSI MUST ALLOW FOR VARIABLE NODE STRUCTURE
DO 200 I=1,NPTS
SUMR=0.
DO 200 J=1,NPTS
SUMR=SUMR+TR
XP=SUMR/TR
XP=ABS(XP)
IF (XP.GT.100.)XP=100.
R(I,J)=EXP(-XP)
IF (ABS(RP(J)).LT.1.E-8)GO TO 200
SUMR=SUMR+RCX(J)/RP(J)
200 R(J,I)=R(I,J)
C COVARIANCE MATRIX OF TEMP ICS, PC
C
DO 210 I = 1,NPTSS
PC(I,I) = (SDIC*USI(I))**2.
IF (I.EQ.NPTSS) GO TO 211
IPI = I+1
DO 210 J = IPI,NPTSS
PC(I,J) = USI(I)*USI(J)*SDIC**2.*R(I,J)
210 PC(J,I) = PC(I,J)
211 CONTINUE
END

```

SUBROUTINE INPUT 74/855 OPT=0,ROUND=A/ S/ M/-D,-DS FTN 5.1+587

84/11/19. 13.14.29 PAGE 2

```
56      C READ CALCOMP PLOT SPECIFICATIONS
57      C
58      C      READ(5,1000)IPLOT
59      C      READ(5,2000)TSCALE,TMIN,TMAX
60      C      READ(5,2000)YSCALE,YMIN,YMAX
61      C      READ(5,2000)ASCALE,AMIN,AMAX
62      C      1000 FORMAT(10X,12,7(3X,12))
63
64      C READ TIMES
65
66      C START-STOP TIMES / PRINT TIME STEP
67      C      READ(5,2000)TSTART,TSTOP,DIPENT
68
69      C DATA FOR I.C. SMOOTHER
70      C      READ(5,100)SMIC,TSMTH
71      C      100 FORMAT(6X,L1,8X,F10.8)
72      C # OF ITERATIONS/FIX=0 FIXES PARAMETER/AUTO FLAG FOR REFERENCE
73      C      READ(5,2010)NRPITER,IFX,FAUTO,KFOPT
74      C NEWTON-RAPHSON ACCELERATION PARAMETERS(O)ACC)2)
75      C      READ(5,2015)ACC
76      C HEATING MODEL INITIAL PARAMETERS
77      C      READ(5,2020)(QP(II),II=1,NPAR)
78      C INITIAL REFERENCE VALUES FOR HEATING MODEL
79      C      READ(5,2030)ALPH
80      C      2010 FORMAT(5X,I2,3X,5I1,5X,7L1,8X,I1)
81      C      2015 FORMAT(5X,8F5.2/5X,8F5.2)
82      C      2020 FORMAT(10X,8F8.4/28X,5F8.4)
83      C      2030 FORMAT(10X,7F8.4)
84      C      RETURN
85
```

```
129
130
131
132
133
134
135
136
140
142
143
144
145
146
147
148
149
150
151
152
153
154
155
156
157
158
159
160
```

SUBROUTINE INPUT 74/855 OPT=0, ROUND= A/ S/ M/D/ -DS FTN 5.1+587
 DO=-LONG/-OT, ARG=-COMMON/ -FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
 FTN5.1, ANSI=0, L=OUTS, LO=S/ A.

```

1      SUBROUTINE INPUT
2      LOGICAL IFIFCNT, IFFPLOT, IFPRINT
3      CHARACTER *30 VEH, FLTDT, TMANV, CTPT
4      COMMON /CFLAG/ IFICIENT, IFFPLOT, IFPRINT
5      COMMON /CTCMNT/ NTCT
6      COMMON /CTIME/ TSTART, TSTOP, DTENT, NRITER, ITPRAM
7      COMMON /ICTPS2/ TINIT(1), ERAQW, E
8      COMMON /CDX/DX(1)
9      LOGICAL FAUTO
10     DIMENSION FAUTO(7)
11     DIMENSION QP(5)
12     COMMON /CPARAM/HO, HALF(2), PHIK, ZP, Z, ALPH(2), KA, S(5),
13     &CIF(5,5), KAF, IFIX(5), ACC(5), IFXSUM, NPAR, DALPH(2)
14     EQUIVALENCE (HO, QP(1))
15     COMMON /CKF/K(B), S1(6), J1(6,6), TC(2), NODES(2), KFOPT
16     REAL K, J1
17     COMMON /CSMTH/ UICSM(8), PICSM(8,6), UAP(6), PAP(6,6),
18     &SMIC, TSMTH, W(6,6)
19     LOGICAL SMIC
20     COMMON /CCON/ VEH, FLTDT, TMANV, CTPT
21     COMMON /CICSTAT/TR, SDIC, SDME, SDBN
22     COMMON /CFPLOT/ IPLOT, TSCALE, TMIN, TAXL, YSCALE, YMIN, YAXL, ASCALE, AMIN
23     * , AAXL
24     C READ VEHICLE/MANEUVER UNIQUES
25     C
26     C READ(5,3000) VEH
27     3000 FORMAT(A30)
28     READ(5,3000) FLTDT
29     READ(5,3000) TMANV
30     READ(5,3000) CTPT
31
32     C
33     C READ(5,1000) NTCT
34
35     C
36     C IFICIENT = T ICS FROM DISK           / F CONSTANT ICS FROM CARDS
37     C IFPLOT = T CREATE CALCOMP          / F NO PLOT FILE
38     C IFPRINT = T PRINT TIME SERIES      / F NO TEMP TIME SERIES OUTPUT
39
40     4000 FORMAT(8(9X,L1))
41     READ(5,4000) IFPLOT
42     READ(5,4000) IFPRINT
43     READ(5,4000) IFPRINT
44     2000 FORMAT(10X,F10.5,3(5X,F10.5)/10X,F10.5,3(5X,F10.5))
45
46     C READ IN INITIAL TEMPERATURES (USED IF IFICIENT = F)
47     C
48     READ(5,2000) TINIT
49
50     C
51     C INITIAL COVARIANCE AND ERROR STATISTICS
52
53     C READ(5,2001) TR, SDIC, SDME, SDBN
54     READ(5,2001) SDBN
55     2001 FORMAT(10X,F10.5,3(5X,F10.5))
  
```

PROGRAM HEATEST 747855 OPT=0,ROUND=A/ S/ M/-D,-DS FTN 5.1+587

84711718 13.14.29 PAGE 7

```
1 341 PRINT *, 'T,USI=' ,TTEST, (USI(II),II=1,NPTSS)
1 342 REWIND 3
1 343 TSTART=TSTOP
1 344 GO TO 4
1 345 END IF
1 346 REWIND 3
1 347 WRITE(3) USI,PC,SUSI
1 348 PRINT *, 'T,USI=' ,TTEST, (USI(II),II=1,NPTSS)
1 349 998 IF(IFPLOT) CALL FPLOT
1 350 C PLOT CALCOMP FILE FROM TAPE12
1 351 C PLOT
1 352 STOP
1 353 999 STOP 'END-OF-FILE ENCOUNTERED ON INPUT TAPE IN HEATEST'
1 354 END
1 355
```

```
FHEATEST2 450
FHEATEST2 451
FHEATEST2 452
FHEATEST2 453
FHEATEST2 454
FHEATEST2 455
FHEATEST2 456
FHEATEST2 457
CHUCK 32
FHEATEST2 459
FHEATEST2 460
FHEATEST2 461
FHEATEST2 462
FHEATEST2 463
FHEATEST2 464
FHEATEST2 465
```

```

1 284 IF(NLAB.EQ.0)GO TO 120
1 285 24 CONTINUE
1 286 GO TO 100
1 287 END IF
C PROPAGATION TO TSTOP
C 288
C 289
C 290
C 291 120 DELT=TSTOP-TTEST
C CALL TPS3(DELT)
C TTEST=TSTOP
C WRITE SMOOTHED I.CS TO TAPE3
C IF(SMIC)THEN
C 292 IF(SMIC)THEN
C 293 REWIND (3)
C 294 WRITE(3) UICSM,PICSM,SUSI
C 295 REWIND (3)
C 296 IIC=1
C 297 ENDIF
C EXIT TEMPERATURE/STATE ESTIMATION LOOP
C 298 IF(ITPRAM.LE.NRPITER.OR.NRPITER.EQ.0) THEN
C 299
C UPDATE PARAMETER ESTIMATES - LIST RESULTS
C 300
C 301 CALL PARENT
C 302 IT=1
C 303 DO 197 II=1,NPAR
C 304 IF(IFX(II).EQ.0)THEN
C 305 CI(II)=0.
C 306 GO TO 197
C 307 END IF
C 308 CI(II)=SQRT(ABS(CIF(IT,IT)))
C 309 CALL PARENT
C 310 IT=1
C 311 DO 197 II=1,NPAR
C 312 IF(IFX(II).EQ.0)THEN
C 313 CI(II)=0.
C 314 GO TO 197
C 315 END IF
C 316 CI(II)=SQRT(ABS(CIF(IT,IT)))
C 317 CONTINUE
C 318 IF(ITPRAM.EQ.1)WRITE(6,3079)
C 319 3079 FORMAT(//1X,'ITER',T18,'HALPH1',T30,'HALPH2',T42,'PHIC',
C 320 &T54,'PHIK',T66,'2P',T78,'Z',T88,'ALPHAT',T100,
C 321 &'ALPHA2',T112,'ALPHA3',T6,'QBETA',T18,'QLOGR',T30,'QDELETE',T42
C 322 &,'ODELBF',T54,'QMACH',T86,'PHIKB',T1X,'(CRANE-RAO BOUND)')
C 323 WRITE(6,3080)ITPRAM,(QP(I),I=1,10),(CI(I),I=1,10),
C 324 &(QP(I),I=1,18),(CI(I),I=1,18)
C 325 FORMAT(1X,12.2X,10(F8.5,4X),5X,10(1X,'( ',E8.2,' ),1X)/
C 326 &5X,6(F8.5,4X)/5X,6(1X,(' ,E8.2,' ),1X))
C 327 WRITE(6,3081)AVERRO
C 328 FORMAT(1X,'AVERAGE ERROR = ',E12.5/)
C 329
C 330 END IF
C 331 CONTINUE
C 332 IIC=1
C 333
C 334 IF(SMIC) GO TO 998
C 335 IF(TSTOP.LT.TSTOPF-1.E-6)THEN
C 336 C RESET INITIAL CONDITIONS WITH FOLLOWING DATA TO DISK
C 337
C 338 REWIND 3
C 339 WRITE(3)USSI,PC,SUSI
C 340

```

PROGRAM HEATEST 74/855 DPT=0, ROUND=A/ S/ M/-D. -DS FIN 5.1+587 84/11/19. 13.14.29 PAGE 5

```

2      C SAVE A PRIORI VALUES
1      DO 27 I=1,NPTSS
1      UAP(I)=USI(I)
1      DO 27 J=1,NPTSS
1      PAP(I,J)=PC(I,J)
1      C      DO 25 I=1,NTCT
1      PU(I)=USI(NODES(I))
1      PE(I)=SQRT(ABS(PC(NODES(I),NODES(I))))
1      25 CALL KF
1      E1 = SQRT(ABS(PC(1,1)))
1      238 DO 26 I=1,NTCT
1      PUF(I)=USI(NODES(I))
1      PEF(I)=SORT(ABS(PC(NODES(I),NODES(I))))
1      240 PEF(I)=FORMAT(1X,F8.99)T,PU(1),PE(1),PUF(1),PEF(1),TC(1)
1      WRITE(6,99)T,PU(1),PE(1),PUF(1),PEF(1),TC(1)
1      241 99 FORMAT(1X,F8.2,3X,2(3X,F9.5,3X,'(,E8.2,'),5X,F9.5)
1      WRITE(6,98)(USI(I),I=1,NPTSS)
1      242 98 FORMAT(11,B(F8.5,3X))
1      C INITIALIZE SMOOTHER
1      244 IF (ICOUNT .EQ. 0) THEN
1      245 DO 1000 I=1,NPTSS
1      246   UICSM(I)=USI(I)
1      247   DO 1000 J=1,NPTSS
1      248     PICSM(I,J)=PC(I,J)
1      249     W(I,J)=PC(I,J)
1      250   1000 ENDIF
1      251   TSMTH=TLEST
1      252 C SMOOTH I.C./S
1      253 IF (SMTH) THEN
1      254   IF (ICOUNT) THEN
1      255     IF (ICOUNT .NE. 0) THEN
1      256       CALL FPSM(TSTART)
1      257       DO 510 IM=1,NPTSS
1      258         VAR(IM)=SQRT(ABS(PICSM(IM,IM)))
1      259       510 ENDIF
1      260     ENDIF
1      261   ENDIF
1      262 C WRITE APOSTERIORI STATE ESTIMATE TO TAPE10/TAPE12
1      263 C
1      264 C
1      265 C
1      266 C
1      267 C
1      268 C
1      269 C
1      270 C
1      271 C
1      272 C
1      273 C
1      274 C
1      275 C
1      276 C
1      277 C
1      278 C
1      279 C
1      280 C
1      281 C
1      282 C
1      283 C
    
```

CHUCK 11
 CHUCK 12
 CHUCK 13
 CHUCK 14
 CHUCK 15
 CHUCK 16
 FHEATEST2 312
 FHEATEST2 313
 FHEATEST2 314
 FHEATEST2 315
 FHEATEST2 316
 FHEATEST2 317
 FHEATEST2 318
 FHEATEST2 319
 OCT30 2
 OCT30 3
 OCT30 4
 OCT30 5
 HAROLD 10
 HAROLD 11
 HAROLD 12
 HAROLD 13
 HAROLD 14
 HAROLD 15
 HAROLD 16
 HAROLD 17
 HAROLD 18
 CHUCK 17
 CHUCK 18
 HAROLD 19
 HAROLD 20
 HAROLD 21
 HAROLD 22
 CHUCK 21
 CHUCK 22
 FHEATEST2 320
 FHEATEST2 321
 FHEATEST2 322
 FHEATEST2 323
 FHEATEST2 324
 UPDCT9 16
 FHEATEST2 325
 FHEATEST2 326
 FHEATEST2 327
 FHEATEST2 328
 FHEATEST2 329
 FHEATEST2 330
 FHEATEST2 331
 FHEATEST2 332
 FHEATEST2 333
 HAROLD 25
 FHEATEST2 334
 FHEATEST2 335
 FHEATEST2 336
 FHEATEST2 337
 FHEATEST2 338
 FHEATEST2 339
 FHEATEST2 340
 UPAG24 1

C NOTE: PROPERTIES MAY NEED UPDATED MORE OFTEN IF TC SAMPLE RATE IS LOW
 C CALL MAKEA
 C READ NEXT THERMOCOUPLE SAMPLE
 C
 DO 24 II=1,ITCSK
 READ(13,END=999)NLAB, TUPDT, (TC(I), I=1,NTCT)

```

170 DO 500 IM=1,NPTSS
171   UICSM(IM)=USI(IM)
172   VAR(IM)=SQRT(ABS(PC(IM,IM)))
173   PRINT*, 'UIC=' ,(UICSM(IM),IM=1,NPTSS)
174   PRINT*, 'VAR=' ,(VAR(IM),IM=1,NPTSS)
175   ICOUNT=0
176 C PROPAGATION TO TRAJECTORY SAMPLE TIME/TIMES
177 C
178 C 100 IF(T.LT.TUPDT.AND.T.LT.TSTOP)THEN
179   DELT=T-TLEST
180   CALL TPS3(DELT)
181   CALL SENS(DELT)
182   CALL TPSOSP2(DELT)
183   TLEST=T
184   DTP=DTP+DELT
185 C WHEN DTP.GE.DTPENT THEN WRITE TEMP/STATE ESTIMATES TO TAPE10/TAPE12
186 C
187 C 188 IF ((DTP.GE.DTPENT).AND.(ITPRAM.GT.NRPITER)) THEN
188   DTP = 0.
189   IF(IFPRINT)WRITE(6,0,TLEST,USI(1),TC(1),HBAR,HREF,ALPHA,TO
190   IF(IFPRINT)WRITE(6,0,TLEST,USI(1),TC(1),HBAR,HREF,ALPHA,TO
191   IF(IFPRINT)WRITE(10,0,TLEST,USI(1),EQUI,USI(NODES(1)),Q,
192   &REF,ALPHA,BETA,RENS,DELE,DELB,F,M1
193   IF(IFPLOT)WRITE(12,3055)TLEST,USI(1),EQUI,USI(NODES(1)),TC(1),
194   &ALPHA,BETA,RENS,DELE,DELB,F,M1,QN
195   FORMAT(6E13.7)
196   TRITE=TLEST
197 END IF
198 C
199 C READ NEXT TRAJECTORY SAMPLE
200 DO 22 I1=1,ITRJSK
201   READ(4,END=999)NLAB,T,(FLT1(I),I=1,NLAB)
202   IF(NLAB.EQ.0)T=TSTOP
203   22 CONTINUE
204   CALL HEATTIN
205   GO TO 100
206   END IF
207 C
208 C PROPAGATION TO THERMOCOUPLE SAMPLE TIME/TIMES
209 C
210 C IF(TUPDT.LE.TSTOP)THEN
211   DELT=TUPDT-TLEST
212   CALL TPS3(DELT)
213   CALL SENS(DELT)
214   CALL TPSOSP2(DELT)
215   TLEST=TUPDT
216   DTP=DTP+DELT
217   IF(TUPDT.GE.T)THEN
218     DO 23 I1=1,ITRJSK
219       READ(4,END=999)NLAB,T,(FLT1(I),I=1,NLAB)
220       IF(NLAB.EQ.0)T=TSTOP
221       23 CONTINUE
222       CALL HEATTIN
223   END IF
224 C KALMAN UPDATES
225 C
226 C

```

```

113      CALL ZERO(CIF,NPAR,NPAR)
114      IF(IIC.EQ.0)CALL ZERO(SUSI,NPTSS,NPAR)
115
116      C READ C & B FILE LABEL ON TRAJ TAPE
117
118      READ(4)C,LIC,ITAIL,TITLE,ITEST,FLT,DFLT,DREQ,DCOM
119      33 IF(EOF(4).NE.0) CALL EXIT
120      IF (C .NE. CLAB) CALL EXIT
121      READ(4) LABEL,NRSECT,NREM,(REMARK(I), I=1,NREM),NLAB,
122      *(PLAB1(I),PLAB2(I),NUM(I),I=1,NLAB)
123      IF (LABEL .NE. LABELX) CALL EXIT
124
125      C READ C & B FILE LABEL ON THE T/C MEAS TAPE
126
127      7 READ(13)C,LIC,ITAIL,TITLE,ITEST,FLT,DFLT,DREQ,DCOM
128      44 IF (EOF(13).NE.0) CALL EXIT
129      IF (C .NE. CLAB) CALL EXIT
130      READ(13)LABEL,NRSECT,NREM,(REMARK(I), I=1,NREM),NLAB,
131      *(PLAB1(I),PLAB2(I),NUM(I),I=1,NLAB)
132      IF (LABEL .NE. LABELX) CALL EXIT
133
134      C SET TEMPERATURE INITIAL CONDITIONS
135
136      C IF(IIC.EQ.0)THEN
137      C INITIAL TEMPERATURES
138
139      DO 403 I=1,NPTSS
140      USI(I)=TINIT(I)
141      IF(IF(ICIENT) THEN
142      READ (3) USI
143      END IF
144      END IF
145      IF(IIC.NE.0)THEN
146      REWIND 3
147      READ(3)USI,PC,SUSI
148      REWIND 3
149      END IF
150      EQUI=USI(1)
151
152      C INITIALIZE SMOOTHER
153      C INITIALIZE PARAMETERS AT TSTART
154      DTPO=0.
155      TTEST=TSTART
156      C READ FIRST TRAJECTORY SAMPLE
157      10 READ(4,END=999)NLAB,T,(FLT(I), I=1,NLAB)
158      IF(NLAB.EQ.0)GO TO 999
159      IF(T.LT.TSTART)GO TO 10
160      C CALCULATE REFERENCE HEATING
161      CALL HEATTUN
162      C READ FIRST THERMOCOUPLE SAMPLES AND LOCAL PRESSURE
163      20 READ(13,END=999)NLAB,TUPDT,(TC(I),I=1,NTCT)
164      IF(NLAB.EQ.0)GO TO 999
165      IF(TUPDT.LT.TSTART)GO TO 20
166      C INITIALIZE THERMAL PROPERTIES/A MATRIX
167      CALL MAKEA
168      IF(IIC.EQ.0)CALL IC
169      CALL QUEMAT

```

SUBROUTINE TPS3 747853 OPT=0, ROUND= A/ S/ M/-D, -DS FTN 5.1+587 84711719. 13.14.29 PAGE 2

```
56      IF (UERMX .GT. ERALOW) WRITE(6, 1000)ERALOW,N,T,UERMX,USI(1)
57      1000 FORMAT(1X,15HMAX ERROR TEMP~,E12.6,2X,I2,2X,4(E12.6,1X))
58      550  CONTINUE
59      570  RETURN
60      END
```

```
      FTPS3   130
      FTPS3   131
      FTPS3   132
      FTPS3   133
      FTPS3   134
```

SUBROUTINE SENS 747855 OPT=0. ROUND= A/ S/ W/-0,-DS FTN 5.1+587 84/11/19. 13.14.28
 DO=-LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB = -TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
 FTNS5, I, ANSI=0, L=OUTS, LO=S/-A.

```

1      SUBROUTINE SENS(DTT)
2      COMMON/COMTUN/T,TAW1,ALPHA,H,V,RHO,P,TEMP,C,TRAD,RHOG,
3      &TD,TSINK,XFT,DEL,PDEL
4      COMMON /CHEAT/Q,T,S,QREF,TW M1 RENS HBAR, HREF
5      COMMON/COSP/NPTSS,USI(6),PHI(6,6),NPT,PC(6,6),RR,
6      &QD(6,6), QDT(6,6), QUE(6,6), A(6,6), RCX(6), RP(6), RM(6),
7      COMMON /ICTPS2/TINIT(1),ERALOW,E
8      COMMON /CDX/DX(1)
9      COMMON/CSENS/SUSI(6,5),UM1(6)
10     LOGICAL FAUTO
11     DIMENSION FAUTO(7)
12     DIMENSION QP(5)
13     COMMON/CPARAM/HO HALF(2),PHIK,PHIK,ZP,Z,ALPH(2),KA,S(5),
14     &CIF(5,5),KAF,IFX(5),ACC(5),1FXSUM,NPAR,DALPH(2)
15     EQUIVALENCE (HO,QP(1))
16     REAL M1
17     DIMENSION AA(6),BB(6),CC(6),DD(6),AAA(6),CCC(6),DDD(6),W(6),G(6)
18     EQUIVALENCE (QD(1,1),AA(1)),(QD(1,2),BB(1)),(QD(1,3),CC(1)),
19     &(QD(1,4),DD(1)),(QD(1,5),AAA(1))
20     EQUIVALENCE (QDT(1,1),CCC(1)),(QDT(1,2),DDD(1)),(QDT(1,3),W(1)),
21     &(QDT(1,4),G(1))
22     DATA SIG/4.781E-13/
23     DATA E/-3/
24     DATA NPTS/40/
25
26     C      IF (DTT.EQ.0.0) GO TO 999
27     C      BACKWARD-DIFFERENCE FORMULATION OF DIFF. EQS.
28     C
29     C
30     C
31     C      I=1      RCX(1)=RHOG*PHIC*ZP*DX(1)*.5
32     C      RP(1)=PHIK*Z/DX(1)
33     C      RM(1)=0.
34     C      SET UP TRIDIAGONAL MATRIX (COMMON TERMS ONLY)
35     DO 520 I=1,NPTSS
36     BB(I)=RCX(I)/DTT+RM(I)+RP(I)
37     AA(I)=RM(I)
38     CC(I)=RP(I)
39     DD(I)=RCX(I)/DTT
40     520 CONTINUE
41     .BB(1)=BB(1)+4.*E*SIG*(USI(1)+460.)***3+ HBAR *HREF
42
43     C      I=1      SENSITIVITY FOR EACH PARAMETER
44     DO 530 IP=1,NPAR
45     IF(IFX(IP).EQ.0) GO TO 530
46     DO 531 I=1,NPTSS
47     DDD(I)=DD(I)*SUSI(I,IP)
48
49     C      SENSITIVITY FOR UNIT SURFACE CONDUCTANCE NAUT (THETA ONE)
50     IF(IP.EQ.1) DDD(1)=DDD(1)+HREF*(TAW1-USI(1))
51     C      SENSITIVITIES FOR EACH HEATING MODEL PARAMETER
52     IF(IP.EQ.2) DDD(1)=DDD(1)+HREF*(TAM1-USI(1))*(ALPHA-ALPH(1))
53     IF(IP.EQ.3) DDD(1)=DDD(1)+HREF*(TAM1-USI(1))*(ALPHA-ALPH(2))
54
55     C      SENSITIVITY FOR SPECIFIC HEAT FACTOR PHIC
      UPAUG1          26
      OCT10          3
      COMTUN          3
      UPAUG16         45
      UPAUG16         6
      UPOCT09        5
      UPAUG16         5
      UPAUG16         46
      UPAUG16         4
      UPAUG16         7
      CPARAM          2
      OCT10          2
      UPAUG16         1
      UPAUG16         2
      UPAUG16         3
      UPAUG16         4
      FSENS3          18
      UPOCT09        9
      FSENS3          18
      FSENS3          19
      FSENS3          20
      FSENS3          21
      UPAUG16         47
      UPAUG16         48
      FSENS3          23
      FSENS3          25
      FSENS3          28
      UPAUG1          28
      FSENS3          28
      FSENS3          29
      FSENS3          30
      FSENS3          35
      FSENS3          36
      UPAUG1          29
      UPAUG1          30
      FSENS3          43
      FSENS3          49
      FSENS3          50
      FSENS3          51
      FSENS3          52
      FSENS3          53
      FSENS3          54
      FSENS3          55
      UPAUG1          31
      FSENS3          66
      FSENS3          67
      FSENS3          68
      FSENS3          69
      FSENS3          70
      FSENS3          71
      FSENS3          72
      UPAUG1          32
      UPAUG1          33
      FSENS3          82
      UPAUG1          34
      UPAUG1          35
      UPAUG1          36
  
```

```

56      IF(IP.EQ.4) THEN          37
57        DDD(1)=DDD(1)-RP(1)*(USI(2)-USI(1))-(HBAR*HREF*(TAW1-
58          &USI(1))+E*SIG*((USI(1)+460.)*4-TRAD*4))/PHIC   OCT30 15
59        DO 533 I=2,NPTSS-1          UPAUG1 16
60          DDD(I)=DDD(I)-(RM(I)*(USI(I-1)+RP(I)*(USI(I)-USI(
61            8*I+1)))/PHIC          OCT30 40
62          DDD(NPTSS)=DDD(NPTSS)+RM(NPTSS)*(USI(NPTSS-1)-USI(NPTSS))/PHIC
63        END IF                      UPAUG1 17
64        C SENSITIVITY FOR CONDUCTIVITY FACTOR PHIK          UPAUG1 42
65        IF(IP.EQ.5) THEN          UPAUG1 43
66          DDD(1)=DDD(1)+RP(1)*(USI(2)-USI(1))/PHIK          UPAUG1 44
67          DO 534 I=2,NPTSS-1          UPAUG1 45
68            DDD(I)=DDD(I)+(RM(I)*USI(I-1)-(RM(I)+RP(I))*USI(I)+RP(I)
69              &*USI(I+1))/PHIK          UPAUG1 46
70            DDD(NPTSS)=DDD(NPTSS)+RM(NPTSS)*(USI(NPTSS-1)-USI(NPTSS))/PHIK
71          END IF                      UPAUG1 47
72          DO 535 I=1,NPTSS          UPAUG1 48
73            AAA(I)=AA(I)/BB(I)          FSENS3 49
74            CCC(I)=CC(I)/BB(I)          FSENS3 50
75            DDD(I)=DDD(I)/BB(I)          FSENS3 51
76          C
77          C TRIDIAGONAL SOLUTION          FSENS3 52
78          C ELIMINATION STEP          FSENS3 105
79            G(1)=DDD(1)          FSENS3 106
80            W(1)=-CCC(1)          FSENS3 107
81            DO 536 I=2,NPTSS          FSENS3 108
82              W(I)=-CCC(I)/(1.+AAA(I)*W(I-1))          FSENS3 109
83              G(I)=(DDD(I)+AAA(I)*G(I-1))/(1.+AAA(I)*W(I-1))          FSENS3 110
84              CCC(I)=G(I)          FSENS3 111
85            C BACKWARD SUBSTITUTION          FSENS3 112
86              SUSI(NPTSS,IP)=G(NPTSS)          FSENS3 113
87              DO 537 L=2,NPTSS          FSENS3 114
88                I=NPTSS-L+1          FSENS3 115
89                SUSI(I,IP)=G(I)-W(I)*SUSI(I+1,IP)          FSENS3 116
90                CONTINUE          FSENS3 117
91                CONTINUE          FSENS3 118
92                RETURN          FSENS3 119
93              END          FSENS3 120

```

SUBROUTINE PAREST 74/855 OPT=0 ROUND= A/ S/ M/-D/ -DS FTN 5. 1+587 FTN 5. 1+587
DO=-LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SL/ ER/-ID/-PMD/-ST, PL=5000
FTNS, I, ANSI=0, L=OUTS, LO=S/-A.

```
1          SUBROUTINE PAREST
2          LOCAL FAUTO
3          DIMENSION FAUTO(7)
4          DIMENSION QP(5)
5          COMMON/CPARAM/HO, HALF(2), PHIC, PHIK, ZP, Z, ALPH(2), KA, S(5),
6          &CIF(5,5), KAF, IFX(5), ACC(5), IFXSUM, NPAR, DALPH(2)
7          EQUIVALENCE (HO, QP(1))
8          COMMON/ICTPS2/TINIT(1), ERALOW, E
9          COMMON/CDX/DX(1)
10         COMMON/CTIME/TSTART, TSTOP, DTSTOP, DTPENT, NRPITER, ITRAM
11         COMMON/MAIN2/IMAA2
12         DIMENSION CIF1(8,6)
13         DATA KIN, KOUT/5,6/
14         IMA2=16
15         NR=IFXSUM
16         IF(NR.EQ.1)THEN
17           CIF(1,1)=1./CIF(1,1)
18           GO TO 20
19         END IF
20
21         C INVERT CONDITIONAL INFORMATION MATRIX, CIF
22         CALL GMINV(NR, NR, CIF, CIF1, MR, 1)
23         DO 15 IR=1, NR
24         DO 15 IC=1, NR
25         CIF(IR, IC)=CIFI(IR, IC)
26         IT=1
27         DO 29 IP=1, NPAR
28         IF(IFX(IP).EQ.0)GO TO 29
29         JT=1
30         DO 28 JP=1, NPAR
31         IF(IFX(JP).EQ.0)GO TO 28
32         QP(IP)=QP(IP)+CIF(IT, JT)*S(JT)
33         JT=JT+1
34         CONTINUE
35         IT=IT+1
36         CONTINUE
37         RETURN
38
39
40
41
42
```

SUBROUTINE KF 74/855 OPT=0, ROUND= A/ S/ M/-D/-OS FIN 5. 1+587
 DO=-LONG/-DT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
 FTNS, I, ANSI=0, L=OUTS, LO=S/-A.

```

1      FKFSUM   2
2      FKFSUM   3
3      FKFSUM   3
4      OCT10   11
5      OCT10   12
6      UPDCT09  25
7      UPDSEP24 12
8      FKFSUM   8
9      CPARAM   2
10     DCT10   2
11     UPAUG16  1
12     UPAUG16  2
13     UPAUG16  3
14     UPAUG16  4
15     UPAUG24  3
16     UPAUG24  4
17     UPAUG16  7
18     UPAUG16  8
19     UPAUG16  9
20     UPAUG16 10
21     UPAUG16 11
22     UPAUG16 12
23     UPAUG16 13
24     UPAUG16 14
25     UPAUG16 15
26     UPAUG16 16
27     UPAUG16 17
28     UPAUG16 18
29     UPAUG16 19
30     UPAUG16 20
31     UPAUG16 21
32     UPAUG16 22
33     UPAUG16 23
34     UPAUG16 24
35     UPAUG16 25
36     UPAUG16 26
37     UPAUG16 27
38     UPAUG16 28
39     UPAUG16 29
40     UPAUG16 30
41     UPAUG16 31
42     UPAUG16 32
43     UPAUG16 33
44     UPAUG16 34
45     UPAUG16 35
46     UPAUG16 36
47     UPAUG16 37
48     UPAUG16 38
49     UPAUG16 39
50     UPAUG16 40
51     UPAUG16 41
52     UPAUG16 42
53     UPAUG16 43
54     UPAUG16 44
55     UPAUG16 45
      FKFSUM   46
      FKFSUM   47
      FKFSUM   48
      FKFSUM   49
      FKFSUM   50
      FKFSUM   51
      FKFSUM   52
      FKFSUM   53
      FKFSUM   54
      FKFSUM   55

1      SUBROUTINE KF
2      LOGICAL IFICIENT, IFPRINT
3      COMMON /CSMTH/ UICSM(6), PICSM(6, 6), UAP(6, 6), PAP(6, 6).
4      &SMIC, TSMTH, W(6, 6)
5      COMMON /CFLAG/ IFICIENT, IFPRINT
6      COMMON /CTCMAT/ NTCT
7      COMMON /CPC/NPTPC
8      LOGICAL FAUTO
9      DIMENSION FAUTO(7)
10     DIMENSION QP(5)
11     COMMON/CPARAM/HO, HALF(2), PHIC, PHIK, ZP, Z, ALPH(2), KA, S(5),
12     &CIF(5, 5), KAF, IFX(5), ACC(5), IFXSUM, NPAR, DALPH(2)
13     EQUIVALENCE (HO, QP(1))
14     COMMON /ICTPS2/TINIT(1), ERALOW, E
15     COMMON /CDX/DX(1)
16     COMMON/CSENS/SUSI(8, 5), UJ(1, 8)
17     COMMON/CO3P/NPTSS, USI(8), PHI(6, 6), NPT, PC(6, 6), RR,
18     &QD(6, 6), QDT(6, 6), QUE(6, 6), A(6, 6), RCX(6), RP(6), RM(6)
19     COMMON/CTIME/TSTART, TSTOP, DTPIENT, NRPITER, ITPRAM
20     COMMON/CPARAM2/AVERRO, EQUI, UMEAS
21     COMMON/GK/K(6), S1(6), J1(6, 6), TC(2), NODES(2), KFOPT
22     REAL K, J1
23     COMMON /MAIN1/INP, IPVT(6), WORK(6)
24     INP=6
25
26     SDMEA=RR
27
28     C THE FOLLOWING CONTROL CONSTRUCT SORTS KF UPDATE ITERATIONS
29     C REQUIRED BY VECTOR UPDATES AS SPECIFIED IN THE INPUT DECK
30
31     DO 98 I=1, NPNTSS
32     DO 5 ITT=1, NTCT
33     IF(I.EQ.NODES(ITT)) THEN
34       NODE=NODES(ITT)
35       UMEAS=TC(ITT)
36       GO TO 10
37     END IF
38     CONTINUE
39     GO TO 98
40
41     10  ERROR=UMEAS-UAP(NODE)
42     C IF(KAF.EQ.1) AVERAGE=ERROR
43     AVERAGE=((KAF-1)*AVERAGE+ERROR)/KAF
44     KAF=KAF+1
45
46     C SCORE RUNNING SUMS FOR JACOBIAN OF LIKELIHOOD FN, S,
47     C AND CONDITIONAL INFORMATION MATRIX, CIF
48
49     R=(SDMEA*UMEAS)**2.
50
51     DO 26 KO=1, NPAR
52       S1(KO)=SUSI(NODE, KO)*IFX(KO)*ERROR/(PC(NODE, NODE)+R)
53       DO 25 L=1, NPAR
54         J1(KO, L)=SUSI(NODE, KO)*IFX(KO)*IPC(NODE, NODE)+*
55         *R
  
```

```

56   CONTINUE
57   IT=1
58   DO 29 IP=1,NPAR
59   IF(IFX(IP).EQ.0)GO TO 29
60   S(IT)=S1(IP)+S(IT)
61   JT=1
62   DO 28 JP=1,NPAR
63   IF(IFX(JP).EQ.0)GO TO 28
64   CIF(IT,JT)=U1(IP,JP)+CIF(IT,JT)
65   JT=JT+1
66   CONTINUE
67   IT=IT+1
68   CONTINUE
69   C COMPUTE KALMAN GAIN, K
70   C STATE UPDATE
71   DO 30 IK=1,NPTSS
72   K(IK)=PAP(IK,NODE)/(PAP(NODE,NODE)+R)
73   C IF KOPT=1 UPDATE
74   C IF KOPT=2 UPDATE EXCEPT ON LAST ITERATION( ITPRAM-NRPITER )
75   C IF KOPT=3 DO NOT UPDATE
76   C IF KOPT=4 UPDATE COVARIANCE AND SENSITIVITY ONLY
77   C IF KOPT=5 ONLY UPDATE TEMP ON LAST ITERATION
78   C IF KOPT=6 ONLY UPDATE ON LAST ITERATION
79   C GO TO 101,102,103,104,101,101)KOPT
80   C STATE UPDATE
81   102 IF(ITPRAM.GT.NRPITER)GO TO 103
82   C
83   C STATE UPDATE
84   C
85   101 IF(KOPT GE 6 .AND. ITPRAM.LE.NRPITER)GO TO 103
86   IF(KOPT.EQ.5 .AND. ITPRAM.LE.NRPITER)GO TO 104
87   DO 40 IO=1,NPTSS
88   USI(IO)=USI(IO)+(K(IO)*(UMEAS-UAP(NODE)))
89   C SENSITIVITY UPDATE
90   C
91   104 CONTINUE
92   DO 35 IP=1,NPAR
93   DO 35 L=1,NPTSS
94   35 SUSI(L,IP)=SUSI(L,IP)-K(L)*SUSI(NODE,IP)
95   C COVARIANCE UPDATE, PC - JOSEPH FORM
96   NPTSS=NPTPC
97   C
98   CALL ZERO(QD(1,1),NPTSS,NPTSS)
99   C
100  DO 50 IC=1,NPTSS
101  QD(IC,IC)=1.0
102  QD(IC,NODE)=QD(IC,NODE)-K(IC)
103  CALL MAT4(NPTSS,NPTSS,PC(1,1),QD(1,1),QDT(1,1))
104  CALL MAT4(NPTSS,1,R,K(1),QD(1,1))
105  DO 55 IPC=1,NPTSS
106  DO 55 JPC=1,NPTSS
107  55 PC(IPC,IPC)=QDT(IPC,JPC)+QD(IPC,JPC)
108  NPTSS=NP
109  CONTINUE
110  RETURN
111
112

```

SUBROUTINE KF

74/855 OPT=0, ROUND= A/ S/ M/-D, -DS

FIN 5.1+587 84/11/19. 13.14.29

PAGE 3

113

END

FFFSUM 111

SUBROUTINE FPSM 747855 OPT=0, ROUND=A/S/M/-D FTN 5 1+587
 DO=-LONG/-01, ARG=-COMMON/-FIXED, CS=USER/-FIXED, DB=-TB/-SB/-SL/ER/-ID/-PMD/-ST.PL=5000
 FTNS. I, ANSI=0, L=OUTS, LO=S/-A.

```

1      SUBROUTINE FPSM(TSTART)
2      COMMON /CONST/ XPI(13)
3      COMMON /CTCMNT/ NTCT
4      COMMON /CSMTH/ UICSM(6), PICSM(6), UAP(6), PAP(6, 6).
5      &SMIC, TSMTH, W(6, 6)
6      LOGICAL SMIC
7      COMMON/COSP/NPTSS, USI(6), PHI(6, 6), NPT, PC(6, 6), RR,
8      &QD(6, 6), QDT(6, 6), QUE(6, 6), A(6, 6), RCX(6), RP(6), RM(6)
9      COMMON/CKF/K(6), S1(6), J1(6, 6), TC(2), NODES(2), KFOP1
10     REAL K, J1
11     DIMENSION RINV(8, 8), HTR(6, 8), SFP(6, 6), GAIN(6),
12     &WRK1(6, 6), WRK2(6, 6)
13     COMMON /MAIN1/INP, IPVT(6), WORK(6)
14     INP=6
15
C      THIS ROUTINE IS A FIXED POINT SMOOTHING ALGORITHM
16
C      CALL ZERO(SFP, NPTSS, NPTSS)
17
18     DO 10 I=1, NTCT
19     10   SFP(NODES(I), NODES(I))=1. / (RR*TC(I))*2.
20
C      FORM I-SP AND FIND PHIT
21     CALL MMUL(SFP, PC, NPTSS, NPTSS, WRK1)
22
23     DO 30 I=1, NPTSS
24     30   DO 30 J=1, NPTSS
25       WRK1(I, J)=-WRK1(I, J)
26
27     IF(I .EQ. J)WRK1(I, J)=1.0+WRK1(I, J)
28     WRK2(I, J)=PHI(J, I)
29     DO 35 I=1, NPTSS
30     35   PHI(I, J)=WRK2(I, J)
31
C      FORM W=W*PHIT*(I-SP)
32     CALL MMUL(PHI, WRK1, NPTSS, NPTSS, WRK2)
33     CALL MMUL(W, WRK2, NPTSS, NPTSS, WRK1)
34     DO 40 I=1, NPTSS
35     40   DO 40 J=1, NPTSS
36       W(I, J)=WRK1(I, J)
37
38     C  SOLVE FOR COVARIANCE -- P=P-W(S*PAP*S + S)WTRAN
39     CALL MMUL(PAP, SFP, NPTSS, NPTSS, WRK1)
40     CALL MMUL(SFP, WRK1, NPTSS, NPTSS, WRK2)
41     DO 50 I=1, NPTSS
42       DO 50 J=1, NPTSS
43         WRK1(I, J)=WRK2(I, J) + SFP(I, J)
44       CALL TRI(NPTSS, WRK1, W, PHI, WRK2, NPTSS)
45     50
46     DO 60 I=1, NPTSS
47       DO 60 J=1, NPTSS
48         PICS(1, J)=PICSM(1, J) - WRK2(I, J)
49
50     C  SOLVE FOR SMOOTHED STATE (SCALAR UPDATES)
51     DO 150 I=1, NPTSS
52       DO 110 ITT=1, NTCT
53         IF(I .EQ. NODES(ITT)) THEN
54           NODE=NODES(ITT)
55
      HAROLD 39
      HAROLD 40
      OCT10 13
      UPOCT09 1
      FCSMTH 2
      UPAUG16 6
      UPOCT09 5
      UPAUG16 8
      FDKF 3
      UPOCT09 3
      UPOCT09 4
      OCT10 4
      OCT10 5
      FPSMIC 8
      FPSMIC 9
      FPSMIC 10
      HAROLD 42
      HAROLD 43
      HAROLD 44
      FPSMIC 25
      FPSMIC 26
      FPSMIC 27
      FPSMIC 28
      FPSMIC 29
      FPSMIC 30
      FPSMIC 31
      HAROLD 45
      HAROLD 46
      HAROLD 47
      HAROLD 48
      FPSMIC 33
      FPSMIC 34
      CHUCK1 4
      FPSMIC 36
      FPSMIC 37
      FPSMIC 38
      FPSMIC 39
      FPSMIC 40
      FPSMIC 41
      FPSMIC 42
      FPSMIC 43
      FPSMIC 44
      FPSMIC 45
      FPSMIC 46
      HAROLD 48
      FPSMIC 48
      FPSMIC 49
      FPSMIC 50
      FPSMIC 51
      FPSMIC 52
      FPSMIC 53
      FPSMIC 54
      FPSMIC 55
      FPSMIC 56
  
```

SUBROUTINE FPSH 74/855 OPT=0, ROUND= A/ S/ M/-D, -DS FTN 5.1+587 84/11/19. 13.14.29 PAGE 2

```
1        UMEAS=TC(1TT)
1        GO TO 120
1        ENDIF
58        CONTINUE
59        GO TO 150
60        GO TO 150
61        120 R=(RR*UMEAS)**2.
62        C
63        C COMPUTE GAIN
64        DO 121 IJ=1,NPTSS
65        121 GAIN(IJ)=W(IJ,NODE)/R
66        C
67        UPDATE
68        DO 125 IO=1,NPTSS
69        TCONST=TSTART+XP1(IO)
70        IF(TSMTH.GT.TCONST) GO TO 125
71        UICSM(IO)=UICSM(IO)+GAIN(IO)*(UMEAS-UAP(NODE))
72        CONTINUE
73        125 CONTINUE
74        RETURN
75        END
```

```
56        FPSMIC
57        FPSMIC
58        FPSMIC
59        FPSMIC
60        FPSMIC
61        FPSMIC
62        FPSMIC
63        FPSMIC
64        FPSMIC
65        FPSMIC
66        FPSMIC
67        FPSMIC
68        FPSMIC
69        HAROLD
70        HAROLD
71        HAROLD
72        HAROLD
73        FPSMIC
74        FPSMIC
75        FPSMIC
```

SUBROUTINE MULT 74/855 OPT=0, ROUND= A/ S/ M/-D/-DS FTN 5.1+587 84/11/19 13.14.29 PAGE
 DO - LONG - OT, ARG= -COMMON- / -FIXED, CS= USER / -FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD / ST, PL=5000
 F7NS, I, ANSI=0, L=OUTS, LO=S/-A.

```

1
2      SUBROUTINE MULT(X,Y,N,Z,N2)
3      DIMENSION X(N2,N2),Y(N2,N2),Z(N2,N2)
4      DO 20 I=1,N
5      DO 20 J=1,N
6      Z(I,J)=0.
7      DO 20 K=1,N
8      Z(I,J)=Z(I,J)+X(I,K)*Y(K,J)
9      RETURN
10     END

```

SUBROUTINE MEXP 74/855 OPT=0, ROUND= A/ S/ M/-D/-DS FTN 5 1+587
 DO LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
 F77NS I ANSI=0, L=OUTS, LO=S/-A.

```

1      SUBROUTINE MEXP (N, SUB1, TIME, SUB2, Q, QT, N2)
2      DIMENSION SUB1(N2,N2), SUB2(N2,N2)
3      DIMENSION Q(N2,N2), QT(N2,N2)
4      C   MULTIPLY ELEMENTS OF SUB1 BY TIME
5      DO 102 I=1,N
6      DO 102 J=1,N
7      SUB1(I,J)=SUB1(I,J)*TIME
8      102 SUB2(I,J)=SUB1(I,J)
9      C   GENERATE IDENTITY MATRIX FOR INPUT Q, FOR HQR
10     DO 30 I=1,N
11     DO 30 J=1,N
12     Q(I,J)=0.
13     30 IF(I .EQ. J) Q(I,J)=1.
14     CALL HQR(N, SUB2, Q, IERR, N2)
15     C   MATRIX SUB2 HAS BEEN DESTROYED
16     C   Q IS NOW AN ORTHOGONAL TRANSFORMATION MATRIX
17     DO 40 I=1,N
18     DO 40 J=1,N
19     40 QT(I,J)=Q(J,I)
20     C   QT IS NOW THE TRANSPOSE, AND THE INVERSE, OF Q
21     CALL MULT(SUB1,Q,N,SUB2,N2)
22     CALL MULT(QT,SUB2,N,SUB1,N2)
23     C   SUB1 NOW CONTAINS THE TRIANGULAR MATRIX QT*A*Q
24     DO 50 I=1,N
25     DO 50 J=1,N
26     50 SUB2(I,J)=0.
27     CALL FUNCT(1, N, SUB1, SUB2, N2)
28     C   SUB2 NOW HOLDS EXP(A*TIME) IN TRIANGULAR FORM
29     CALL MULT(SUB2,QT,N,SUB1,N2)
30     CALL MULT(Q,SUB1,N,SUB2,N2)
31     C   SUB2 NOW HOLDS EXP(A*TIME) IN ORIGINAL BASIS FORM
32     RETURN
33     END
34

```

PROGRAM FUNCT 74/855 OPT=0, ROUND= A/ S/ M/-D/-BS FTN 5.1+587
X LONG OT ARG - COMMON - FIXED, CS= USER/-FIXED, DB=- TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
CNS I ANSI-O,L-OUTS, LO=S/-A.

```
      SUBROUTINE FUNCT(R,S,T,F,MM)
      DIMENSION T(MM,MM),F(MM,MM)
      INTEGER R,S
      REAL EXP
      DO 10 I=R,S
      C THE IF-BLOCK GIVES 14-DIGIT ACCURACY WITHOUT UNDERFLOW
      IF( T(I,I) .LT. -43.) THEN
         F(I,I)=0.
      ELSE
         F(I,I)=EXP( T(I,I) )
      END IF
      10 CONTINUE
      C PROCESS THE KTH SUPERDIAGONAL
      N=S-R+1
      NN=N-1
      C NN = NUMBER OF SUPERDIAGONALS IN THE BLOCK
      IF(NN .EQ. 0) RETURN
      DO 13 K=1,NN
         LL=S-K
         DO 12 I=R,LL
            DIFF=T(I,I)-T(I+K,I+K)
            IF(ABS(DIFF) .EQ. 0.0) GO TO 14
            G=T(I,I+K)*(F(I,I)-F(I+K,I+K))
            KK=K-1
            IF(KK .EQ. 0) GO TO 12
            DO 11 M=1,KK
               G=G+F(I,I+M)*T(I+M,I+K)-T(I,I+K-M)*F(I+K-M,I+K)
            11 CONTINUE
            12 F(I,I+K)=G/DIFF
            13 CONTINUE
            14 MM=-MM
            RETURN
         14 MM=-MM
      RETURN
      END
```

```
      FUNCT 2
      FUNCT 3
      FUNCT 4
      FUNCT 5
      FUNCT 6
      FUNCT 7
      FUNCT 8
      FUNCT 9
      FUNCT 10
      FUNCT 11
      FUNCT 12
      FUNCT 13
      FUNCT 14
      FUNCT 15
      FUNCT 16
      FUNCT 17
      FUNCT 18
      FUNCT 19
      FUNCT 20
      FUNCT 21
      FUNCT 22
      FUNCT 23
      FUNCT 24
      FUNCT 25
      FUNCT 26
      FUNCT 27
      FUNCT 28
      FUNCT 29
      FUNCT 30
      FUNCT 31
      FUNCT 32
      FUNCT 33
      FUNCT 34
```

ATF 747855 OPT=0, ROUND=A/ S/ M/-D - DS - FTN 5.1+587 - 84/11/19 - 13 14 28 PAGE 1
 ARG COMMON /FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST PL=5000
 N510 OUTS LO=S/-A.

```

SUBROUTINE HQR(IGH,H,Z,IERR,N2)
  INTEGER I,J,K,L,M,N,EN,II,JJ,LL,MM,NA,NM,NN,N2,
  X      IGH,ITS,LOW,MP2,EMM2,IERR,MINO
  REAL HI,N2,N2,I2(N2,N2)
  REAL P,Q,R,S,T,W,X,Y,RA,SA,VI,VR,ZZ,NORM
  REAL MACHEP,SORT,ABS,SIGN,REAL,AIMAG
  LOGICAL NOTLAS
  COMPLEX Z3,CMPLX

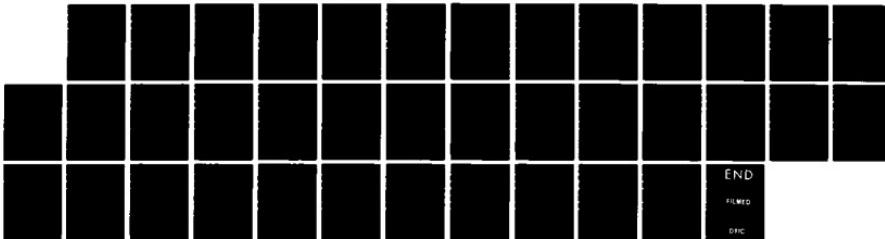
C
C MACHEP IS A PARAMETER THAT SPECIFIES PRECISION
C MACHEP=0.000000000001
C NM=IGH
C
C      IGH
C      LOW= 1
C      IERR= 0
C      NORM= 0
C      K= 1
C
C      COMPUTE MATRIX NORM
C      DO 50 I = 1,N
C      DO 40 J = K,N
C      40 NORM = NORM + ABS(H(I,J))
C      K = I
C 50 CONTINUE
C      EN = IGH
C      T = 0.0
C
C      *** SEARCH FOR NEXT EIGENVALUES****
C 60 IF(EN .LT. LOW) GO TO 1001
C      ITS = 0
C      NA = EN - 1
C      ENM2 = NA - 1
C
C      ***LOOK FOR SINGLE SMALL SUB-DIAGONAL ELEMENT
C      FOR L=EN STEP -1 UNTIL LOW DO ***
C 70 DO 80 LL = LOW, EN
C      L = EN + LOW - LL
C      IF(L .EQ. LOW) GO TO 100
C      S = ABS(H(L-1,L-1)) + ABS(H(L,L))
C      IF(S .EQ. 0.0) S = NORM
C      IF(ABS(H(L,L-1)) .LE. MACHEP * S) GO TO 100
C 80 CONTINUE
C      *** FORM SHIFT ***
C 100 X = H(EN,EN)
C      IF(L .EQ. EN) GO TO 270
C      Y = H(NA,NA)
C      W = H(EN,NA) * H(NA,EN)
C      IF(L .EQ. NA) GO TO 280
C      IF(ITS .EQ. 30) GO TO 1000
C      IF(ITS .NE. 10 .AND. ITS .NE. 20) GO TO 130
C      *** FORM EXCEPTIONAL SHIFT ***
C      T = T + X
C      DO 120 I = LOW,EN
C      H(I,I) = H(I,I) - X
C      S = ABS(H(EN,NA)) + ABS(H(NA,ENM2))
C      X = 0.75 * S
  
```

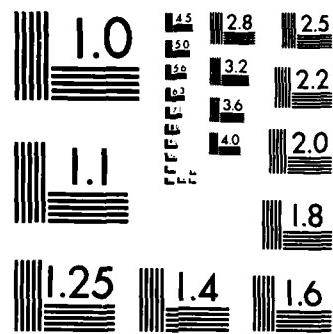
```

      X
      W 0.4375 * S * S
130 ITS = ITS + 1
C   *** LOOK FOR TWO CONSECUTIVE SMALL SUB-DIAGONAL ELEMENTS ***
C   *** FOR M=EN-2 STEP -1 UNTIL L DO ***
DO 140 MM = L, ENM2
  M = ENM2 + L - MM
  ZZ = H(M,M)
  R = X - ZZ
  S = Y - ZZ
  P = (R * S - W) / H(M+1,M) + H(M,M+1)
  Q = H(M+1,M+1) - ZZ - R - S
  R = H(M+2,M+1)
  S = ABS(P) + ABS(Q) + ABS(R)
  P = P/S
  Q = Q/S
  R = R/S
  IF (M .EQ. L) GO TO 150
  IF (ABS(H(M,M-1)) * (ABS(Q) + ABS(R)) .LE. MACHEP * ABS(P))
    X = -(ABS(H(M-1,M-1)) + ABS(ZZ) + ABS(H(M+1,M+1)))
    GO TO 150
140 CONTINUE
150 MP2 = M + 2
DO 160 I = MP2, EN
  H(I,I-2) = 0.0
  IF (I .EQ. MP2) GO TO 160
  H(I,I-3) = 0.0
160 CONTINUE
C   * DOUBLE QR STEP INVOLVING ROWS L TO EN AND COLUMNS M TO EN *
DO 280 K = M, NA
  NOTLAS = K .NE. NA
  IF (K .EQ. M) GO TO 170
  P = H(K,K-1)
  Q = H(K+1,K-1)
  R = 0.0
  IF (NOTLAS) R = H(K+2,K-1)
  X = ABS(P) + ABS(Q) + ABS(R)
  IF (X .EQ. 0.0) GO TO 260
  P = P/X
  Q = Q/X
  R = R/X
  170 S = SIGN(SQRT(P+Q*Q+R*R),P)
  IF (K .EQ. M) GO TO 180
  H(K,K-1) = -S * X
  GO TO 190
180  IF (L .NE. M) H(K,K-1) = -H(K,K-1)
  P = P + S
  X = P/S
  Y = Q/S
  ZZ = R/S
  Q = Q/P
  R = R/P
  *** ROW MODIFICATION ***
DO 210 J = K, N
  P = H(K,J) + Q * H(K+1,J)
  IF (NOTLAS) GO TO 200
  P = P + R * H(K+2,J)
  H(K+2,J) = H(K+2,J) - P + ZZ

```

AD-A153 839 HEATING PARAMETER ESTIMATION USING COAXIAL THERMOCOUPLE 2/2
GRAGES IN WIND TUN. (U) AIR FORCE INST OF TECH
WRIGHT-PATTERSON AFB OH SCHOOL OF ENGI. N T CAHOON
UNCLASSIFIED DEC 84 AFIT/GAE/RA/84D-3 F/G 9/2 NL





MICROCOPY RESOLUTION TEST CHART
NATIONAL BUREAU OF STANDARDS-1963-A

```

113      200      H(K+1,J) = H(K+1,J) - P * Y
114      H(K,J) = H(K,J) - P * X
115      CONTINUE
116      J = MINO(EN,K+3)
117      C  *** COLUMN MODIFICATION ***
118      DO 230 I = 1, J
119      P = X * H(I,K) + Y * H(I,K+1)
120      IF( .NOT. NOTLAS) GO TO 220
121      P = P + ZZ * H(I,K+2)
122      H(I,K+2) = H(I,K+2) - P * R
123      H(I,K+1) = H(I,K+1) - P * Q
124      H(I,K) = H(I,K) - P
125      CONTINUE
126      C  *** ACCUMULATE TRANSFORMATIONS ***
127      DO 250 I = LOW, IGH
128      P = X * Z(I,K) + Y * Z(I,K+1)
129      IF( .NOT. NOTLAS) GO TO 240
130      P = P + ZZ * Z(I,K+2)
131      Z(I,K+2) = Z(I,K+2) - P * R
132      Z(I,K+1) = Z(I,K+1) - P * Q
133      Z(I,K) = Z(I,K) - P
134      CONTINUE
135      260  CONTINUE
136      GO TO 70
137      C  *** ONE ROOT FOUND ***
138      270  H(EN,EN) = X + T
139      EN = NA
140      GO TO 60
141      C  *** TWO ROOTS FOUND ***
142      280  P = (Y - X)/ 2.0
143      Q = P*p + N
144      ZZ = SQRT(ABS(Q))
145      H(EN,EN) = X + T
146      X = H(EN,EN)
147      H(NA,NA) = Y + T
148      IF( Q .LT. 0.0) GO TO 320
149      C  *** REAL PAIR ***
150      ZZ = P + SIGN(ZZ,P)
151      X = H(EN,NA)
152      S = ABS(X) + ABS(ZZ)
153      P = X / S
154      Q = ZZ / S
155      R = SQRT(P*p + Q*q)
156      P = P / R
157      Q = Q / R
158      C  *** ROW MODIFICATION ***
159      DO 280 J = NA, N
160      ZZ = H(NA,J)
161      H(NA,J) = Q*ZZ + P*H(EN,J)
162      H(EN,J) = Q*H(EN,J) - P*ZZ
163      CONTINUE
164      C  *** COLUMN MODIFICATION ***
165      DO 300 I = 1, EN
166      ZZ = H(I,NA)
167      H(I,NA) = Q*ZZ + P*H(I,EN)
168      H(I,EN) = Q*H(I,EN) - P*ZZ
169      CONTINUE

```

```
170      C *** ACCUMULATE TRANSFORMATIONS ***
171      DO 310  I = LOW, IGH
172      ZZ = Z(I,NA)
173      Z(I,NA) = Q*ZZ + P*Z(I,EN)
174      Z(I,EN) = Q*Z(I,EN) - P*ZZ
175      310 CONTINUE
176      GO TO 330
177      C *** COMPLEX PAIR ***
178      320 CONTINUE
179      330 EN = ENM2
180      GO TO 60
181      C * SET ERROR - NO CONVERGENCE TO EIGENVALUE AFTER 30 ITERATIONS
182      1000 IERR = EN
183      1001 RETURN
184      END
```

```
171      HQR
172      HQR
173      HQR
174      HQR
175      HQR
176      HQR
177      HQR
178      HQR
179      HQR
180      HQR
181      HQR
182      HQR
183      HQR
184      HQR
185      HQR
```

```

SUBROUTINE MUL72   77/355   OPT=0, ROUND= A/ST/ N/D/-DS   FTN 5.1+87    84/11/18
DO=-LONG/-OT ARG=-COMMON/-FIXED, CS= USER/-FIXED, L=0/-OUTS, LO=S/-A.
FTNS, I, ANSI=0, L=OUTS, LO=S/-A.

```

```

1      SUBROUTINE MULT2(N,X,Y,Z,N2)
2      COMPUTE Z=X*Y"
3      DIMENSION X(N2,N2),Y(N2,N2),Z(N2,N2)
4      DO 20 I=1,N
5      DO 20 J=1,N
6      Z(I,J)=0.
7      DO 20 K=1 N
8      Z(I,J)= Z(I,J)+X(I,K)*Y(J,K)
9      RETURN
10     END

```

SUBROUTINE TRI 74/855 OPT=0, ROUND=A/ S/ W/-DS FTN 5.1+587
DO=-LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/-SL/ SB/-TB/-DB/-ID/-PMD/-ST, PL=5000
FTNS,I,ANSI=0,L=OUTS,LO=S/-A.

```
1      SUBROUTINE TRI(N,Q,X,W,Z,N2)
2      C COMPUTES Z=XQX"
3      C DIMENSION Q(N2,N2),X(N2,N2),W(N2,N2),Z(N2,N2)
4      C CALL MULT(X,Q,N,W,N2)
5      C X*Q IS STORED IN W
6      C CALL MULT2(N,W,X,Z,N2)
7      C Z=W*X"
8      C RETURN
9      END
```

1 TRI 2
2 TRI 3
3 TRI 4
4 TRI 5
5 TRI 6
6 TRI 7
7 TRI 8
8 TRI 9
9 TRI 10

SUBROUTINE SGEFA 747855 OPT=O ROUND= A7 S7 M7-D,-DS FTN 5.1+587 84/11/19 13.14.29 PAGE 1
 DO=-LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
 FTNS, I, ANSI=0, L=OUTS, LO=S/-A.

```

1      SGEFA 2
2      SGEFA 3
3      SGEFA 4
4      SGEFA 5
5      SGEFA 6
6      SGEFA 7
7      SGEFA 8
8      SGEFA 9
9      SGEFA 10
10     SGEFA 11
11     SGEFA 12
12     SGEFA 13
13     SGEFA 14
14     SGEFA 15
15     SGEFA 16
16     SGEFA 17
17     SGEFA 18
18     SGEFA 19
19     SGEFA 20
20     SGEFA 21
21     SGEFA 22
22     SGEFA 23
23     SGEFA 24
24     SGEFA 25
25     SGEFA 26
26     SGEFA 27
27     SGEFA 28
28     SGEFA 29
29     SGEFA 30
30     SGEFA 31
31     SGEFA 32
32     SGEFA 33
33     SGEFA 34
34     SGEFA 35
35     SGEFA 36
36     SGEFA 37
37     SGEFA 38
38     SGEFA 39
39     SGEFA 40
40     SGEFA 41
41     SGEFA 42
42     SGEFA 43
43     SGEFA 44
44     SGEFA 45
45     SGEFA 46
46     SGEFA 47
47     SGEFA 48
48     SGEFA 49
49     SGEFA 50
50     SGEFA 51
51     SGEFA 52
52     SGEFA 53
53     SGEFA 54
54     SGEFA 55
55     SGEFA 56

SUBROUTINE SGEFA(A,LDA,N,IPVT,INFO)
INTEGER LDA,N,IPVT(1),INFO
REAL A(LDA,1)

C SGEFA FACTORS A REAL MATRIX BY GAUSSIAN ELIMINATION.

C ON ENTRY:
C   A: THE MATRIX TO BE FACTORED
C   LDA: THE LEADING DIMENSION OF THE ARRAY A
C   N: THE ORDER OF THE ARRAY A

C ON RETURN
C   A: AN UPPER TRIANGULAR MATRIX AND THE MULTIPLIERS
C      WHICH WERE USED TO OBTAIN IT.
C   IPVT: AN INTEGER VECTOR OF PIVOT INDICES
C   INFO: = 0 NORMAL VALUE.
C         = K IF U(K,K) EQ. 0.0. THIS IS NOT AN ERROR
C            CONDITION FOR SGEFA. BUT INDICATES THAT
C            SGEFA WILL DIVIDE BY ZERO WHEN CALLED.
C THIS IS FROM LINPACK USER'S GUIDE, VERSION 08/14/78
C REAL T
INTEGER ISAMAX,J,K,KP1,L,NM1

C GAUSSIAN ELIMINATION WITH PARTIAL PIVOTING
INFO=0
NM1=N-1
IF(NM1.LT.1) GO TO 70
DO 60 K=1,NM1
  KP1=K+1
  C FIND L = PIVOT INDEX
  L=ISAMAX(N-K+1,A(K,K),1)+K-1
  IPVT(K)=L
  C ZERO PIVOT IMPLIES THIS COLUMN IS TRIANGULARIZED
  IF(A(L,K).EQ.0.OEO) GO TO 40
  C INTERCHANGE IF NECESSARY
  IF(L.EQ.K) GO TO 10
  T=A(L,K)
  A(L,K)=A(K,K)
  A(K,K)=T
  10 CONTINUE
  C COMPUTE MULTIPLIERS
  T=-1.OEO/A(K,K)
  CALL SSCAL(N-K,T,A(K+1,K),1)
  C ROW ELIMINATION WITH COLUMN INDEXING
  DO 30 J=KP1,N
    T=A(L,J)
    IF(L.EQ.K) GO TO 20
    A(L,J)=A(K,J)
    A(K,J)=T
    20 CONTINUE
  END

```

```
56      CALL SAXPY(N-K,T,A(K+1,K),1,A(K+1,J),1)
57      30 CONTINUE
58      GO TO 50
59      40 CONTINUE
60      INFO= K
61      50 CONTINUE
62      60 CONTINUE
63      70 CONTINUE
64      IPVT(N)= N
65      IF(A(N,N) .EQ. 0.0E0) INFO= N
66      RETURN
67      END
```

```
57      SGEFA 58
58      SGEFA 59
59      SGEFA 60
60      SGEFA 61
61      SGEFA 62
62      SGEFA 63
63      SGEFA 64
64      SGEFA 65
65      SGEFA 66
66      SGEFA 67
67      SGEFA 68
```

SUBROUTINE SGEDI 74/855 OPT=0 ROUND= A/ S/ M/-D,-DS FTN 5.1+587 84/11/19. 13.14.29
 DO -LONG/-OUT ARG=-COMMON/-FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
 FTNS. I,ANSI=0,L=OUTS,LO=S/-A.

```

1
2   SUBROUTINE SGEDI(A,LDA,N,IPVT,WORK)
3     INTEGER LDA,N,IPVT(1)
4     REAL A(LDA,1),WORK(1)
5
6     C SGEDE COMPUTES INVERSE OF MATRIX A USING
7     C FACTORS COMPUTED BY SGEFA.
8
9     C ON ENTRY:
10    C   A: THE OUTPUT FROM SGEFA, REAL(LDA,N)
11    C   LDA: THE LEADING DIMENSION OF ARRAY A
12    C   N: THE ORDER OF MATRIX A
13    C   IPVT: THE PIVOT VECTOR FROM SGEFA, INTEGER(N)
14    C   WORK: WORK VECTOR, CONTENTS DESTROYED, REAL(N)
15
16     C ON RETURN:
17    C   A: INVERSE OF THE ORIGINAL MATRIX
18    C   ERROR CONDITION: A DIVISION BY ZERO WILL OCCUR IF THE
19    C   INPUT FACTOR CONTAINS A ZERO ON THE DIAGONAL.
20    C   IT WILL NOT OCCUR IF SGEFA HAS SET INFO=0
21
22    C THIS IS FROM LINPACK USER'S GUIDE, VERSION 08/14/78
23    REAL T
24    INTEGER I,J,K,KB,KP1,L,NM1
25
26    C COMPUTE INVERSE
27    DD 100 K=1,N
28    A(K,K)= 1.0EO/A(K,K)
29    T= -A(K,K)
30    CALL SSCAL(K-1,T,A(1,K),1)
31    KP1= K+1
32    IF (N .LT. KP1) GO TO 80
33    DO 80 J=KP1,N
34    T= A(K,J)
35    A(K,J)= 0.0EO
36    CALL SAXPY(K,T,A(1,K),1,A(1,J),1)
37    80 CONTINUE
38    80 CONTINUE
39    100 CONTINUE
40
41    C FORM INVERSE(U)*INVERSE(L)
42    NM1= N-1
43    IF (NM1 .LT. 1) GO TO 140
44    DO 130 KB=1,NM1
45    K= N-KB
46    KP1= K+1
47    DO 110 I=KP1,N
48    WORK(I)= A(I,K)
49    A(I,K)= 0.0EO
50    110 CONTINUE
51    DO 120 J=KP1,N
52    T= WORK(J)
53    CALL SAXPY(N,T,A(1,J),1,A(1,K),1)
54    120 CONTINUE
55    L= IPVT(K)

```

SUBROUTINE SGEDI 747855 OPT=0, ROUND= A/ S/ M/-D, -DS FTN 5.1+587

84711719. 13.14.28 PAGE 2

```
56       IF(L_.NE._K) CALL SSHAP(N,A(1,K),1,A(1,L),1)
57       130 CONTINUE
58       140 CONTINUE
59       RETURN
60       END
```

```
57       SGEDI
58       SGEDI
59       SGEDI
60       SGEDI
61
```

FUNCTION ISAMAX 747855 OPT=O, ROUND= A/ S/ M/-D,-DS FTN 5.1+587 84/11/19. 13.14.29
00=-LONG/-DT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
FTN5, I, ANSI=0, L=OUTS, LO=S/-A.

```
1      INTEGER FUNCTION ISAMAX(N,SX,INCX)
2      C   ISAMAX FINDS INDEX OF ELEMENT WITH MAX. ABSOLUTE VALUE.
3      C   LINPACK USER'S GUIDE, VERSION 03/11/78
4      REAL SX(1),SMAX
5      INTEGER I,INCX,IX,N
6      ISAMAX= 0
7      IF (N .LT. 1) RETURN
8      ISAMAX= 1
9      IF (N .EQ. 1) RETURN
10     IF (INCX .EQ. 1) GO TO 20
11
12     C   CODE FOR INCREMENT NOT EQUAL TO 1
13    IX= 1
14    SMAX= ABS(SX(1))
15    IX= IX+INCX
16    DO 10 I=2,N
17    IF (ABS(SX(I)) .LE. SMAX) GO TO 5
18    ISAMAX= 1
19    SMAX= ABS(SX(IX))
20    IX= IX+INCX
21    10 CONTINUE
22    RETURN
23
24    C   CODE FOR INCREMENT EQUAL TO 1
25    20 SMAX= ABS(SX(1))
26    DO 30 I=2,N
27    IF (ABS(SX(I)) .LE. SMAX) GO TO 30
28    ISAMAX= 1
29    SMAX= ABS(SX(1))
30    30 CONTINUE
31    RETURN
32
33
```

```

1      SUBROUTINE SAXPY(N,SA,SX,INCX,SY,INCY)
2      CONSTANT TIMES A VECTOR PLUS A VECTOR.
3      USES UNROLLED LOOP FOR INCREMENTS = 1.
4      FROM LINPACK USER'S GUIDE, VERSION 03/11/78
5      REAL SX(1), SY(1), SA
6      INTEGER I, INCX, INCY, IX, IY, M, MP1, N
7      IF(N .LE. 0) RETURN
8      IF(SA .EQ. 0.0) RETURN
9      IF(INCX .EQ. 1 .AND. INCY .EQ. 1) GO TO 20
10
11      C   CODE FOR UNEQUAL INCREMENTS OR FOR
12      C   EQUAL INCREMENTS NOT EQUAL TO 1
13      IX= 1
14      IY= 1
15      IF(INCX .LT. 0) IX= (-N+1)*INCX + 1
16      IF(INCY .LT. 0) IY= (-N+1)*INCY + 1
17      DO 10 I=1,N
18      SY(IY)= SY(IY)+ SA*SX(IX)
19      IX= IX+INCX
20      IY= IY+INCY
21      10 CONTINUE
22      RETURN
23
24      C   CODE FOR BOTH INCREMENTS EQUAL TO 1
25      C   CLEAN-UP LOOP
26      20 M=MOD(N,4)
27      IF(M .EQ. 0) GO TO 40
28      DO 30 I=1,M
29      SY(I)= SY(I)+ SA*SX(I)
30      30 CONTINUE
31      IF(N .LT. 4) RETURN
32      40 MP1= M+1
33      DO 50 I=MP1,N,4
34      SY(I)= SY(I)+ SA*SX(I)
35      SY(I+1)= SY(I+1)+ SA*SX(I+1)
36      SY(I+2)= SY(I+2)+ SA*SX(I+2)
37      SY(I+3)= SY(I+3)+ SA*SX(I+3)
38      50 CONTINUE
39      END

```

```

1      SUBROUTINE SSCAL(N,SA,SX,INCX)
2      SCALES A VECTOR BY A CONSTANT.
3      USES UNROLLED LOOPS FOR INCREMENT EQUAL TO 1.
4      LINPACK USER'S GUIDE, VERSION 03/11/78
5
6      REAL SA, SX(1)
7      INTEGER I, INCX, M, MP1, N, NINCX
8      IF(N .LE. 0) RETURN
9      IF(INCX .EQ. 1) GO TO 20
10
11      C   CODE FOR INCREMENT NOT EQUAL TO 1
12      NINCX= N*INCX
13      DO 10  I=1,NINCX, INCX
14      SX(I)= SA*SX(I)
15      CONTINUE
16      RETURN
17
18      C   CODE FOR INCREMENT EQUAL TO 1.
19      C   CLEAN-UP LOOP
20      M= MOD(N,5)
21      IF(M .EQ. 0) GO TO 40
22      DO 30  I=1,M
23      SX(I)= SA*SX(I)
24      CONTINUE
25      IF(N .LT. 5) RETURN
26      MP1= M+1
27      DO 50  I=MP1,N,5
28      SX(I)= SA*SX(I)
29      SX(I+1)= SA*SX(I+1)
30      SX(I+2)= SA*SX(I+2)
31      SX(I+3)= SA*SX(I+3)
32      SX(I+4)= SA*SX(I+4)
33      CONTINUE
34      END

```

SUBROUTINE SSWAP 74/855 OPT=0, ROUND= A/ S/ M/-D/-DS FTN 5.1+587 84/11/19 13.14.29
OO - LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
FTN5, I, ANSI=0, L=OUTS, LO=S/-A.

```
1      SUBROUTINE SSWAP(N,SX,INCX,SY,INCY)
2      C   INTERCHANGES TWO VECTORS.
3      C   USES UNROLLED LOOPS FOR INCREMENTS EQUAL TO 1.
4      C   LINPACK USER'S GUIDE, VERSION 03/11/78
5      C
6      REAL SX(1), SY(1), STEM
7      INTEGER I, INCX, INCY, IX, IY, M, MP1, N
8      IF(N .LE. 0) RETURN
9      IF(INCX .EQ. 1 .AND. INCY .EQ. 1) GO TO 20
10     C   CODE FOR UNEQUAL INCREMENTS OR EQUAL INCREMENTS NOT EQUAL TO 1
11     IX= 1
12     IY= 1
13     IF(INCX .LT. 0) IX= (-N+1)*INCX+1
14     IF(INCY .LT. 0) IY= (-N+1)*INCY+1
15     DO 10 I=1,N
16       STEM= SX(IX)
17       SX(IX)= SY(IY)
18       SY(IY)= STEM
19       IX= IX+INCX
20       IY= IY+INCY
21
22     CONTINUE
23     RETURN
24
25     C   CODE FOR BOTH INCREMENTS EQUAL TO 1.
26     C   CLEAN-UP LOOP
27     20 M= MOD(N,3)
28     IF(M .EQ. 0) GO TO 40
29     DO 30 I=1,M
30     STEM= SX(I)
31     SX(I)= SY(I)
32     SY(I)= STEM
33     CONTINUE
34     IF(N .LT. 3) RETURN
35     40 MP1= M+1
36     DO 50 I=MP1,N,3
37     STEM= SX(I)
38     SX(I)= SY(I)
39     SY(I)= STEM
40     STEM= SX(I+1)
41     SX(I+1)= SY(I+1)
42     SY(I+1)= STEM
43     STEM= SX(I+2)
44     SX(I+2)= SY(I+2)
45     SY(I+2)= STEM
46
47     CONTINUE
48     RETURN
49   END
```

1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
66
67
68
69
70
71
72
73
74
75
76
77
78
79
80
81
82
83
84
85
86
87
88
89
90
91
92
93
94
95
96
97
98
99
100

```
1 SUBROUTINE FLINE (M, TSCALE, TWIN, I, ASCALE, AMIN, AYL)
2   DIMENSION U(12)
3   EQUIVALENCE (T, U(1))
4   DATA IUNIT /12/
5   REWIND IUNIT
6   II=0
7   READ(IUNIT, 1000, END=190)U
8   1000 FORMAT(6E13.7)
9   XO=(U(M)-TWIN)/TSCALE
10  YO=(U(I)-AMIN)/ASCALE+AYL
11  II=II+1
12  IF (II EQ 1) CALL PLOT(XO, YO, 3)
13  CALL PLOT(XO, YO, 2)
14  GO TO 100
15  RETURN
16
17
```

84/11/19 13.14.29 PAGE 1

SUBROUTINE FTN5 74/855 OPT=0, ROUND= A/ S/ M/-D/-DS FTN 5.1+587 84/11/19 . 13. 14.29 PAGE
 DO=-LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=50000
 FTNS, I, ANSI=0, L=OUTS, LO=S/-A.

```

1      SUBROUTINE F_LINES(M,TSCALE,TMIN,I,ASCALE,AMIN,AYL,HT,ISKIP,NCHAR)
2      DIMENSION U(12)
3      EQUIVALENCE (T,U(1))
4      DATA IUNIT/12/
5      REWIND IUNIT
6      II=0
7      READ(IUNIT,1000,END=190)U
8      1000 FORMAT(6E13.7)
9      X0=(U(M))-TMIN)/TSCALE
10     Y0=(U(I)-AMIN)/ASCALE+AYL
11     II=II+1
12     III=(II/ISKIP)*ISKIP
13     IF(II.NE.III)GO TO 100
14     CALL SYMBOL(X0,Y0,HT,NCHAR,O,-1)
15     GO TO 100
16     CONTINUE
17     RETURN
18
19

```

```

56      C START PLOT SEQUENCE
57      HT=.07
58      CALL PLOT(4.,5.,-3)
59
60      C      YLAB="T2(DEG F)"
61      TLAB="TIME(SEC)"
62      CALL AXIS(0.,0.,TLAB,-10,TAXL,0.,YMIN,TSCALE)
63      CALL AXIS(0.,0.,YLAB,10,YAXL,90.,YMIN,YSCALE)
64      C 4 IN CALL POINTS TO 4TH VARIABLE IN READ Y
65      CALL FLINE(1,TSCALE,TMIN,4,YSCALE,YMIN,0.)
66      C 5 POINTS TO 2
67      CALL FLINE(1,TSCALE,TMIN,5,YSCALE,YMIN,0.,HT,1.3)
68      C PLOT DEPENDENT VARIABLE
69      AYL=YAXL+1.
70      ALAB="ALPHA(DEG)"
71      CALL AXIS(0.,AYL,ALAB,10,AAXL,90.,AMIN,ASCALE)
72      C 6 POINTS TO A
73      CALL FLINE(1,TSCALE,TMIN,6,ASCALE,AMIN,AYL)
74
75      C NEXT PLOT SEQUENCE
76      AXO=TAXL+2.
77      CALL PLOT(AXO,0.,-3)
78      YLAB="Q/QREF"
79      YAXL=10.
80      YSCALE=.1
81      YMIN=0.
82
83      IFOX(1)=IFX(4)
84      DO 32 1=2,8
85      IFOX(1)=IFX(1+9)
86      CONTINUE
87      DO 200 I=1,6
88      IF(IFOX(I).EQ.0)GO TO 200
89      DATA XL/"ALPHA(DEG)" "BETA(DEG)" "LOG(RE)" ,
90      &"DELE(DEG)" "DELB(F(DEG))" "MACH" "/
91      DATA XXL/5.,6.,4.,5.,5./
92      DATA XSC/5.,1.,1.,5.,5.,5./
93      DATA XM/20.,-3.,5.,-10.,0.,0./
94      CALL AXIS(0.,0.,YLAB,10,YAXL,90.,YMIN,YSCALE)
95      CALL AXIS(0.,0.,XL(I),-10,XXL(I),0.,XM(I),XSC(I))
96      C I+5 POINTS TO ALPHA / 12 POINTS TO Q/QREF
97      IPT=I+5
98      CALL FLINE(IPT,XSC(I),XM(I),12,YSCALE,YMIN,0.)
99      CALL PLOT(AXO,0.,-3)
100     CONTINUE
101    CALL PLOT(N)
102    RETURN
103

```

SUBROUTINE FPLOT 74/855 OPT=0, ROUND= A/ S/ M/-D, -DS FTN5.1+587 84/11/19. 13.14.29 PAGE 1
 DO=-LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/- FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMOD/-ST, PL=5000
 FTNS.1, ANSI=0, L=DUTS, LO=S/-A.

```

1      SUBROUTINE FPLOT
2      COMMON/CFPLOT/IPILOT, TSCALE, TMIN, TAXL, YSCALE, YMIN, YAXL,
3      $ASCALE, AMIN, AAXL
4      DIMENSION XL(8), XXL(8), XSC(8), XM(8)
5      DIMENSION IFOX(8)
6      DIMENSION DUM(1024)
7      LOGICAL FAUTO
8      DIMENSION FAUTO(7)
9      DIMENSION QP(5)
10     COMMON/CPARAM/HO, HALF(2), PHIC, PHIK, ZP, Z, ALPH(2), KA, S(5),
11     &CIF(5,5), KAF, IFX(5), ACC(5), IFXSUM, INPAR, DALPH(2),
12     EQUIVALENCE (HO, QP(1))
13     C INITIALIZE PLOTS AND WRITE PLOT FILE TO UNIT 2
14     CALL PLOTS(DUM, 1024, 2)
15     CALL FACTDR(.787402)
16     DATA IUNIT/12/
17     REINTD IUNIT
18     C FIND MAX AND MINS FOR SCALING
19     DATA TMN, TMX, YMN, YMX/1. E7, 0., 5000., -460./
20     DATA AMN, AMX/25., 45./
21     C READ T, USI1, EQUI, USI2, UMEAS, ALPHA
22     100   READ(TUNIT, 1000, END=190) T, U, V, Y, Z, A, B, R, DE, DB, DM, QN
23     1000  FORMAT(6E13.7)
24     TMN=AMIN1(TMN, T)
25     TMX=AMAX1(TMX, T)
26     YMN=AMIN1(YMN, U, V, Y, Z)
27     YMX=AMAX1(YMX, U, V, Y, Z)
28     AMN=AMIN1(AMN, A)
29     AMX=AMAX1(AMX, A)
30     GO TO 100
31     190  CONTINUE
32     IF(IPILOT, GT, 0)GO TO 195
33     C DEFAULT TIME AXIS LENGTH = 4 INCHES
34     TAXL=4.
35     TSCALE=IFIX(((TMX-TMN)/TAXL)+.999)
36     TMIN=TMN
37     C DEFAULT Y AXIS LENGTH = 4 INCHES
38     YAXL=4
39     DYMNTN=25.
40     YSCALE=DYMIN*IFIX((YMX-YMN)/DYMIN/YAXL+.999)
41     YMIN=YSCALE*IFIX(YMN/YSCALE)
42     C DEFAULT A AXIS LENGTH = 2 INCHES
43     AAXL=2.
44     DAMIN=5
45     ASCALE=DAMIN*IFIX((AMX-AMN)/DAMIN/AAXL+.999)
46     AMIN=ASCALE*IFIX(AMN/ASCALE)
47     195  CONTINUE
48     C SCALE TIME USING INPUT TSCALE ONLY
49     C PUT IN NEGATIVE OR ZERO FOR TMIN AND TAXL
50     IF(TAXL, GT, 0)GO TO 198
51     TMIN=TMN
52     TAXL=IFIX((TMX-TMN)/TSCALE+.999)
53     198  CONTINUE
54     C
55

```

SUBROUTINE MSCALE 74/855 OPT=0, ROUND= A/ S/ W/ O/ DS FTN 5: 1+587
DO=-LONG/-OT ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
FTNS, I, ANSI=0, L=OUTS, LO=S/-A.

```
1      SUBROUTINE MSCALE (N1,N2,A,X,B)
2      DIMENSION A(1), B(1)
3      COMMON /MAIN1/ NDIM
4      DIMENSION IRAY(6)
5      DATA IRAY/6* -0/
6      IRAY(4)=0
7      JEND=N2*NDIM
8      DO 1 I=1,N1
9      CALL SYSTEMC(144,IRAY)
10     DO 1 IJ=1,JEND,NDIM
11     1 B(IJ)=X*A(IJ)
12     RETURN
13     END
```

```
1      MSCALE
2      MSCALE
3      MSCALE
4      MSCALE
5      SYSTEMC
6      SYSTEMC
7      SYSTEMC
8      SYSTEMC
9      MSCALE
10     MSCALE
11     DCT24
12     MSCALE
13     MSCALE
14     MSCALE
15     MSCALE
16     MSCALE
17     MSCALE
18     MSCALE
19     MSCALE
20     MSCALE
```

PAGE

13

84/11719 13.14.29

SUBROUTINE MMUL 777855 OPT=0,ROUND= A/ S/ M/-D,-DS FIN 5.1+587 84711/19. 13.14.26 PAGE 1
DO=-LONG/-OT, ARG=-COMMON/-FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
FTNS,I,ANSI=0,L=OUTS,L0=S/-A.

```
1 SUBROUTINE MMUL(X,Y,N1,N2,N3,Z)
2   DIMENSION X(1),Y(1),Z(1)
3   COMMON/MAIN1/NDIM
4   NDIM3=NDIM*N3
5   NDIM2=NDIM*N2
6   DO 1 I=1,N1
7   DO 1 J=1,NEND3,NDIM
8   TM=0.
9   K=I
10  KK=J-I
11  KK=KK+1
12  TM=TM+X(K)*Y(KK)
13  K=K+NDIM
14  IF(K.LE.NEND2) GO TO 5
15  1 Z(J)=TM
16  RETURN
17  END
```

1 MMUL
2 MMUL
3 MMUL
4 MMUL
5 MMUL
6 MMUL
7 MMUL
8 MMUL
9 MMUL
10 MMUL
11 MMUL
12 MMUL
13 MMUL
14 MMUL
15 MMUL
16 MMUL
17 MMUL
18 MMUL

SUBROUTINE MAT4 74/855 OPT=O, ROUND= A/ S/ N/-D,-DS FTN 5.1+587
DO=-LONG/-OT, ARG=-COMMON,-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMND/-ST, PL=5000
FTNS, I, ANSI=0, L=OUTS, LD=S/-A.

```
1          SUBROUTINE MAT4 (N1,N2,X,Y,Z)
2          C
3          Z=YXY" X=X" IS N2*XN2, Y IS N1*XN2, Z IS N1*XN1
4          DIMENSION X(1), Y(1), Z(1)
5          COMMON /MAIN1/ NDIM
6          CALL MMUL (Y,X,N1,N2,Z)
7          ND2=N2*NDIM
8          DO 3 I=1,N1
9          IM1=I-1
10         II=IM1*NDIM
11         JJ=I+II
12         DO 2 J=I,N1
13         TEMP=0,
14         KK=J
15         DO 1 K=1,NDIM
16         TEMP=TEMP+Y(K)*Z(KK)
17         KK=KK+NDIM
18         Z(JJ)=TEMP
19         JJ=JJ+NDIM
20         K=II+1
21         KK=II+IM1
22         DO 3 J=K,KK
23         Z(JJ)=Z(J)
24         JJ=JJ+NDIM
25         3 CONTINUE
26         RETURN
27         END
```

```

FUNCTION XNDRM    74/855  OPT=0, ROUND= A/ $/ M/-0/-DS   FTN 5.1+587    84/11/19.  11
DO=-LONG/-0T ARG=-COMMON/-FIXED,CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-IO/-PMD/-ST,PL=5000
FTNS5,I,ANSI=0,L=QUTS,LO=S/-A.

```

```

1      FUNCTION XNORM (NN, A)
2      COMPUTES AN APPROXIMATION TO NORM OF A-- NOT A BOUND
3      DIMENSION A(1)
4      COMMON /MAIN1/ NDIM
5      NDIM1=NDIM+1
6      NN=N*NDIM
7      C1=0.
8      TR=A(1)
9      IF (N.EQ.1) GO TO 4
10     I=2
11     DO 2 II=NDIM1,NN,NDIM
12     J=I
13     DO 1 JU=I,II,NDIM
14     C1=C1+ABS(A(J)*A(JU))
15     1 JU=JU+1
16     TR=TR+A(J)
17     2 I=I+1
18     TR=TR/FLOAT(N)
19     DO 3 II=1,NN,NDIM1
20     C1=C1+(A(II)-TR)**2
21     3 XNORM=ABS(TR)+SQRT(C1)
22     RETURN
23     END

```

SUBROUTINE TRANS 717055 OPT=0,ROUND= A7 \$7 M7-D,-DS FTN 5.1+587 84/17/19. 13.14.29 PAGE 1
DO=-LONG/-OT,ARG=-COMMON/-FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=50000
FTNS,I,ANSI=0,L=OUTS,LO=S/-A.

```
1      SUBROUTINE TRANS(A,N,X)
2      DIMENSION A(1),X(1)
3      COMMON/MATIN1/NDIM
4      DO 10 I=1,N
5      DO 10 J=1,N
6      II=I+NDIM*(J-1)
7      JJ=J+NDIM*(I-1)
8      A(II)=X(JJ)
9      RETURN
10     END
```

```
2 3      TRANSP
4      TRANSP
5      TRANSP
6      TRANSP
7      TRANSP
8      TRANSP
9      TRANSP
10     TRANSP
11     TRANSP
```

SUBROUTINE SGTSI 74/855 OPT=0, ROUND=A/ S/ M/-D, -DS FN 5.1+587 84/11/19. 13.14.29 PAGE 2

58 CONTINUE
57 60 CONTINUE
58 70 CONTINUE
59 80 CONTINUE
59 90 CONTINUE
60 100 CONTINUE
61 RETURN
62 END

SGTSI 57
SGTSI 58
SGTSI 59
SGTSI 60
SGTSI 61
SGTSI 62
SGTSI 63

SUBROUTINE SGTS_L 74/855 OPT=0, ROUND= A/ S/ M/-D/-DS / FN 5. 1+587 84/11/19. 13.14.29
 DO=-LONG/-OT, ARG=-COMMON/-FIXED, CSF= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
 FTNS, I, ANSI=O, L=OUTS, LO=S/-A.

```

1      SUBROUTINE SGTSL(N,C,D,E,B,INFO)
2      INTEGER N,INFO
3      REAL C(1),E(1),B(1),D(1)
4
5      C  SGTSL GIVEN A GENERAL TRIDIAGONAL MATRIX AND A RIGHT HAND
6      C SIDE WILL FIND THE SOLUTION - SEE THE LINPAC USER'S GUIDE
7
8      INTEGER K,KB,KP1,NM1,NM2
9
10     REAL T
11     INFO=0
12     C(1)=D(1)
13     NM1=N-1
14     IF (NM1.LT.1) GO TO 40
15     D(1)=E(1)
16     E(1)=0.0E0
17     DO 30 K=1,NM1
18     KP1=K+1
19     IF (ABS(C(KP1)).LT.ABS(C(K))) GO TO 10
20     T=C(KP1)
21     C(KP1)=C(K)
22     C(K)=T
23     T=D(KP1)
24     D(KP1)=D(K)
25     D(K)=T
26     T=E(KP1)
27     E(KP1)=E(K)
28     E(K)=T
29     T=B(KP1)
30     B(KP1)=B(K)
31     B(K)=T
32     CONTINUE
33     IF (C(K).NE.0.0E0) GO TO 20
34     INFO=K
35     GO TO 100
36     CONTINUE
37     T=-C(KP1)/C(K)
38     C(KP1)=D(KP1)+T*D(K)
39     D(KP1)=E(KP1)+T*E(K)
40     E(KP1)=0.0E0
41     B(KP1)=B(KP1)+T*B(K)
42     CONTINUE
43     IF (C(N).NE.0.0E0) GO TO 50
44     INFO=N
45     GO TO 90
46     CONTINUE
47     NM2=N-2
48     B(N)=B(N)/C(N)
49     IF (N.EQ.1) GO TO 80
50     B(NM1)=(B(NM1)-D(NM1)*B(N))/C(NM1)
51     IF (NM2.LT.1) GO TO 70
52     DO 60 KB=1,NM2
53     K=NM2-KB+1
54     B(K)=(B(K)-D(K)*B(K+1)-E(K)*B(K+2))/C(K)
55

```

SUBROUTINE INTEG 747855 OPT=0,ROUND= A/ S/ M/-D,-DS F7M 5.1+587
DO=-LONG/-OT,ARG=-COMMON/-FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
FTNS,I,ANSI=0,L=OUTS,LO=S/-A.

```
1          SUBROUTINE INTEG(N,A,T,QUE,PHI,QT,QDT,N2)
2          DIMENSION A(N2,N2),QUE(N2,N2),PHI(N2,N2),QDT(N2,N2)
3          T2=-T*0.5
4          CALL MEXP(N,A(1,1),T2,PHI(1,1),QD(1,1),QDT(1,1),N2)
5          CALL TRI(N,QUE(1,1),PHI(1,1),QD(1,1),QDT(1,1),N2)
6          CALL MSCALE(N,N,QDT(1,1),4.0,QDT(1,1))
7          CALL MULT(PHI(1,1),PHI(1,1),N,A(1,1),N2)
8          CALL TRI(N,QUE(1,1),A(1,1),QD(1,1),PHI(1,1),N2)
9          DO 10 I=1,N
10         DO 10 J=1,N
11         QDT(I,J)=QUE(I,J)+ QDT(I,J)+ PHI(I,J)
12         T6=T/6.0
13         CALL MSCALE(N,N,QDT(1,1),T6,QDT(1,1))
14         RETURN
15         END
```

```
2          INTEG2
3          INTEG2
4          INTEG2
5          INTEG2
6          INTEG2
7          INTEG2
8          INTEG2
9          INTEG2
10         INTEG2
11         INTEG2
12         INTEG2
13         INTEG2
14         INTEG2
15         INTEG2
16         INTEG2
```

SUBROUTINE HEATTUN 74/855 OPT=0, ROUND= A/ S/ M/-D, -DS FIN 5.1+587 84/11/19. 13. 14. 29 PAGE 1
DO=-LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
FTNS, I, ANSI=0, L=OUTS, LO=S/-A.

```
1 SUBROUTINE HEATTUN
2 COMMON/CHEAT/Q, TS, QREF, TW, W1, RENS, HBAR, HREF
3 COMMON/COMTUN/T, TAW1, ALPHA, H, V, RHO, P, TEMP, C, TRAD, RHOG,
4 ATO, TSINK, XFT, DEL, PDEL
5 LOGICAL FAUTO
6 DIMENSION FAUTO(7)
7 DIMENSION QP(5)
8 COMMON/CPARAM/HO, HALF(2), PHIC, PHIK, ZP, Z, ALPH(2), KA, S(5),
9 &CIF(5,5), KAF, IFX(5), ACC(5), IFXSUM, NPAR, DALPH(2),
10 EQUIVALENCE (HO, QP(1))
11 COMMON/ICTPS2/TINIT(1), ERALOW, E
12 COMMON/CDX/DX(1)
13 DATA HREF/1./
14
15 C CHECK FOR ALPHA SEGMENT
16 C
17 IF(ALPHA.LT.ALPH(2)) THEN
18   DALPH(1)=ALPHA
19   DALPH(2)=ALPH(2)
20 ENDIF
21 IF(ALPHA.GE.ALPH(2)) THEN
22   DALPH(1)=ALPH(1)
23   DALPH(2)=ALPHA
24 ENDIF
25
26 HBAR=HO+HALF(1)*(DALPH(1)-ALPH(1))+HALF(2)*(DALPH(2)-ALPH(2))
27 RETURN
28 END
```

SUBROUTINE TPSOSP2 74/855 OPT=0, ROUND=A/S/ M/-D,-US
DO=-LONG/-DT, ARG=-COMMON, CS= USER/-FIXED, DB=-TB/-SB/-SL/
FTN5, I, ANSI=0, L=OUTS, LO=S/-A.

```
1          SUBROUTINE TPSOSP2(DT)
2          COMMON/CDSP/NPTSS,USI(6),PHI(6,6),NPT,PC(6,6),RR,
3          &QD(6,6),QDT(6,6),QUE(6,6),A(6,6),RCX(6),RP(6),RM(6)
4          COMMON/COPT/N,IOA(8),IOB(8),IOC(8),IOFACE(8),IOD(8)
5          COMMON/CCPC/NPTPC
6          COMMON /MAIN1/INP ,IPVT(6),WORK(6)
7          INP=6
8          IF(DT .LE. 0.) GO TO 999
9          T=DT
10         CALL INTEG(NPTPC,A,T,QUE,PHI,QD,QDT,NPTSS)
11         CALL SGFFA(A(1,1),NPTSS,NPTPC,IPVT(1))
12         CALL SGEDA(A(1,1),NPTSS,NPTPC,IPVT(1))
13         CALL MAT4(NPTPC,NPTPC,QDT(1,1),A(1,1),QD(1,1))
14         CALL MAT4(NPTPC,NPTPC,PC(1,1),A(1,1),QDT(1,1))
15         DO 20 I=1,NPTPC
16         DO 20 J=1,NPTPC
17         PHI(I,J)=A(I,J)
18         20 PC(I,J)= QDT(I,J)+ QD(I,J)
19         999 RETURN
20         END
```

PAGE 1

84/11/19.

13.14.29

FTN 5.1+587

PL=5000

```
2          FTPSOSP3
3          UPAUG16
4          UPOCT09
5          FTPSOSP3
6          FTPSOSP3
7          OCT10
8          OCT10
9          FTPSOSP3
10         UPSEP24
11         FTPSOSP3
12         FTPSOSP3
13         FTPSOSP3
14         FTPSOSP3
15         FTPSOSP3
16         FTPSOSP3
17         FTPSOSP3
18         FTPSOSP3
19         FTPSOSP3
20         FTPSOSP3
```

SUBROUTINE EQUATE 74/855 OPT=0 ROUND= A/ S/ M/-D/-DS FTN 5.1+587
DO=-LONG/-OT ARG=-COMMON/-COMMON/-FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
FTN5,I,ANSI=0,L=OUTS,LO=S/-A.

```
1          SUBROUTINE EQUATE (NR,NC,A,B)
2          DIMENSION A(1), B(1)
3          COMMON /MAIN1/ NDIM
4          NN=NC*NDIM
5          NR1=NR-1
6          DO 1 J=1,NN,NDIM
7          II=J+NR1
8          DO 1 IJ=J,II
9          A(IJ)=B(IJ)
10         1 CONTINUE
11         RETURN
12         END
```

SUBROUTINE GMINV 74/855 OPT=0, ROUND=A/S/ M/-D/-DS
 DO=-LONG/-OT, ARG=-COMMON, CS=-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/
 FTN5, I, ANSI=0, L=OUTS, LO=S/-A.

FTN 5.1+587 84/11/19 13.14.29 PAGE 1

```

1          SUBROUTINE GMINV (NR, NC, A, U, MR, MT)
2          DIMENSION A(1), U(1), S(30)
3          COMMON /MAIN2/ NDIM
4          COMMON /INDU/ KIN, KOUT
5          NDIM1=NDIM+1
6          TOL=1.E-14
7          ADV=1.E-24
8          MR=MC
9          NRM1=NR-1
10         TOL1=0.
11         JU=1
12         DO 1 J=1, NC
13         S(J)=DOT(NR,A(JJ),A(JJ))
14         IF (S(J).GT.TOL1) TOL1=S(J)
1      JU=JJ+NDIM
2      TOL1=ADV*TOL1
3      ADV=TOL1
4      JU=1
5      DO 14 J=1, NC
6      FAC=S(J)
7      JMI=J-1
8      JRN=JJ+JN1
9      JCM=JJ+JU
10     DO 2 I=JJ, JCM
11     U(I)=0.
12     U(JCM)=1.0
13     IF (J.EQ.1) GO TO 5
14     KK=1
15     DO 3 K=1, JMI
16     IF (S(K).EQ.1.0) GO TO 3
17     TEMP=-DOT(NR,A(JJ),A(KK))
18     CALL VADD (K, TEMP, U(JJ), U(KK))
19     3 KK=KK+NDIM
20     DO 4 L=1, 2
21     KK=1
22     DO 4 K=1, JMI
23     IF (S(K).EQ.0.) GO TO 4
24     TEMP=-DOT(NR,A(JJ),A(KK))
25     CALL VADD (K, TEMP, U(JJ), U(KK))
26     4 KK=KK+NDIM
27     DO 5 I=JU, JRN
28     KK=1
29     DO 3 K=1, JMI
30     IF (S(K).EQ.0.) GO TO 3
31     TEMP=-DOT(NR,A(JJ),A(KK))
32     CALL VADD (K, TEMP, U(JJ), U(KK))
33     3 KK=KK+NDIM
34     DO 4 L=1, 2
35     KK=1
36     DO 4 K=1, JMI
37     IF (S(K).EQ.0.) GO TO 4
38     TEMP=-DOT(NR,A(JJ),A(KK))
39     CALL VADD (NR, TEMP, A(JJ), A(KK))
40     CALL VADD (K, TEMP, U(JJ), U(KK))
41     4 KK=KK+NDIM
42     TOL1=TOL*FAC+ADV
43     FAC=DOT(NR,A(JJ),A(JJ))
44     5 IF (FAC.GT.TOL1) GO TO 9
45     DO 6 I=JU, JRN
46     A(I)=0.
47     S(J)=0.
48     KK=1
49     IF (S(K).EQ.0.) KK=KK+NDIM
50     IF (S(K).EQ.0.) GO TO 8
51     DO 7 K=1, JMI
52     TEMP=-DOT(K, U(KK), U(JJ))
53     CALL VADD (NR, TEMP, A(JJ), A(KK))
54     7 KK=KK+NDIM
55     8 FAC=DOT(J, U(JJ), U(JJ))

```

```

56          MR=MR-1
57          GO TO 11
58          * S(J)=1.0
59          KK=1
60          DO 10 K=1, JN1
61          IF (S(K).EQ.1.) GO TO 10
62          TEMP=-DOT(NR,A(JJ),A(KK))
63          CALL VADD (K,TEMP,U(JJ),U(KK))
64          KK=KK+NDIM
65          FAC=1./SQRT(FAC)
66          DO 12 I=JJ,JRN
67          A(I)=A(I)*FAC
68          DO 13 I=JJ,JCM
69          U(I)=U(I)*FAC
70          IF (JJ+NDIM
71          IF (MR.EQ.NR.OR.MR.EQ.NC) GO TO 15
72          IF (MT.NE.0) WRITE (KOUT,19) NR,NC,MR
73          NEND=NC*NDIM
74          JJ=1
75          DO 18 J=1, NC
76          DO 16 I=1, NR
77          II=I-J
78          S(I)=0.
79          DO 18 KK=JJ,NEND,NDIM
80          S(I)=S(I)+A(II+KK)*U(KK)
81          II=J
82          DO 17 I=1, NR
83          U(II)=S(I)
84          17 II=II+NDIM
85          18 JJ=JJ+NDIM
86          RETURN
87          19 FORMAT (13, 1HX, I2, 8H M: RANK, I2)
88          END

```

FUNCTION DOT 747855 OPT=0,ROUND= A/ S/ M/-D, -DS FTN 5, T+587 84/11/19. 13.14.28 PAGE 1
DO=LONG/, OT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SL/ ER/-ID/ PMD/-ST, PL=5000
FTNS, I, ANSI=0, L=OUTS, LO=S/-A.

```
FUNCTION DOT (NR,A,B)
DIMENSION A(1), B(1)
DOT=0.
DO 1 I=1, NR
 1 DOT=DOT+A(I)*B(I)
RETURN
END
```

1 2 3 4 5 6 7

2 3 4 5 6 7 8

FDOT FDOT FDOT FDOT FDOT FDOT FDOT FDOT

SUBROUTINE VADD 747855 OPT=0,ROUND= A/ S/ M/-D/-DS F7N 5.1+587
00=-LONG/-OT,ARG=-COMMON,-FIXED,CS= USER/-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
FTNS,I,ANSI=0,L=OUTS,LO=S/-A.

```
1 SUBROUTINE VADD (N,C1,A,B)
2   DIMENSION A(1), B(1)
3   DO 1 I=1,N
4   1 A(I)=A(I)+C1*B(I)
5   RETURN
6   END
```

2 3 4 5 6 7
FVADD
FVADD
FVADD
FVADD
FVADD
FVADD
FVADD

ROUTINE ZERO 747855 OPT=0, ROUND=A/ S/ M/-D, -DS FTN 5.1+587
DO=-LONG/-OT, ARG=-COMMON/-FIXED, CS= USER/-FIXED, DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST, PL=5000
FTNS, I, ANSI=0, L=OUTS, LO=S/-A.

```
1 SUBROUTINE ZERO (A, NR, NC)
2   DIMENSION A(NR, NC)
3   DO 1 IC=1 NC
4     DO 1 IR=1 NR
5       1 A(IR, IC)=0.0
6     RETURN
7   END
```

```
2 FZERO
3 FZERO
4 FZERO
5 FZERO
6 FZERO
7 FZERO
8 FZERO
```

SUBROUTINE YADD 747155 OPT=O,ROUND=A/ S/ N/-D,-DS FTN 5.1+587 87/11/19 13.14.29 PAGE 1
DO=-LONG/-OT,ARG=-COMMON,-FIXED,CS=USER,-FIXED,DB=-TB/-SB/-SL/ ER/-ID/-PMD/-ST,PL=5000
FTNS,I,ANSI=O,L=OUTS,LO=S/-A.

```
1      SUBROUTINE YADD (N,C1,A,B)
2      DIMENSION A(1), B(1)
3      DO 1 I=1,N
4      1 A(I)=A(I)+C1*B(I)
5      RETURN
6      END
```

2 FVADD
3 FVADD
4 FVADD
5 FVADD
6 FVADD
7 FVADD

BIBLIOGRAPHY

1. Neumann, R.D., Hayes, J.R., "Advanced Techniques for the Acquisition of Aerodynamic Heating Data in Hypersonic Continuous Flow Test Facilities." AFWAL/FIMG, Wright-Patterson AFB, OH. Paper presented at 3rd Aerothermal Workshop, 28 June 1984.
2. Hodge, J.K., Woo, Y.K., Cappelano, P.T., "Parameter Estimation for Imbedded Thermocouples in Space Shuttle Wind Tunnel Test Articles with a Nonisothermal Wall." AIAA-83-1533, June 1983.
3. Knox, E.C., Martindale, W.R., "Coax Heat Transfer Gage Usage in the VKF Tunnels B and C," VKF-TM-b.6, Sverdrup ARO, Inc. AEDC Division, January 1979.
4. Hodge, J.K., Audley, D.R., Phillips, P.W., and Hertzler, E.K., "Aerothermodynamic Flight Envelope Expansion for a Manned Lifting Reentry Vehicle (Space Shuttle)." Paper 3-B, AGARD CP-339, October 1982.
5. Kreith, F., Principles of Heat Transfer, 2nd edition, International Textbook Company, July 1968.
6. Lutes, C.D., Hodge, J.K., "Nonlinear Modeling and Initial Condition Estimation for Identifying the Aerothermodynamic Environment of the Space Shuttle Orbiter." AIAA-84-1749
7. Maybeck, P.S., Stochastic Models, Estimation, and Control. Vols I and II, Academic Press, 1979 (Vol I) and 1982 (Vol II).
8. Eykhoff, Pieter, System Identification, John Wiley, 1974.
9. Carslaw, H.S. and Jaeger, J.C., Conduction of Heat in Solids, Second Edition, Oxford, Clarendon Press, 1959.

VITA

Neil Thomas Cahoon was born on 31 July 1954 in St.Paul, Minnesota, the son of Thomas C. and Eleanor I. Cahoon. He graduated from Henry Sibley Senior High School in 1972 and the United States Air Force Academy in 1976 from which he received a Bachelor's Degree in Aeronautical Engineering. He completed pilot training and received his wings in September 1977 whereupon he was assigned as an RF4C pilot to the 67th Tactical Reconnaissance Wing, Bergstrom AFB, Texas until May 1981. He was then assigned as an RF4C instructor pilot and later as a flight examiner to the 363rd Tactical Fighter Wing, Shaw AFB, South Carolina until entering the school of Engineering, Air Force Institute of Technology, in May 1983.

permanent address: 3530 E. Marshal Gulch
Tucson, Arizona 85718

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE

REPORT DOCUMENTATION PAGE

1a. REPORT SECURITY CLASSIFICATION UNCLASSIFIED		1b. RESTRICTIVE MARKINGS	
2a. SECURITY CLASSIFICATION AUTHORITY		3. DISTRIBUTION/AVAILABILITY OF REPORT Approved for public release; distribution unlimited	
2b. DECLASSIFICATION/DOWNGRADING SCHEDULE AFIT/GAE/AA/84D-3			
4. PERFORMING ORGANIZATION REPORT NUMBER(S)		5. MONITORING ORGANIZATION REPORT NUMBER(S)	
6a. NAME OF PERFORMING ORGANIZATION School of Engineering	6b. OFFICE SYMBOL (If applicable) AFIT/ENY	7a. NAME OF MONITORING ORGANIZATION	
6c. ADDRESS (City, State and ZIP Code) Air Force Institute of Technology Wright-Patterson AFB, Ohio 45433		7b. ADDRESS (City, State and ZIP Code)	
8a. NAME OF FUNDING/SPONSORING ORGANIZATION	8b. OFFICE SYMBOL (If applicable)	9. PROCUREMENT INSTRUMENT IDENTIFICATION NUMBER	
8c. ADDRESS (City, State and ZIP Code)		10. SOURCE OF FUNDING NOS.	
		PROGRAM ELEMENT NO.	PROJECT NO.
		TASK NO.	WORK UNIT NO.
11. TITLE (Include Security Classification) See Box 19			
12. PERSONAL AUTHOR(S) Neil T. Cahoon, B.S., Captain, USAF			
13a. TYPE OF REPORT MS Thesis	13b. TIME COVERED FROM _____ TO _____	14. DATE OF REPORT (Yr., Mo., Day)	15. PAGE COUNT 130
16. SUPPLEMENTARY NOTATION <i>L</i> <i>11.1</i>			
17. COSATI CODES		18. SUBJECT TERMS (Continue on reverse if necessary and identify by block number) Aeronautics, Aerodynamics, Thermodynamics, Heat Transfer, Heat Transfer Coefficients, Thermal Conductivity, Stochastic Processes (Field 12, Group 01)	
19. ABSTRACT (Continue on reverse if necessary and identify by block number)			
<p>Title: Heating Parameter Estimation Using Coaxial Thermocouple Gages in Wind Tunnel Test Articles</p> <p>Thesis Advisor: James K. Hodge, Major, USAF</p> <p>Approved for public release: IAW AFR 190-17. Lynn E. Wolden Director, Education and Professional Development Air Force Institute of Technology (AFTC) Wright-Patterson AFB, OH 45433</p>			
20. DISTRIBUTION/AVAILABILITY OF ABSTRACT <input checked="" type="checkbox"/> UNCLASSIFIED/UNLIMITED <input type="checkbox"/> SAME AS RPT <input type="checkbox"/> DTIC USERS <input type="checkbox"/>		21. ABSTRACT SECURITY CLASSIFICATION UNCLASSIFIED	
22a. NAME OF RESPONSIBLE INDIVIDUAL James K. Hodge, Major, USAF		22b. TELEPHONE NUMBER (Include Area Code) 513-255-3517	22c. OFFICE SYMBOL AFIT/ENY

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE

A heat energy balance is applied to a coaxial thermocouple gage for parameter estimation in wind tunnel test articles. This method can significantly reduce wind tunnel test costs and time. Modifications to the data reduction program HEATEST (HEATing ESTimation) are made. The program allows for transient test techniques to be used as well as assuming an isothermal wall. A non-linear convective heat transfer coefficient model may also be used. Data is generated to test the new program. Temperature profiles throughout the thermocouple gage were good and were compared with changes in time step, thermocouple length, and number of discrete node points. The estimation of the convective heat transfer coefficient and thermal conductivity were excellent.

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE

END

FILMED

5-85

DTIC